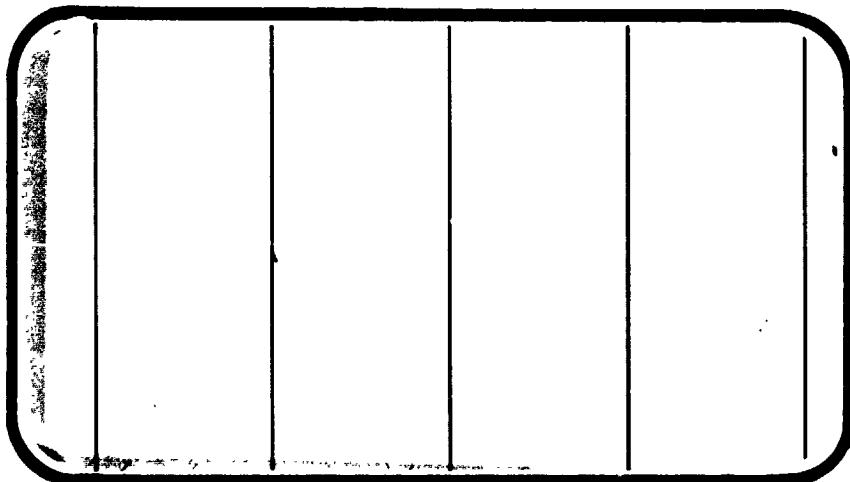


NASA

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION



(NASA-CR-151062) SUPERSONIC CONTROL
EFFECTIVENESS FOR FULL AND PARTIAL SPAN
ELEVON CONFIGURATIONS ON A 0.0165 SCALE
MODEL SPACE SHUTTLE ORBITER TESTED IN THE
LARC UNITARY PLAN WIND TUNNEL (Chrysler

N77-20147

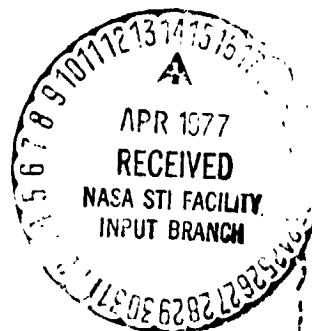
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Unclassified
G3/16 22842

SPACE SHUTTLE

AEROTHERMODYNAMIC DATA REPORT

JOHNSON SPACE CENTER
HOUSTON, TEXAS

DATA MANAGEMENT SERVICES
SPACE DIVISION  CHRYSLER
CORPORATION



March 1977

DMS-DR-2182
NASA CR-151,062

SUPersonic CONTROL EFFECTIVENESS FOR FULL AND
PARTIAL SPAN ELEVON CONFIGURATIONS ON A 0.0165
SCALE MODEL SPACE SHUTTLE ORBITER TESTED IN
THE LaRC UNITARY PLAN WIND TUNNEL (LA49)

Prepared under NASA Contract Number NAS9-13247

by

Data Management Services
Chrysler Corporation Michoud Defense-Space Division
New Orleans, La. 70189

for

Engineering Analysis Division

Johnson Space Center
National Aeronautics and Space Administration
Houston, Texas

WIND TUNNEL TEST SPECIFICS:

Test Number: LaRC UPWT (Leg 2) 1101
NASA Series Number: LA49
Model Number: 089B-139
Test Dates: April 24 through 26, 1974
Occupancy Hours: 30

FACILITY COORDINATOR:

Bernard Spencer, Jr.
Langley Research Center
Mail Stop 365
Langley Station
Hampton, Virginia 23665

Phone: (804) 827-3911

PROJECT ENGINEERS:

Bernard Spencer, Jr.
Langley Research Center
Mail Stop 365
Langley Station
Hampton, Virginia 23665

Phone: (804) 827-3911

George Ware
Langley Research Center
Mail Stop 365
Langley Station
Hampton, Virginia 23665

Phone: (804) 827-3911

DATA MANAGEMENT SERVICES:

Prepared by: Liaison--J. W. Ball
Operations--Maurice Moser, Jr.

Reviewed by: G. G. McDonald

Approved: J. J. Glynn
J. L. Glynn, Manager
Data Operations

Concurrence: N. D. Kemp
N. D. Kemp, Manager
Data Management Services

Chrysler Corporation Michoud Defense-Space Division assumes no responsibility for the data presented other than display characteristics.

SUPersonic CONTROL EFFECTIVENESS FOR FULL AND
PARTIAL SPAN ELEVON CONFIGURATIONS ON A 0.0165
SCALE MODEL SPACE SHUTTLE ORBITER TESTED IN
THE LARC UNITARY PLAN WIND TUNNEL (LA49)

ABSTRACT

An experimental investigation has been conducted in the NASA-Langley Research Center Unitary Plan Wind Tunnel (Ref. 2) on an early version of the space shuttle orbiter (designated 089B-139) 0.0165 scale model to systematically determine both longitudinal and lateral control effectiveness associated with various combinations of inboard, outboard, and full span wing trailing edge controls. This report presents results from supersonic investigations conducted from April 24 through 26, 1974. The test Mach numbers were 2.5 and 4.63 over an angle of attack range from -4° to 40° at 0° sideslip.

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PLOTTED COEFFICIENTS SCHEDULE:

- A) CL, CLM, CD, L/D versus ALPHA
- B) CLM versus ELEVTR, DCL/DE, DCD/DE, DCLMDE versus ALPHA
- C) CLM versus ELV-LI, DCL/DE, DCD/DE, DCLMDE versus ALPHA
- D) CLM versus ELV-LO, DCL/DE, DCD/DE, DCLMDE versus ALPHA
- E) DCL/DE, DCD/DE, DCLMDE versus ALPHA
- F) DCY/DA, DCYMDA, DCBILDA versus ALPHA
- G) DCMIDE, DCLMDE, DCMI/F versus ALPHA
- H) DCMODE, DCLMDE, DCMO/F

NOMENCLATURE
General

<u>SYMBOL</u>	<u>HEMNIC</u>	<u>DEFINITION</u>
a		speed of sound; m/sec, ft/sec
C_p	CP	pressure coefficient; $(p_1 - p_\infty)/q$
M	MACH	Mach number; V/a
p		pressure; N/m ² , psf
q	$Q(\text{NSM})$ $Q(\text{PSF})$	dynamic pressure; $1/2\rho V^2$, N/m ² , psf
RN/L	RN/L	unit Reynolds number; per m, per ft
V		velocity; m/sec, ft/sec
α	ALPHA	angle of attack, degrees
β	BETA	angle of sideslip, degrees
ψ	PSI	angle of yaw, degrees
ϕ	PHI	angle of roll, degrees
ρ		mass density; kg/m ³ , slugs/ft ³

Reference & C.G. Definitions

A_b		base area; m ² , ft ²
b	BREF	wing span or reference span; m, ft
c.g.		center of gravity
L_{REF}	LREF	reference length or wing mean aerodynamic chord; m, ft
S	SREF	wing area or reference area; m ² , ft ²
	MRP	moment reference point
	XMRP	moment reference point on X axis
	YMRP	moment reference point on Y axis
	ZMRP	moment reference point on Z axis

SUBSCRIPTS

b	base
l	local
s	static conditions
t	total conditions
∞	free stream

NOMENCLATURE (Continued)

Body-Axis System

<u>SYMBOL</u>	<u>MNEMONIC</u>	<u>DEFINITION</u>
C_N	CN	normal-force coefficient; $\frac{\text{normal force}}{qS}$
C_A	CA	axial-force coefficient; $\frac{\text{axial force}}{qS}$
C_Y	CY	side-force coefficient; $\frac{\text{side force}}{qS}$
C_{Ab}	CAB	base-force coefficient; $\frac{\text{base force}}{qS}$ $-A_b(p_b - p_\infty)/qS$
C_{Af}	CAF	forebody axial force coefficient, $C_A - C_{Ab}$
C_m	CTM	pitching-moment coefficient; $\frac{\text{pitching moment}}{qS/\text{REF}}$
C_n	CYN	yawing-moment coefficient; $\frac{\text{yawing moment}}{qSb}$
C_l	CBL	rolling-moment coefficient; $\frac{\text{rolling moment}}{qSb}$

Stability-Axis System

C_L	CL	lift coefficient; $\frac{\text{lift}}{qS}$
C_D	CD	drag coefficient; $\frac{\text{drag}}{qS}$
C_{D_b}	CDB	base-drag coefficient; $\frac{\text{base drag}}{qS}$
C_{D_f}	CDF	forebody drag coefficient; $C_D - C_{D_b}$
C_Y	CY	side-force coefficient; $\frac{\text{side force}}{qS}$
C_m	CLM	pitching-moment coefficient; $\frac{\text{pitching moment}}{qS/\text{REF}}$
C_n	CLN	yawing-moment coefficient; $\frac{\text{yawing moment}}{qSb}$
C_l	CSL	rolling-moment coefficient; $\frac{\text{rolling moment}}{qSb}$
L/D	L/D	lift-to-drag ratio; C_L/C_D
L/D_f	L/DF	lift to forebody drag ratio; C_L/C_{D_f}

NOMENCLATURE (Continued)

<u>SYMBOL</u>	<u>MNEMONIC</u>	<u>DEFINITION</u>
$C_{D\delta_e}$	DCD/D ^e	Slope of drag coefficient vs. elevon deflection curve; $dC_D/d\delta_e$, per degree
$C_{L\delta_e}$	DCL/D ^e	Slope of lift coefficient vs. elevon deflection curve; $dC_L/d\delta_e$, per degree
$C_I\delta_a$	DCBLDA	Slope of rolling moment coefficient vs. aileron deflection curve, $dC_I/d\delta_a$, per degree
$C_m\delta_e$	DCLMDE	Slope of pitching moment coefficient vs. elevon deflection curve, $dC_m/d\delta_e$, per degree
$C_Y\delta_a$	DCY/D ^a	Slope of side force coefficient vs. aileron deflection curve, $dC_Y/d\delta_a$, per degree
$C_n\delta_a$	DCYNDA	Slope of yawing moment coefficient vs. aileron deflection curve, $dC_n/d\delta_a$, per degree
δ_a	AILRON	Aileron deflection angle; elevon deflection for roll control, $(\delta_{aL} - \delta_{aR})/2$, positive deflection left panel trailing edge down.
δ_e	ELEVTR	Elevator deflection angle; elevon deflection for pitch control $(\delta_{eL} + \delta_{eR})/2$, positive deflection trailing edge down
δ_{eL_O}	ELV-LO	Left outboard elevon panel deflection, degrees
δ_{eL_I}	ELV-LI	Left inboard elevon panel deflection, degrees
δ_{eR_I}	ELV-RI	Right inboard elevon panel deflection, degrees
δ_{eR_O}	ELV-RO	Right outboard elevon panel deflection, degrees
$C_m\delta_{eI}$	DCMIDE	Slope of pitching moment coefficient versus inboard elevon deflection curve, per degree

NOMENCLATURE (Concluded)

<u>SYMBOL</u>	<u>MEMONIC</u>	<u>DEFINITION</u>
$C_m \delta_{e_o}$	DC10DE	Slope of pitching moment coefficient versus outboard elevon deflection curve, per degree.
	DCMI/F	Ratio of the slopes of the inboard elevon pitching moment curve over the full span elevon pitching moment curve.
	DCMO/F	Ratio of the slopes of the outboard elevon pitching moment curve over the full span elevon pitching moment curve.
δ_{SB}	SPDBRK	speed brake deflection angle, degrees
δ_{BF}	BDFLAP	body flap deflection angle, degrees

INTRODUCTION

As a continuing effort to identify the most suitable space shuttle concept, a joint study between Langley Research Center, Johnson Space Center, and Rockwell International has been undertaken to determine if the independent operation of the four elevon surfaces of the orbiter could result in a more efficient use of available control power, reduced elevon hinge moments, and associated aeroelastic wing bending, or allow a more flexible flight profile without adverse control characteristics.

Therefore, an experimental investigation at subsonic and supersonic speeds was initiated at Langley to systematically determine both longitudinal and lateral control effectiveness associated with various combinations of inboard, outboard, and full span wing trailing edge controls for a shuttle orbiter configuration. Due to the unavailability of a current vehicle model, the model employed in this study was an 0.0165 scale earlier version of the orbiter designated by Rockwell International as configuration 089B-139. The differences between this configuration and the current design ("Vehicle 5") were not felt to be sufficient to alter the incremental effectiveness presented herein. This report presents the initial supersonic results obtained in the overall study. Utilizing the Langley Unitary Plan Wind Tunnel, the Mach numbers of the investigation were 2.5 and 4.63. Angle of attack was varied from about -1° to as much as 42° at 0° of sideslip. Transonic results are presented in the reference.

CONFIGURATIONS INVESTIGATED

The configuration tested was a 0.0165 scale model of a blend of Rockwell International Shuttle configurations consisting of a 089B configuration with a 139B configuration nose forward of fuselage station 500. A sketch and photographs of the model are shown in figures 2 and 3, respectively. Body base flap was fixed at 0° deflection.

Elevon controls were split at 0.60 b/2 giving the inboard and outboard segments approximately 53 percent and 47 percent of the total elevon area, respectively. The surfaces could be deflected in unison or as individual panels. Maximum range of deflection for each panel was from 0° to -40°. Combinations tested included: for pitch control, inboards only, outboards only and full span; for roll control, outboards, only with full span and inboard deflected for pitch control.

To expedite testing, the elevons were remotely controlled by four internal electric motors (see fig. 2c).

A complete description of model dimensional data is given in table III.

TEST CONDITIONS

The model was sting supported, with aerodynamic forces and moments measured by an internally mounted six-component strain gage balance. Model angle of attack was varied from about -4° to as much as 42° at an angle of sideslip of 0° . Reynolds number was constant at a nominal 2.0×10^6 per foot. Angle of attack has been corrected for deflection of the sting and balance under load.

Transition strips 0.16 cm wide composed of No. 60 sand grit were located 1.0 inch aft of the apex of the fuselage and 0.5 inch (measured streamwise) aft of the wing and fillet and vertical tail leading edges.

Drag data presented herein represent gross drag; in that measured drag is uncorrected for base pressure effects.

TEST FACILITY DESCRIPTION

The NACA Langley Research Center 4 foot Unitary Plan Wind Tunnel (UPWT) is a closed-circuit continuous flow, variable density facility. The test section is 4 feet by 4 feet by 7 feet long.

Two tunnel legs are available for supersonic testing in the Mach number ranges 1.47 to 2.86 (Leg No. 1) and 2.29 to 4.63 (Leg No. 2). Leg No. 2 was used for this test. An asymmetric, sliding block nozzle position and total pressure setting provide the test Mach numbers at a specified Reynolds number. Reynolds number can be varied from 0.76 to 7.78 million per foot. Available stagnation pressure variation is 4.0 to 142 psia. Dynamic pressure variation is 95 to 1260 psf with normal operating stagnation temperature about 150°F in Mach modes 2 or 3 and about 175°F in Mach mode 4. The tunnel is equipped with a dry air supply, an evacuating system, and a cooling system. The facility power is approximately 83,000 horsepower.

Model mounting provisions consist of various sting arrangements, including axial (longitudinal), lateral (independent pitch and yaw), and roll movement with side wall support. A Schlieren system and oil flow visualization equipment are available. Data are recorded at the tunnel and reduced off-line at the Langley Computer Center. The tunnel is used for force and moment, pressure, and dynamic stability tests. Hot and cold jet effects and heat transfer have been studied in the UPWT.

DATA REDUCTION

Data was recorded at the facility and reduced off-line at the LaRC Computation Center. Longitudinal data are referred to the stability-axis system and lateral-directional data are referred to the body-axis system. All coefficients are normalized with respect to the projected wing area (excluding the fillet), mean aerodynamic chord or span, which are:

$$SPEF = \text{wing projected area} = 0.732 \text{ ft.}^2$$

$$LREF = \text{wing mean aerodynamic chord} = 7.83\frac{1}{4} \text{ in.}$$

$$BREF = \text{wing span} = 15.45 \text{ in.}$$

All data are presented along a set of body and stability axes (Figure 1) passing through the estimated forward center of gravity located at a full scale fuselage station of 1076.48 in. or 65% of the actual body length.

Elevon and aileron derivative data were computer-generated by the Chrysler DATAMAN-GADSAC Program and represent the local slope of the coefficient vs. control deflection at each value of angle of attack.

REFERENCE

DMS-DR-2184, "TRANSONIC CONTROL EFFECTIVENESS FOR FULL AND PARTIAL SPAN ELEVON CONFIGURATIONS ON A 0.0165 SCALE MODEL SPACE SHUTTLE ORBITER TESTED IN THE LaRC 8-FOOT TRANSONIC PRESSURE TUNNEL (LA48)."

TABLE I

TEST : LARC UPWT 1101 (LA49)

DATE : 1-21-77

TEST CONDITIONS

BALANCE UTILIZED: UT27-50

	CAPACITY:	ACCURACY:	COEFFICIENT TOLERANCE:
NF	<u>600 lb.</u>	<u>± 3.0 lb.</u>	_____
SF	<u>300 lb.</u>	<u>± 1.5 lb.</u>	_____
AF	<u>50 lb.</u>	<u>± 0.25 lb.</u>	_____
PM	<u>800 in-lb.</u>	<u>± 4.0 in-lb.</u>	_____
RM	<u>400 in-lb.</u>	<u>± 2.0 in-lb.</u>	_____
YM	<u>600 in-lb.</u>	<u>± 3.0 in-lb.</u>	_____

COMMENTS:

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TEST : UPWT 1101 (LA-49)

TABLE II
DATA SET RUN NUMBER COLLATION SUMMARY

DATA SET IDENTIFIER	CONFIGURATION	SCHD. ELEVON DEFLECTION						NO. OF RUNS	MACH NUMBERS
		α	β	LO	LI	RI	RO		
RH1001	089B-139 Orb.	A	0	0	0	0	0	25	0
002				0	-10	-10	0		
003				0	-20	-20	0		
004				0	-40	-40	0		
005				-10	-10	-10	-10		
006				-20	-20	-20	-20		
007				-40	-40	-40	-40		
008				-10	0	0	-10		
009				-20	0	0	-20		
010				-40	0	0	-40		
011				0	0	-20	-20		
012				-10	-10	-30	-30		
013				0	-10	-10	-20		
014				0	-20	-20	-40		
015				-10	-20	-20	-30		
016				+10	-20	-20	-10		
017				+10	-40	-40	-10		
018				-20	-30	-30	-40		
		7	13	19	25	31	37	43	49
								55	61
									67
									75 76

DATA SET SCHEDULES
 $\alpha = \beta$ A) -4° to 32° at $M = 2.5$
-4° to 40° at $M = 4.63$

COEFFICIENTS
IDVAR (1) IC-AF (2) IC-CV

TABLE III
MODEL DIMENSIONAL DATA

MODEL COMPONENT : BODY - B20

GENERAL DESCRIPTION : 089B-139B (MODIFIED NOSE), NOSE SECTION FROM
FULL-SCALE STATION 230, 0 TO STATION 500 FROM NAR DRAWING VL70-000139B.
REMAINING BODY AFT OF STATION 500 FROM NAR VL70-000093.

MODEL SCALE 0.0165

DRAWING NUMBER : VL70-000093, VL70-000139B

DIMENSIONS :	FULL SCALE	MODEL SCALE
Length	1290.0	21.200 IN.
Max Width	265.0	4.372 IN.
Max Depth	248.0	4.092 IN.
Fineness Ratio	4.069	4.862 IN.
Area	15.4000 SQ. FT.	17.8927 SQ. FT.
Max. Cross-Sectional		
Planform		
Wetted		
Base		

TABLE III (Continued)
MODEL DIMENSIONAL DATA

MODEL COMPONENT : BODY FLAP-F1

GENERAL DESCRIPTION : O90B-139

MODEL SCALE: 0.0165

DRAWING NUMBER : VL70-000024A

DIMENSIONS :	FULL SCALE	MODEL SCALE
Length	<u>.84.700</u>	<u>1.398</u>
Max Width	<u>265.000</u>	<u>4.372</u>
Max Depth	<u>21.000</u>	<u>.346</u>
Fineness Ratio	<u>—</u>	<u>—</u>
Area	<u>—</u>	<u>—</u>
Max. Cross-Sectional	<u>—</u>	<u>—</u>
Planform	<u>142.6400</u>	<u>5.5921</u>
Wetted	<u>—</u>	<u>—</u>
Base	<u>38.0460</u>	<u>1.5151</u>

TABLE III (Continued)
MODEL DIMENSIONAL DATA

MODEL COMPONENT : OMS PODS - M¹⁴

GENERAL DESCRIPTION 032B-132

MODEL SCALE: 0.0165

DRAWING NUMBER : VI70-000094

DIMENSIONS :

FULL SCALE MODEL SCALE

Length	<u>346.000</u>	<u>5.709</u>
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Max Width	<u>108.000</u>	<u>1.782</u>
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Max Depth	<u>113.800</u>	<u>1.878</u>
-----------	----------------	--------------

Fineness Ratio	<u> </u>	<u> </u>
----------------	-------------------------	-------------------------

Area	<u> </u>	<u> </u>
------	-------------------------	-------------------------

Max. Cross-Sectional	<u> </u>	<u> </u>
----------------------	-------------------------	-------------------------

Planform	<u> </u>	<u> </u>
----------	-------------------------	-------------------------

Wetted	<u> </u>	<u> </u>
--------	-------------------------	-------------------------

Base	<u> </u>	<u> </u>
------	-------------------------	-------------------------

ONE POD CONTINUATION

2 Axis Orientation	<u>113.000</u>	<u>7.451</u>
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1 Axis Rotation	<u>10.000</u>	<u>1.320</u>
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TABLE III (Continued)
MODEL DIMENSIONAL DATA

COMPONENT SLOTTED CLEVON (1-inch GAP) - D-3
PAL DESCRIPTION Configuration 140A/D Orbiter clevron.
NOTE: D-3 is a slotted version of "26. Data are for one side.
MODEL SCALE: 0.0165 MODEL DRAWING: 113-A00110
DRAWING NUMBER _____

DIMENSIONS	FULL SCALE	MODEL SCALE
Area - Ft^2	010.0	0.0572
Span (equivalent) - In.	340.2	5.762
Inb'd equivalent chord - In.	110.00 ^b	1.947
Outb'd equivalent chord/ total surface chord	55.19 ^a	0.910 ^b
Ratio movable surface chord/ total surface chord	_____	_____
At Inb'd equiv. chord	0.2026	0.2305
At Outb'd equiv. chord	0.400 ^b	0.440 ^b
Sweep Back Angles, degrees	_____	_____
Leading Edge	0.00	0.00
Trailing Edge	-10.05 ^c	-10.05 ^c
Hingeline	0.00	0.00
Area Moment (Normal to hinge line)	1507.0 ^c	0.0013 ^c
Mean Aerodynamic Chord (\bar{c}), in.	21.7	1.401 ^c

TABLE III (Continued)
MODEL DIMENSIONAL DATA

MODEL COMPONENT WING - R5

GENERAL DESCRIPTION CONFIGURATION FWD LINES VIZZ-000005.

MODEL SCALE: 0.015

DRAWING NUMBER VIZZ-000005

DIMENSIONS	FULL SCALE	MODEL SCALE
Area	<u>106.380 SQ. IN.</u>	<u>.0290 SQ. FT.</u>
Span (equivalent)	<u>201.00 IN.</u>	<u>3.31 IN.</u>
Inb'd equivalent chord	<u>.21.585 IN.</u>	<u>1.51 IN.</u>
Outb'd equivalent chord	<u>.50.933 IN.</u>	<u>.34 IN.</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	<u>.400</u>	<u>.400</u>
At Outb'd equiv. chord	<u>.400</u>	<u>.400</u>
Sweep Back Angles, degrees		
Leading Edge	<u>34.93</u>	<u>34.93 deg.</u>
Trailing Edge	<u>26.25</u>	<u>26.25 deg.</u>
Hingeline	<u>34.83</u>	<u>34.83 deg.</u>
Area Moment (Normal to hinge line)	<u>506.1250 IN. .IN.</u>	<u>4.0840 CU. FT.</u>

TABLE III (Continued)
MODEL DIMENSIONAL DATA

MODEL COMPONENT VERTICAL TAIL - V₅

GENERAL DESCRIPTION CENTERLINE VERTICAL TAIL DOUBLE WEDGE AIRFOIL
WITH ROUNDED LEADING EDGE.

MODEL SCALE: 0.0165

DRAWING NUMBER V170-000025

DIMENSIONS	FULL SCALE	MODEL SCALE
Area	<u>413.2500 SQ.FT.</u>	<u>16.2011 SQ.IN.</u>
Span (equivalent)	<u>315.72</u>	<u>5.21</u> IN.
Inb'd equivalent chord	<u>268.50</u> IN.	<u>4.43</u> IN.
Outb'd equivalent chord	<u>103.47</u> IN.	<u>1.79</u> IN.
Ratio movable surface chord/ total surface chord	—	—
At Inb'd equiv. chord	—	—
At Outb'd equiv. chord	—	—
Sweep Back Angles, degrees	—	—
Leading Edge	<u>45.00</u> DEG.	<u>45.00</u> DEG.
Trailing Edge	<u>26.240</u> DEG.	<u>26.25</u> DEG.
Hingeline	—	—
Area Moment (Normal to hinge line)	—	—

TA 52 TII (Conv. Conv.)

MODEL COMPONENT: WING - "C"

GENERAL DESCRIPTION: CONVENTIONAL PLANE WITH TAPERED WINGS

DEFINING WING LOC. IN CHORD: EXPOSING AT THE MAC. (THE CHORD LENGTH IS 1.00).

FIRST 0.25 OF PLANE IS TAILLESS. TO THE FUSELAGE (WINGSPAN 1.00).

TOWING CRATE: - 0.015

DRAWING NUMBER: V170-000003

DIMENSIONS: FULL-SCALE MODEL SCALETOTAL DATA

Area	FULL-SCALE	MODEL SCALE
Planform	2630.000	105.4510
Wetted	926.600	35.0600
Span (equivalent)	6.000	0.250
Aspect Ratio	1.17	1.17
Rate of Taper	.700	.700
Taper Ratio	.500	.500
Dihedral Angle, degrees	3.500	2.500
Incidence Angle, degrees	-1.000	-0.00
Aerodynamic Twist, degrees	-0.000	-0.00
Toe-In Angle		
Cant Angle		
Sweep Back Angles, degrees		
Leading Edge	45.000	45.000
Trailing Edge	-10.000	-10.000
0.25 Element Line	15.000	15.000
Chords:		
Root (Wing Sta. 0.0)	180.00	11.00
Tip, (equivalent)	130.00	8.00
MAC	110.00	7.00
Fus. Sta. of .25 MAC	110.00	7.00
W.P. of .25 MAC	100.00	6.00
B.L. of .25 MAC	100.00	6.00
Airfoil Section		
Root		
Tip		

EXPOSED DATA

Area	FULL-SCALE	MODEL SCALE
Span, (equivalent)	1700.000	70.0000
Aspect Ratio	1.17	1.17
Taper Ratio	.700	.700
Chords		
Root	52.00	3.00
Tip	17.00	1.00
MAC	11.00	0.70
Fus. Sta. of .25 MAC	11.00	0.70
W.P. of .25 MAC	10.00	0.60
B.L. of .25 MAC	10.00	0.60

Notes:

1. Positive directions of force coefficients, moment coefficients, and angles are indicated by arrows
2. For clarity, origins of wind and stability axes have been displaced from the center of gravity

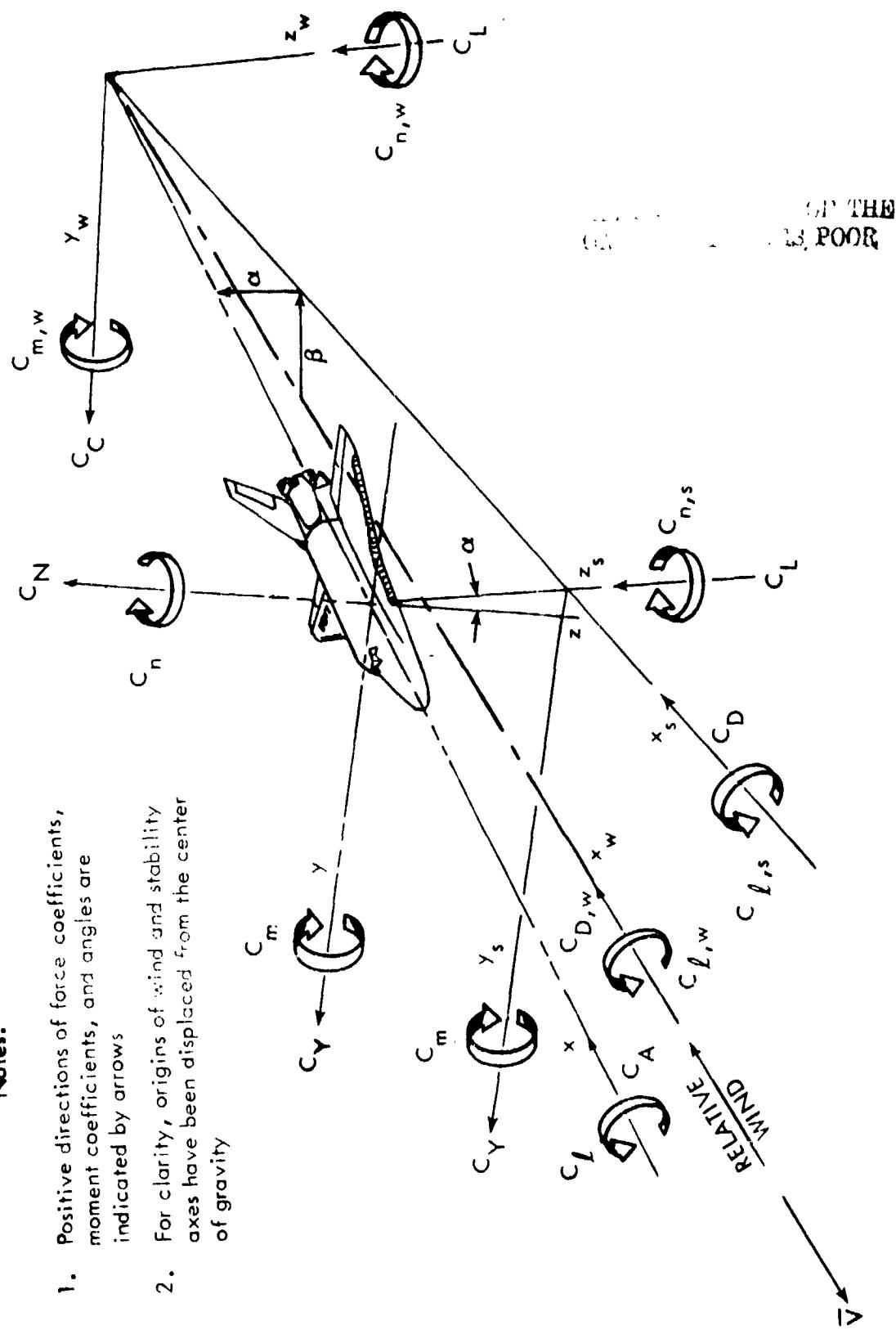
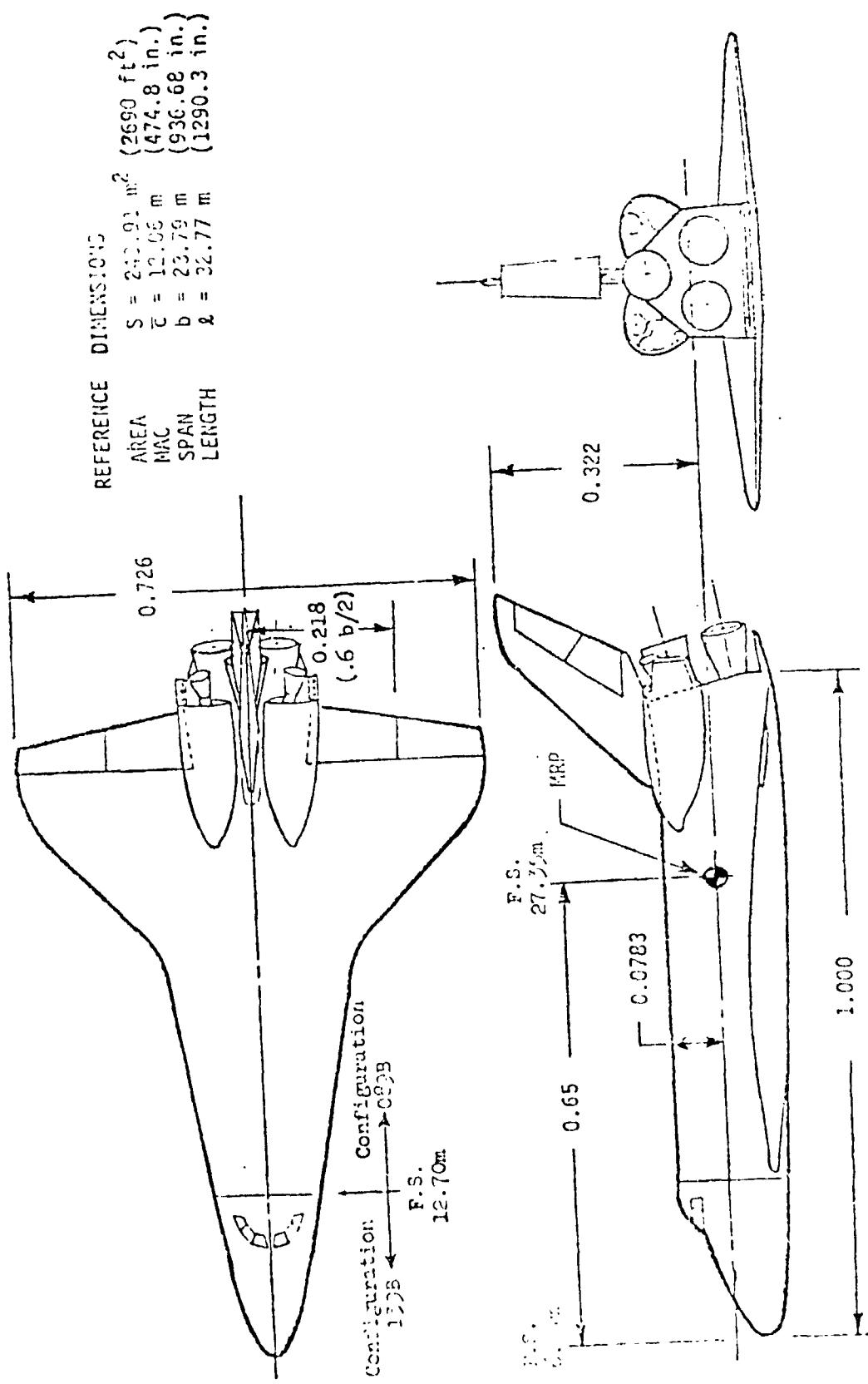
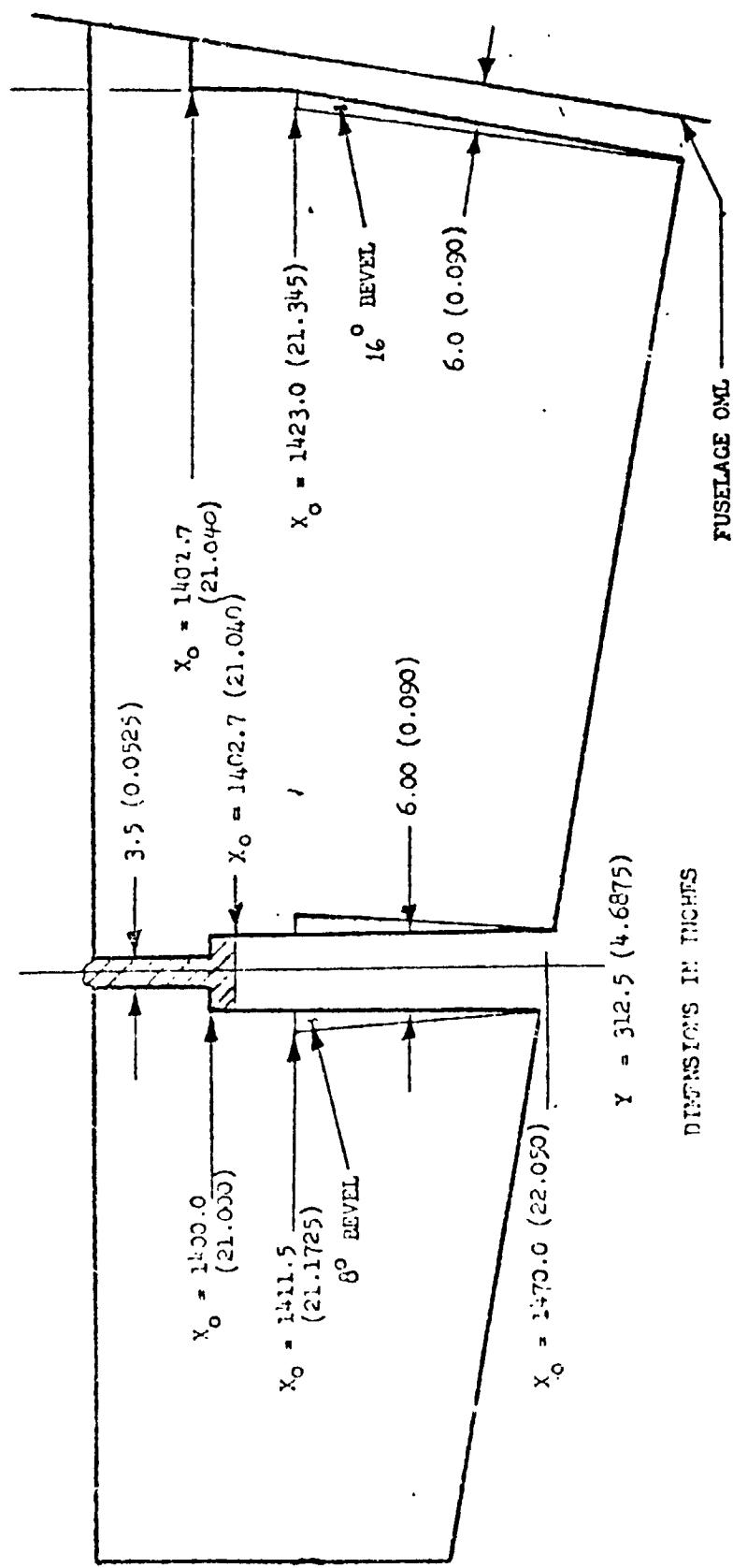


FIGURE 1. - AXIS SYSTEMS



a. SSV Orbiter Configuration
Figure 2. - Model Sketches

$Y = 128.50$ (1.928)



b. Slotted Elevon E43 (6-inch gap)
Figure 2. - Concluded.



a. Orbiter Configuration, Front, 3/4 View
Figure 3. - Model Photographs

REPRODUCED FROM THE
ORIGINAL AS RECEIVED



b. Orbiter Configuration, Rear, 3/4 View
Figure 3. - Continued



c. View of Elevon Drive Motors
Figure 3. - Concluded

DATA FIGURES

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ELV-L0	ELV-L1	ELV-R1	ELV-R0
R4001	LA-19 SPW 1101 RI-0898/139 CRB SP-LT ELEVON	.000	.000	.000	.000
R4005	LA-19 SPW 1101 RI-0898/139 CRB SP-LT ELEVON	-10.000	-10.000	-10.000	-10.000
R4006	LA-19 SPW 1101 RI-0898/139 CRB SP-LT ELEVON	-20.000	-20.000	-20.000	-20.000
R4007	LA-19 SPW 1101 RI-0898/139 CRB SP-LT ELEVON	-40.000	-40.000	-40.000	-40.000

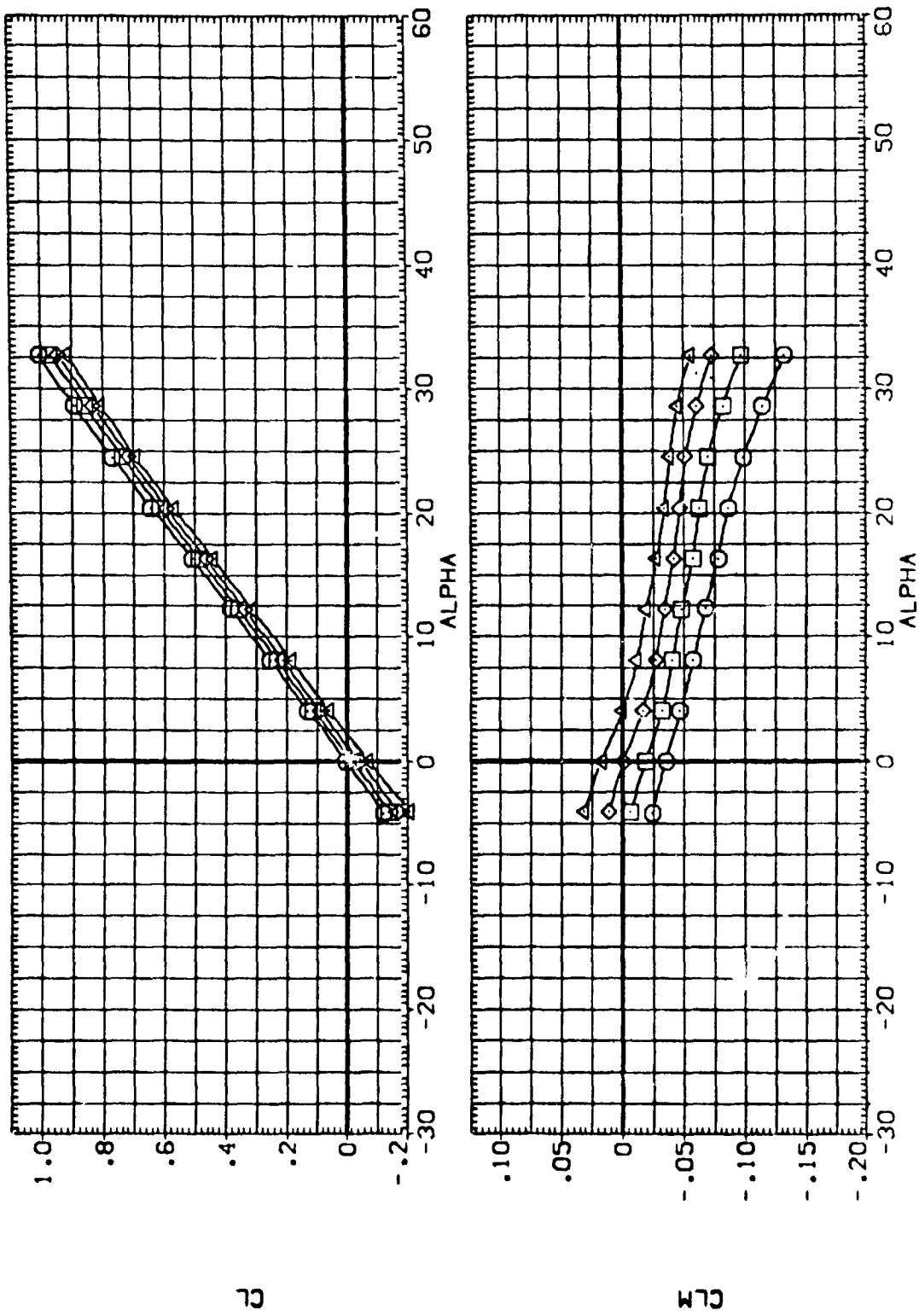
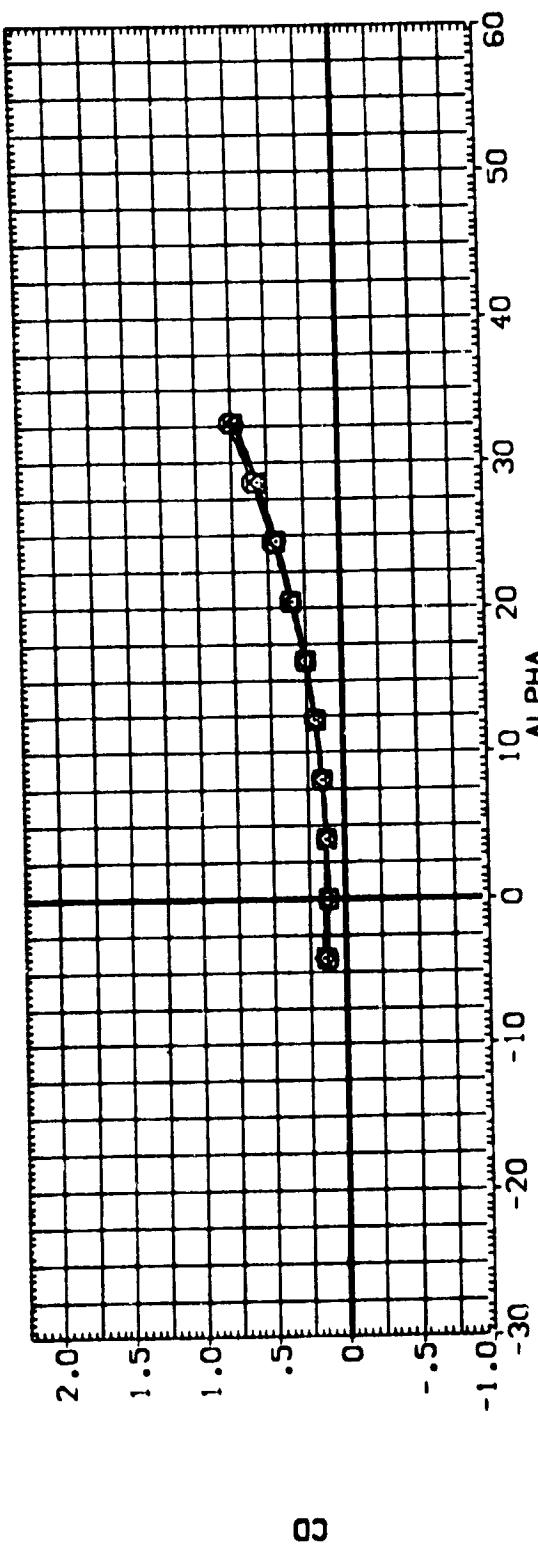


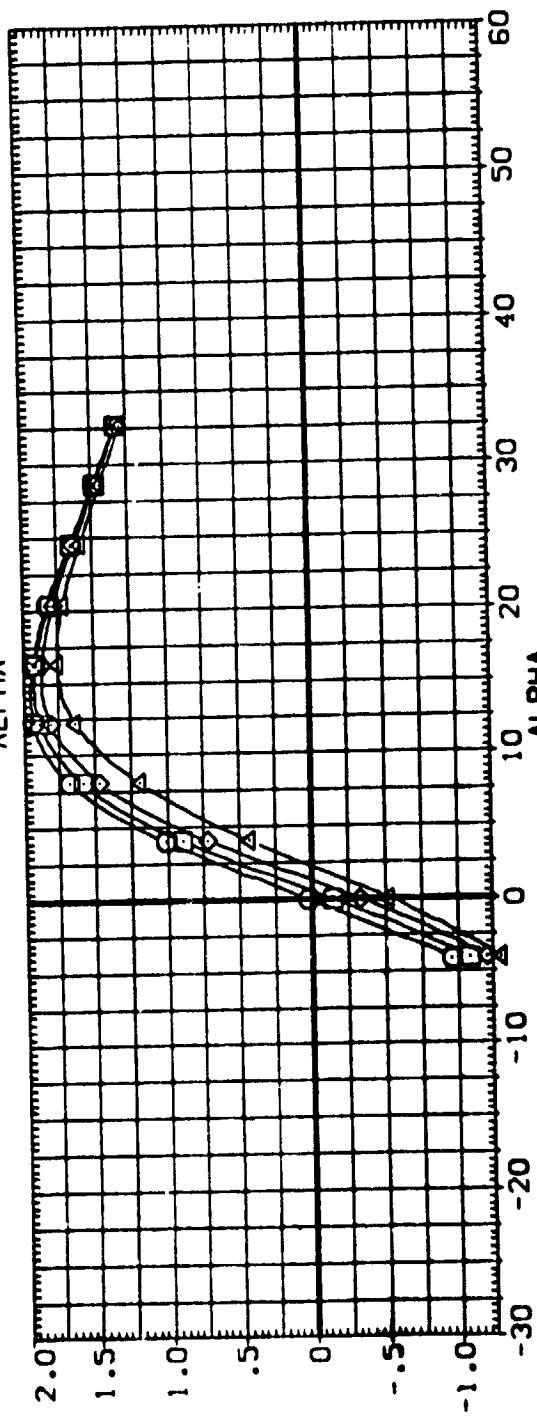
FIGURE 4. FULL SPAN ELEVON PITCH CHARACTERISTICS
 $(A)_WACH = 2.50$

DATA SET SYMBOL CONFIGURATION DESCRIPTION

	LA-49 SPN	101 RI-0888/139 088 SPLIT ELEVON	ELV-L0	ELV-L1	ELV-RI	ELV-R0
Q	LA-49 SPN	101 RI-0888/139 088 SPLIT ELEVON	-10.000	-10.000	-10.000	-10.000
X	LA-49 SPN	101 RI-0888/139 088 SPLIT ELEVON	-20.000	-20.000	-20.000	-20.000
+	LA-49 SPN	101 RI-0888/139 088 SPLIT ELEVON	-40.000	-40.000	-40.000	-40.000
*	LA-49 SPN	101 RI-0888/139 088 SPLIT ELEVON				



C_L



C_L

FIGURE 4. FULL SPAN ELEVON PITCH CHARACTERISTICS
 $(\lambda)_{MACH} = 2.50$

DATA SET SOURCE CONFIGURATION DESCRIPTION ELEV-LI ELEV-RI ELEV-RD

LA-49 SPAN 1100 R-0898/139 068 SPAN ELEV66	LA-49 SPAN 1100 R-0898/139 068 SPAN ELEV66	-10.000 -10.000 -10.000
LA-49 SPAN 1100 R-0898/139 068 SPAN ELEV66	LA-49 SPAN 1100 R-0898/139 068 SPAN ELEV66	-20.000 -20.000 -20.000
LA-49 SPAN 1100 R-0898/139 068 SPAN ELEV66	LA-49 SPAN 1100 R-0898/139 068 SPAN ELEV66	-40.000 -40.000 -40.000

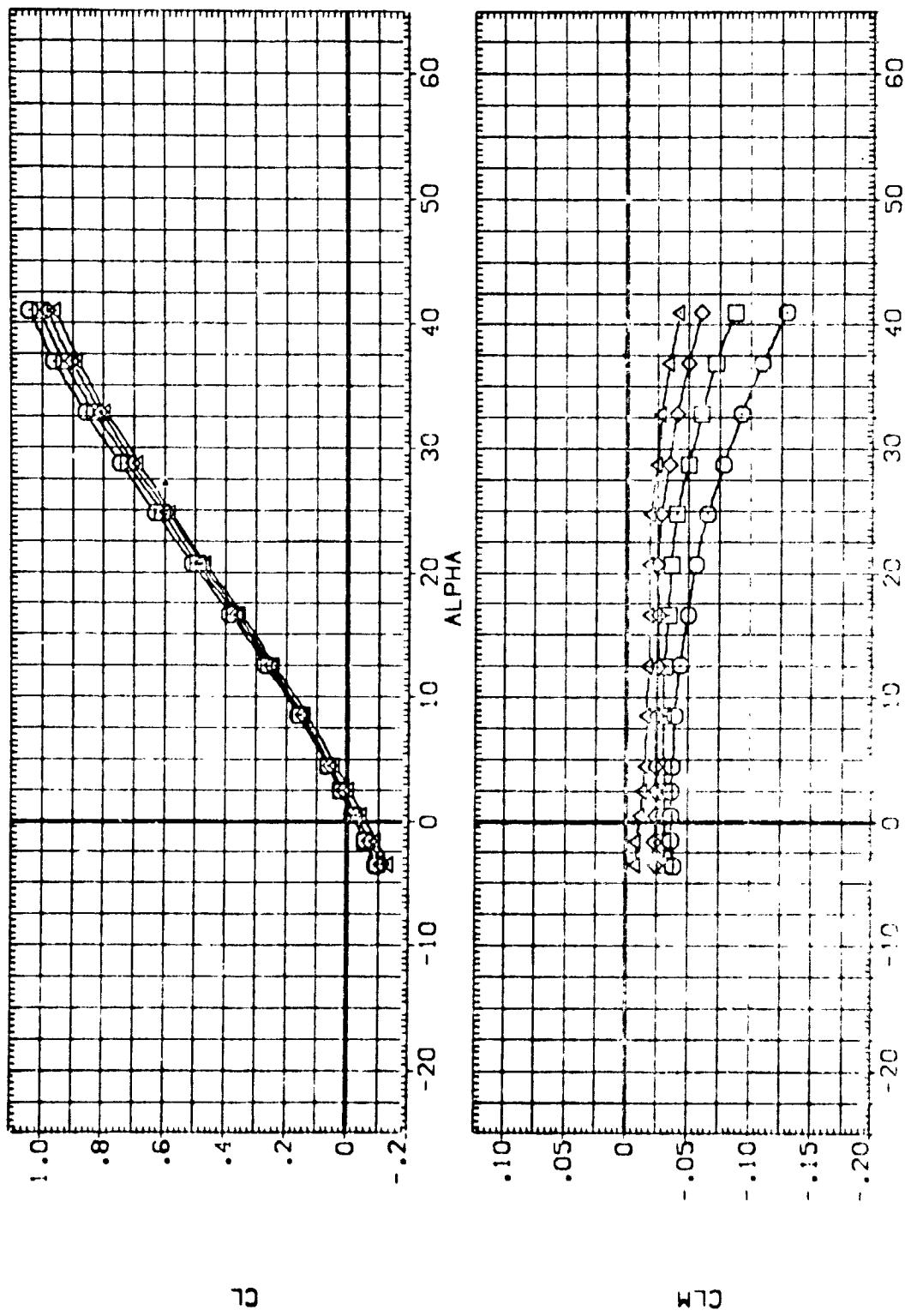


FIGURE 4. FULL SPAN ELEVON PITCH CHARACTERISTICS

(B)_{MACH} = 4.60

PAGE 3

DATA SET 3000 CONFIGURATION DESCRIPTION ELEVON
 LA-49 SPN 110 RI-0698/139 089 SPN 111 ELEVON
 LA-49 SPN 110 RI-0698/139 098 SPN 111 ELEVON
 LA-49 SPN 110 RI-0698/139 098 SPN 111 ELEVON
 LA-49 SPN 110 RI-0698/139 088 SPN 111 ELEVON

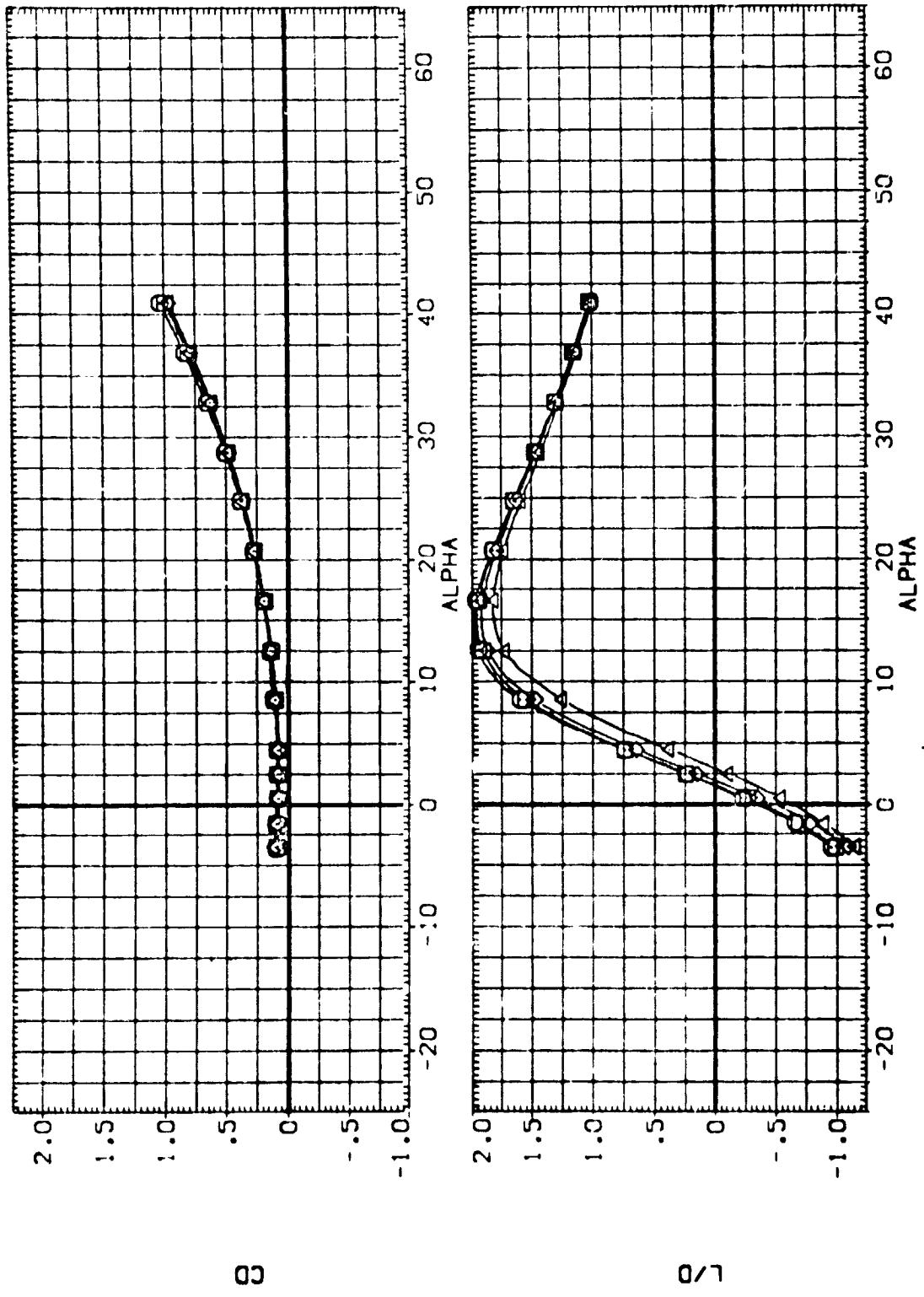


FIGURE 4. FULL SPAN ELEVON PITCH CHARACTERISTICS
 $(B)_MACH = 4.60$

DATA SET SYMBOL	CONFIGURATION	DESCRIPTION	ELEV-L0	ELEV-L1	ELEV-R0	ELEV-R1
R-3001	L-49 SP	R1-0038/39 SP	.000	-40.000	-40.000	.000
R-3002	L-49 SP	R1-0038/39 SP	.000	-20.000	-20.000	.000
R-3003	L-49 SP	R1-0038/39 SP	.000	-10.000	-10.000	.000
R-3004	L-49 SP	R1-0038/39 SP	.000	.000	.000	.000

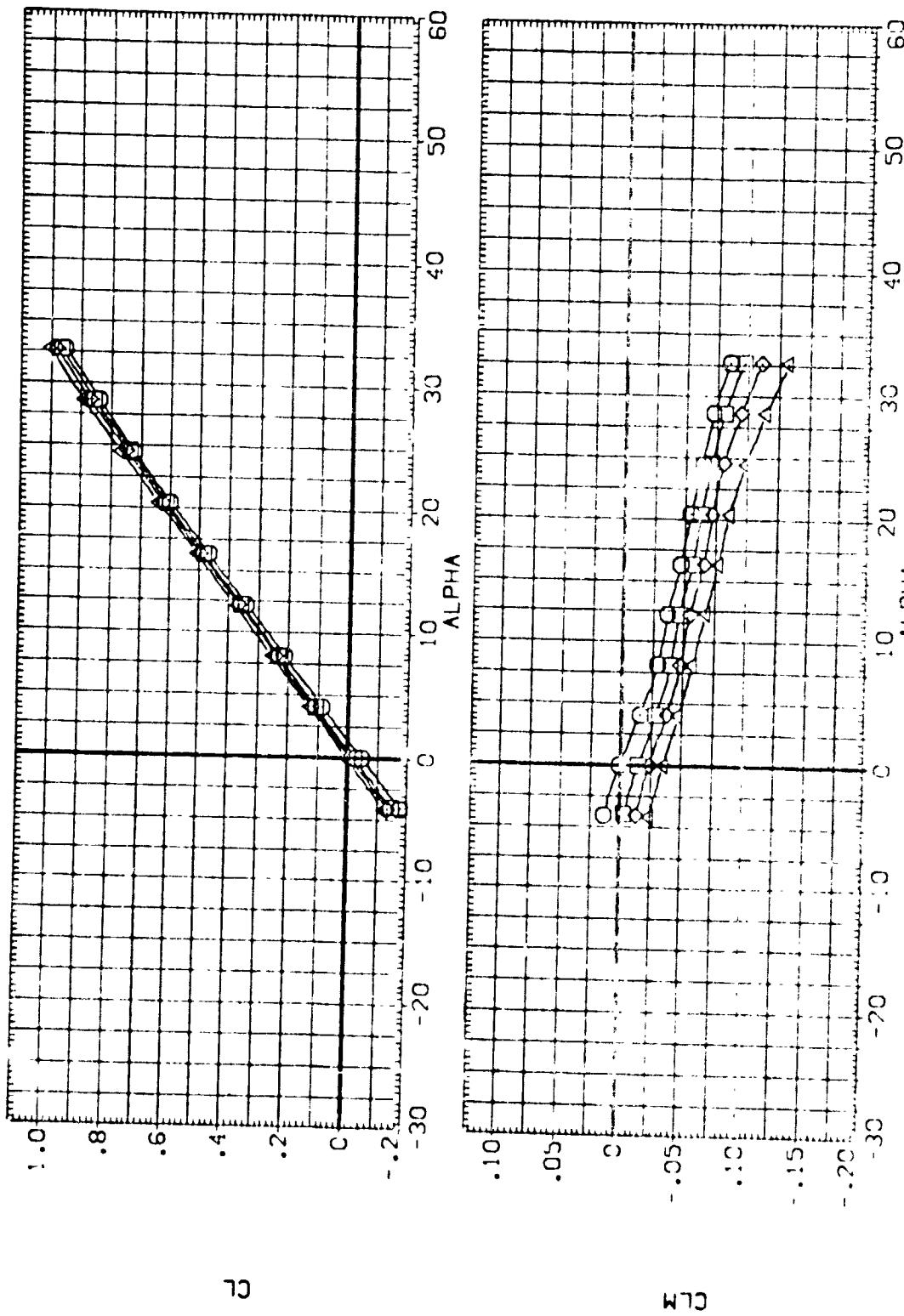


FIGURE 5. INBOARD ELEVON PITCH CHARACTERISTICS
CLMACH = 2.5C

REPRODUCIBILITY OF THE
ORIGINAL PLOT IS POOR

DATA SET SWBQ CONFIGURATION DESCRIPTION
A-19 UP RI-0688/138 688 SP.11 ELEVON
A-19 SP RI-0688/138 688 SP.11 ELEVON

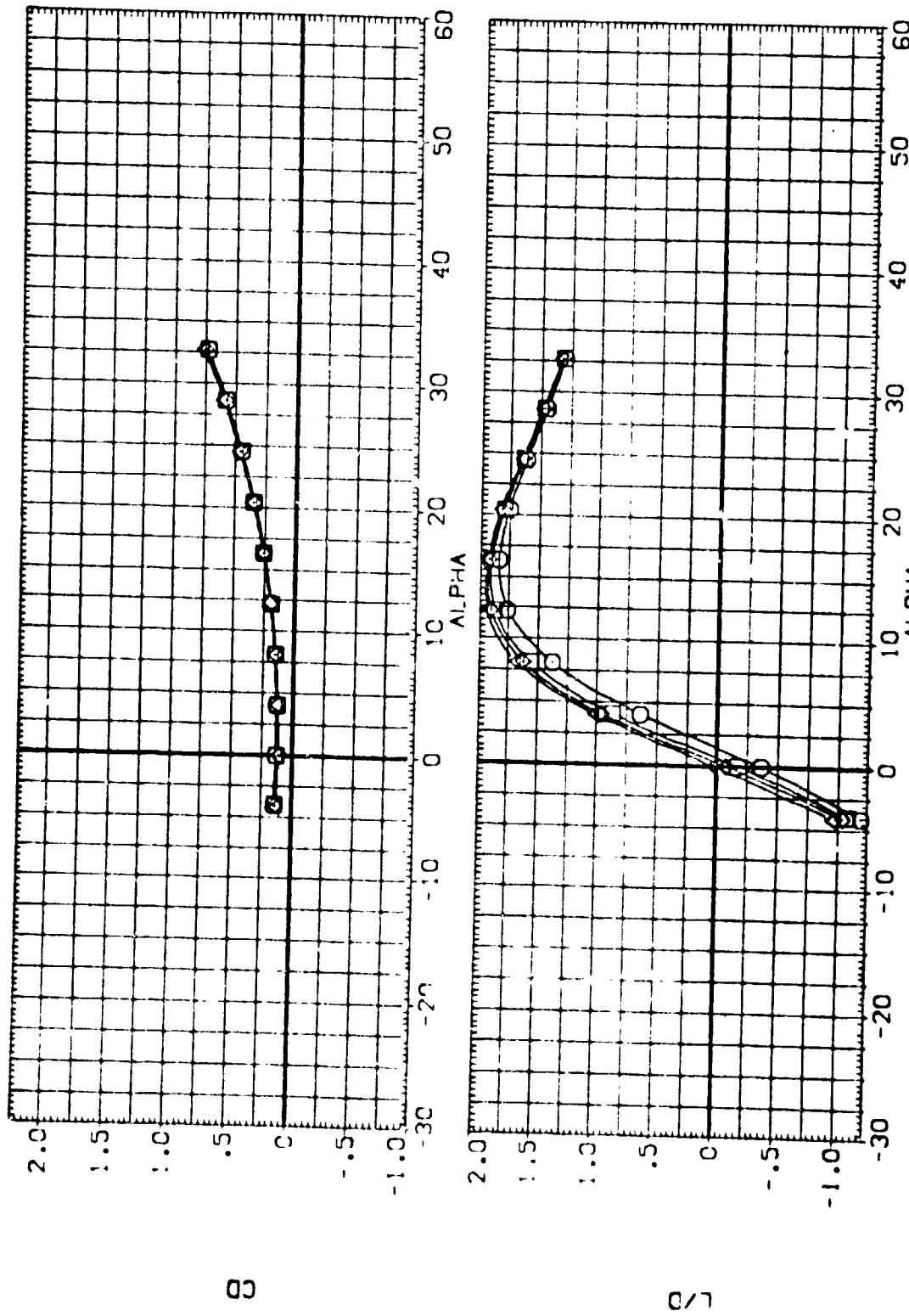


FIGURE 5. INBOARD ELEVON PITCH CHARACTERISTICS
($A_{MACH} = 2.50$)

DATA SET SPIN IGNATION DESCRIPTION
 1.00001 R1-0898/35 0898 SPIN ELEVON
 0.98931 R1-0898/35 0898 SPIN ELEVON
 0.97861 R1-0898/35 0898 SPIN ELEVON
 0.96781 R1-0898/35 0898 SPIN ELEVON
 0.95701 R1-0898/35 0898 SPIN ELEVON
 0.94621 R1-0898/35 0898 SPIN ELEVON
 0.93541 R1-0898/35 0898 SPIN ELEVON
 0.92461 R1-0898/35 0898 SPIN ELEVON
 0.91381 R1-0898/35 0898 SPIN ELEVON
 0.90301 R1-0898/35 0898 SPIN ELEVON
 0.89221 R1-0898/35 0898 SPIN ELEVON
 0.88141 R1-0898/35 0898 SPIN ELEVON
 0.87061 R1-0898/35 0898 SPIN ELEVON
 0.85981 R1-0898/35 0898 SPIN ELEVON
 0.84901 R1-0898/35 0898 SPIN ELEVON
 0.83821 R1-0898/35 0898 SPIN ELEVON
 0.82741 R1-0898/35 0898 SPIN ELEVON
 0.81661 R1-0898/35 0898 SPIN ELEVON
 0.80581 R1-0898/35 0898 SPIN ELEVON
 0.79501 R1-0898/35 0898 SPIN ELEVON
 0.78421 R1-0898/35 0898 SPIN ELEVON
 0.77341 R1-0898/35 0898 SPIN ELEVON
 0.76261 R1-0898/35 0898 SPIN ELEVON
 0.75181 R1-0898/35 0898 SPIN ELEVON
 0.74101 R1-0898/35 0898 SPIN ELEVON
 0.73021 R1-0898/35 0898 SPIN ELEVON
 0.71941 R1-0898/35 0898 SPIN ELEVON
 0.70861 R1-0898/35 0898 SPIN ELEVON
 0.69781 R1-0898/35 0898 SPIN ELEVON
 0.68701 R1-0898/35 0898 SPIN ELEVON
 0.67621 R1-0898/35 0898 SPIN ELEVON
 0.66541 R1-0898/35 0898 SPIN ELEVON
 0.65461 R1-0898/35 0898 SPIN ELEVON
 0.64381 R1-0898/35 0898 SPIN ELEVON
 0.63301 R1-0898/35 0898 SPIN ELEVON
 0.62221 R1-0898/35 0898 SPIN ELEVON
 0.61141 R1-0898/35 0898 SPIN ELEVON
 0.60061 R1-0898/35 0898 SPIN ELEVON
 0.58981 R1-0898/35 0898 SPIN ELEVON
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 0.56821 R1-0898/35 0898 SPIN ELEVON
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 0.53581 R1-0898/35 0898 SPIN ELEVON
 0.52501 R1-0898/35 0898 SPIN ELEVON
 0.51421 R1-0898/35 0898 SPIN ELEVON
 0.50341 R1-0898/35 0898 SPIN ELEVON
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 0.48181 R1-0898/35 0898 SPIN ELEVON
 0.47101 R1-0898/35 0898 SPIN ELEVON
 0.46021 R1-0898/35 0898 SPIN ELEVON
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 0.37381 R1-0898/35 0898 SPIN ELEVON
 0.36301 R1-0898/35 0898 SPIN ELEVON
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 0.33061 R1-0898/35 0898 SPIN ELEVON
 0.31981 R1-0898/35 0898 SPIN ELEVON
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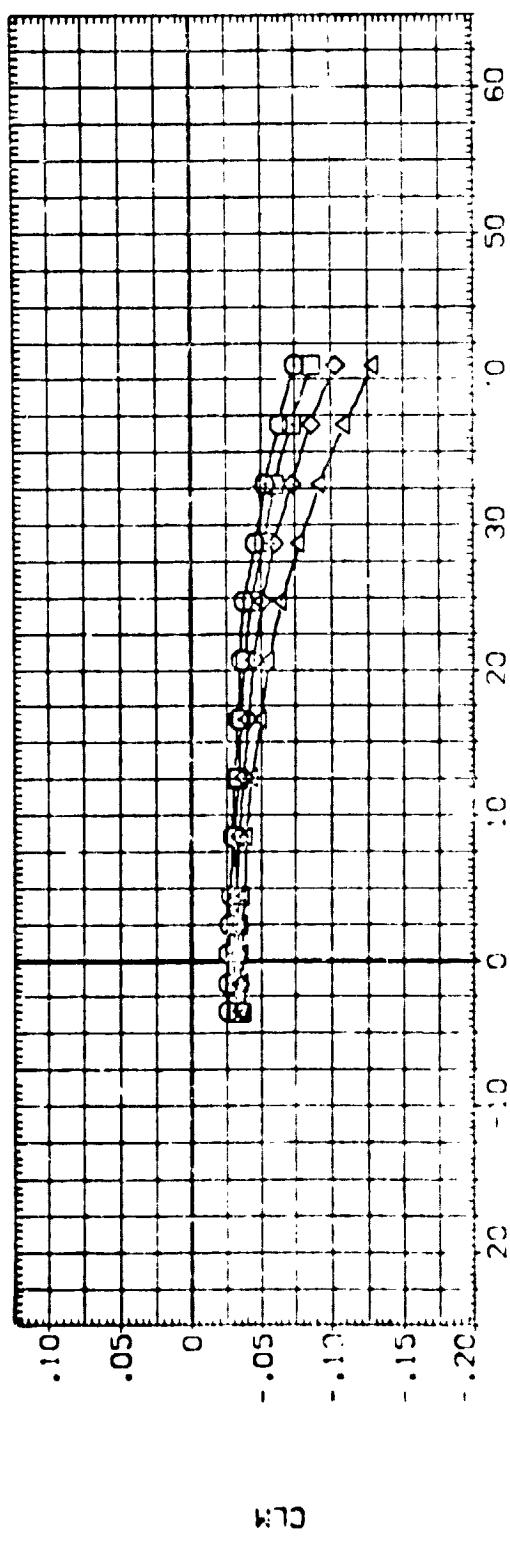
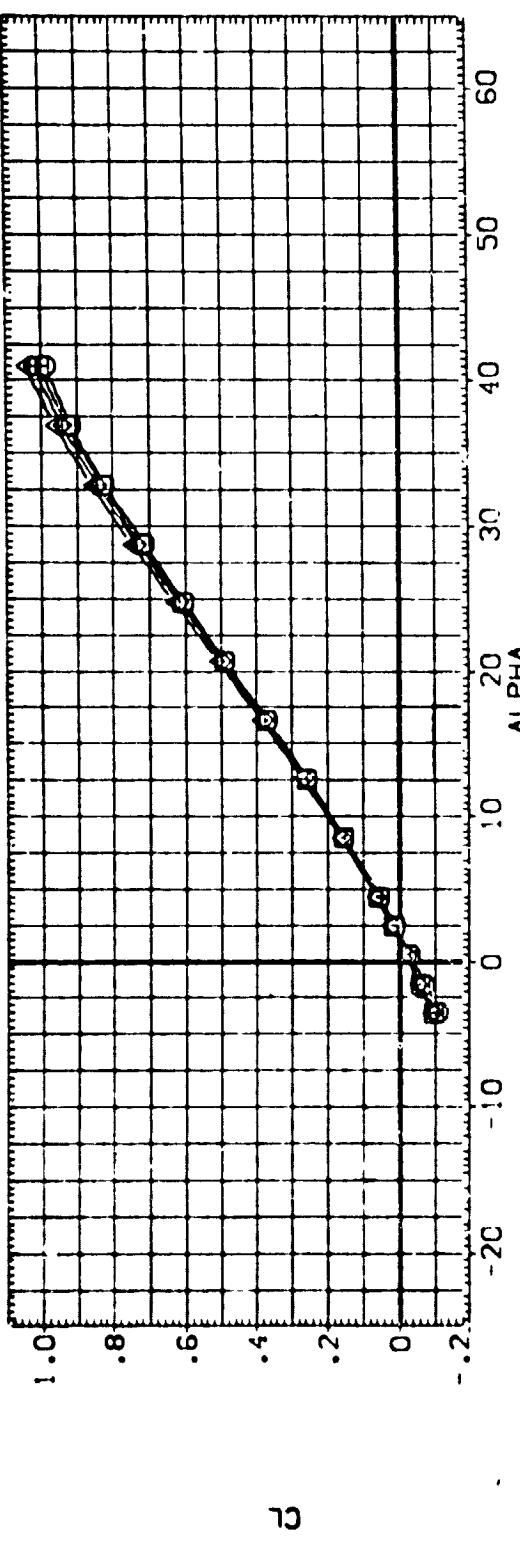


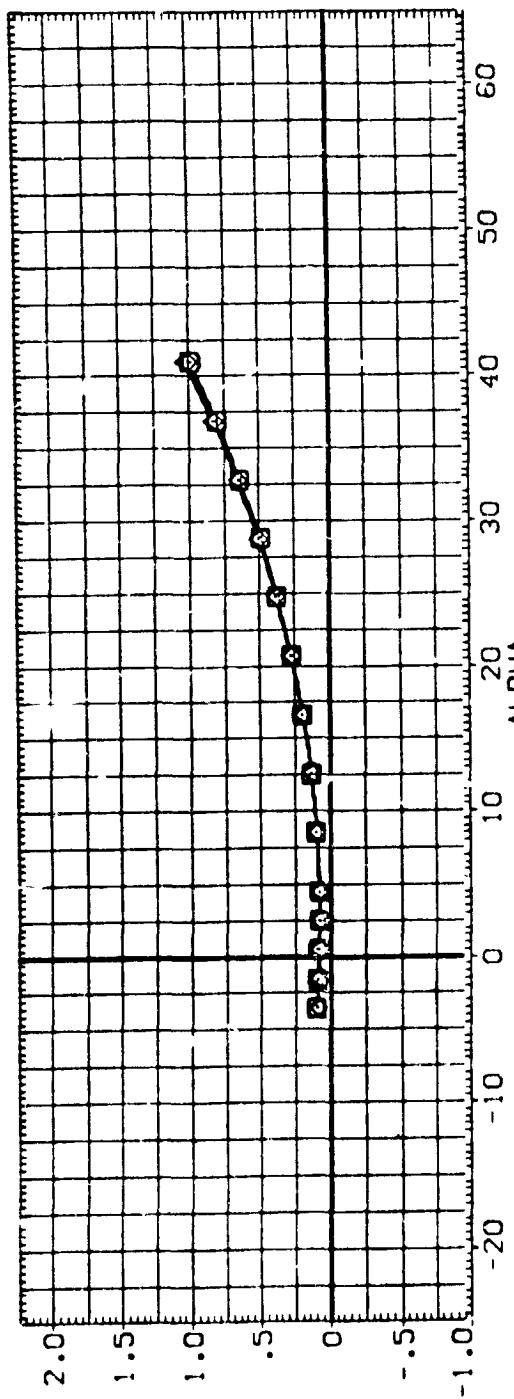
FIGURE 5. INBOARD ELEVON PITCH CHARACTERISTICS

$$C_{MACH} = 4.60$$

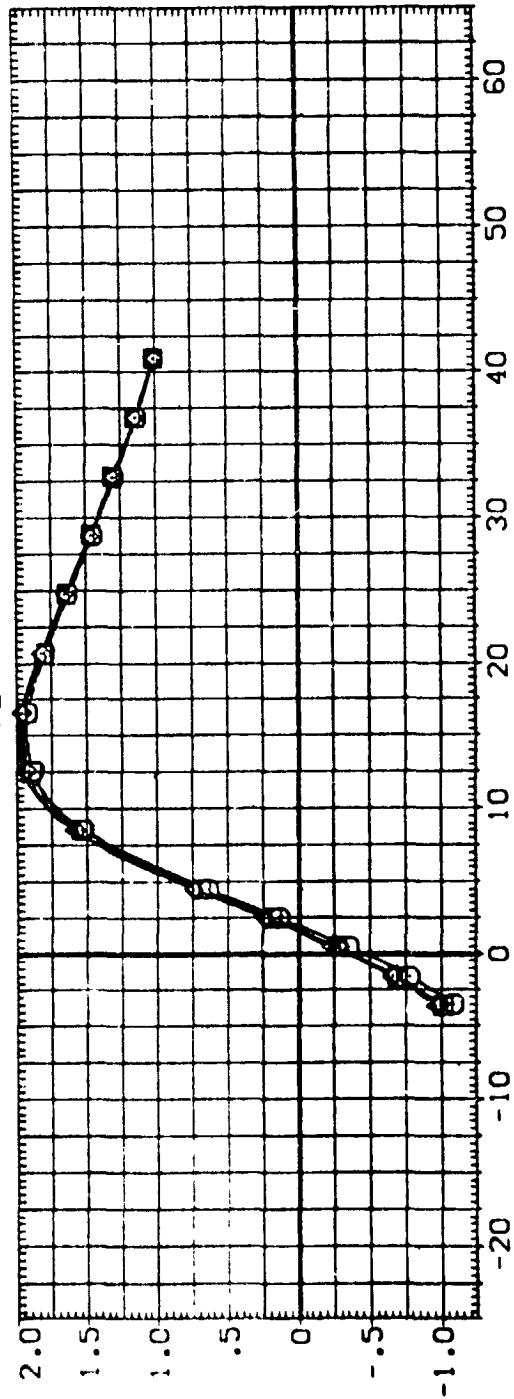
PAGE 7.

DATA SET SYMBOL CONFIGURATION DESCRIPTION

	ELV-LG	ELV-LI	ELV-RI	ELV-RD
RW004	101	RI-0888/139	088 SPLIT ELEVON	.000 -10.000 -40.000 .000
RW003	101	RI-0888/139	088 SPLIT ELEVON	.000 -20.000 -20.000 .000
RW002	101	RI-0888/139	088 SPLIT ELEVON	.000 -10.000 -10.000 .000
RW001	101	RI-0888/139	088 SPLIT ELEVON	.000 .000 .000 .000



CD



LD

FIGURE 5. INBOARD ELEVON PITCH CHARACTERISTICS
(B)_{MACH} = 4.60

PAGE 8

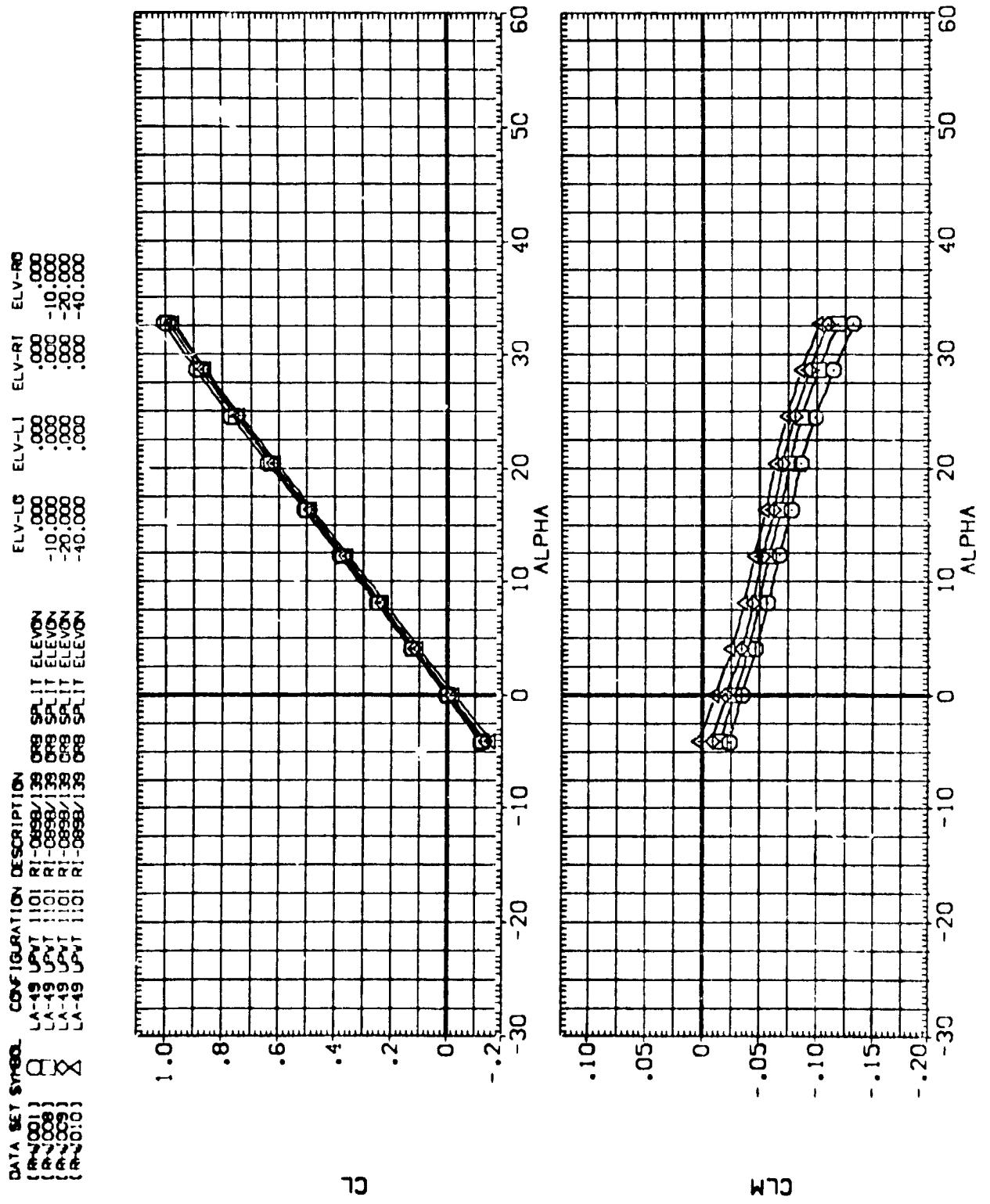


FIGURE 6. OUTBOARD ELEVON PITCH CHARACTERISTICS

$(\lambda)_{MACH} = 2.50$

DATA SET SYMBOL CONFIGURATION DESCRIPTION

	ELV-L0	ELV-R1	ELV-R0
LA-149 LIFT	.101	R1-.0898/.139	088 SPLIT ELEVON
LA-149 RSVT	.101	R1-.0898/.139	088 SPLIT ELEVON
LA-149 SPT	.101	R1-.0898/.139	088 SPLIT ELEVON
LA-149 SPVT	.101	R1-.0898/.139	088 SPLIT ELEVON
LA-149 SPWT	.101	R1-.0898/.139	088 SPLIT ELEVON

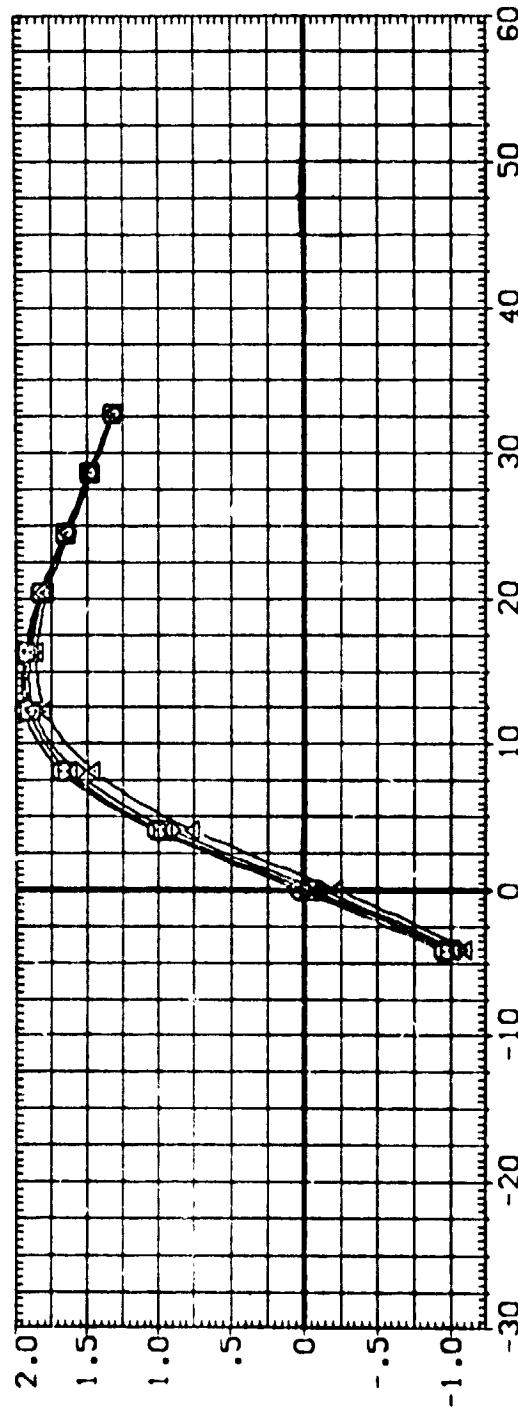
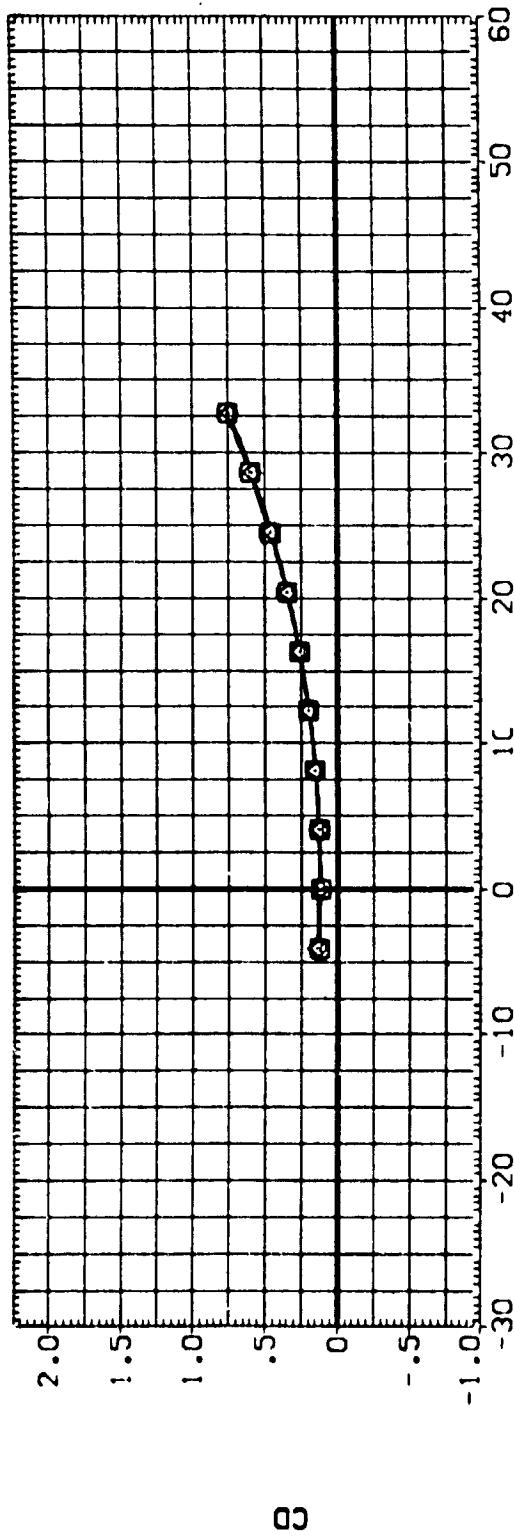


FIGURE 6. OUTBOARD ELEVON PITCH CHARACTERISTICS
 $MACH = 2.50$

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ELV-LD	ELV-L1	ELV-RI	ELV-R0
[RJU001]	LA-49 UPNT	1101	R1-0898/135	088 SPLIT ELEVON	.000 .000
[RJU008]	LA-49 UPNT	1101	R1-0898/135	088 SPLIT ELEVON	.000 .000
[RJU009]	LA-49 UPNT	1101	R1-0898/135	088 SPLIT ELEVON	.000 .000
[RJU010]	LA-49 UPNT	1101	R1-0898/135	088 SPLIT ELEVON	.000 .000

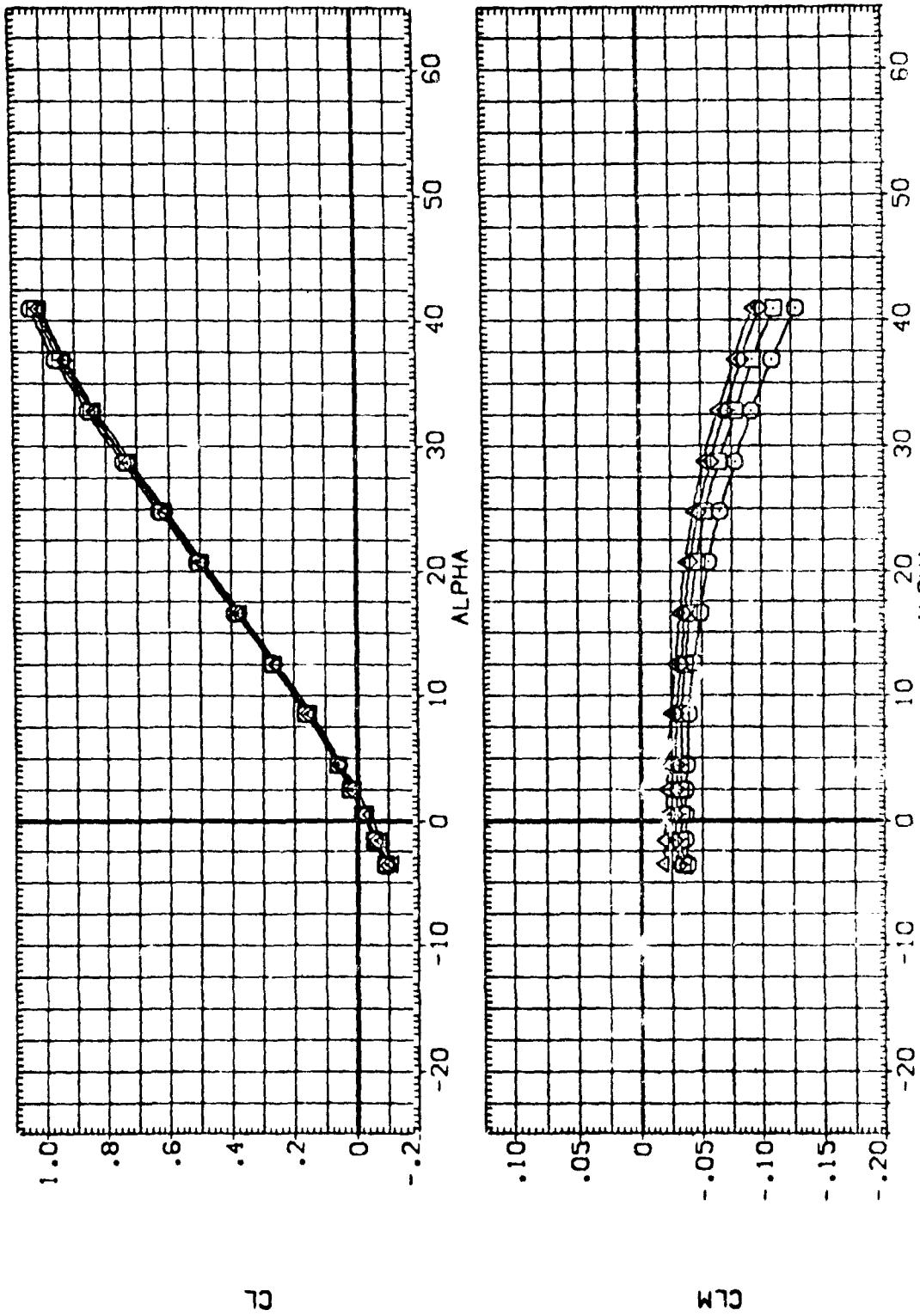


FIGURE 6. OUTBOARD ELEVON PITCH CHARACTERISTICS

(BOWMACH = 4.60)

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ELV-L0	ELV-L1	ELV-R1	ELV-R0
(1000)	LA-19 UNI R1-0898/138	.000	.000	.000	.000
(4700)	LA-19 R1-0898/138	.000	.000	.000	-10.000
(4701)	LA-19 SSVT R1-0898/138	.000	.000	.000	-20.000
(4702)	LA-19 SSVT R1-0898/138	.000	.000	.000	-40.000
(4703)	LA-19 SSVT R1-0898/138	.000	.000	.000	.000

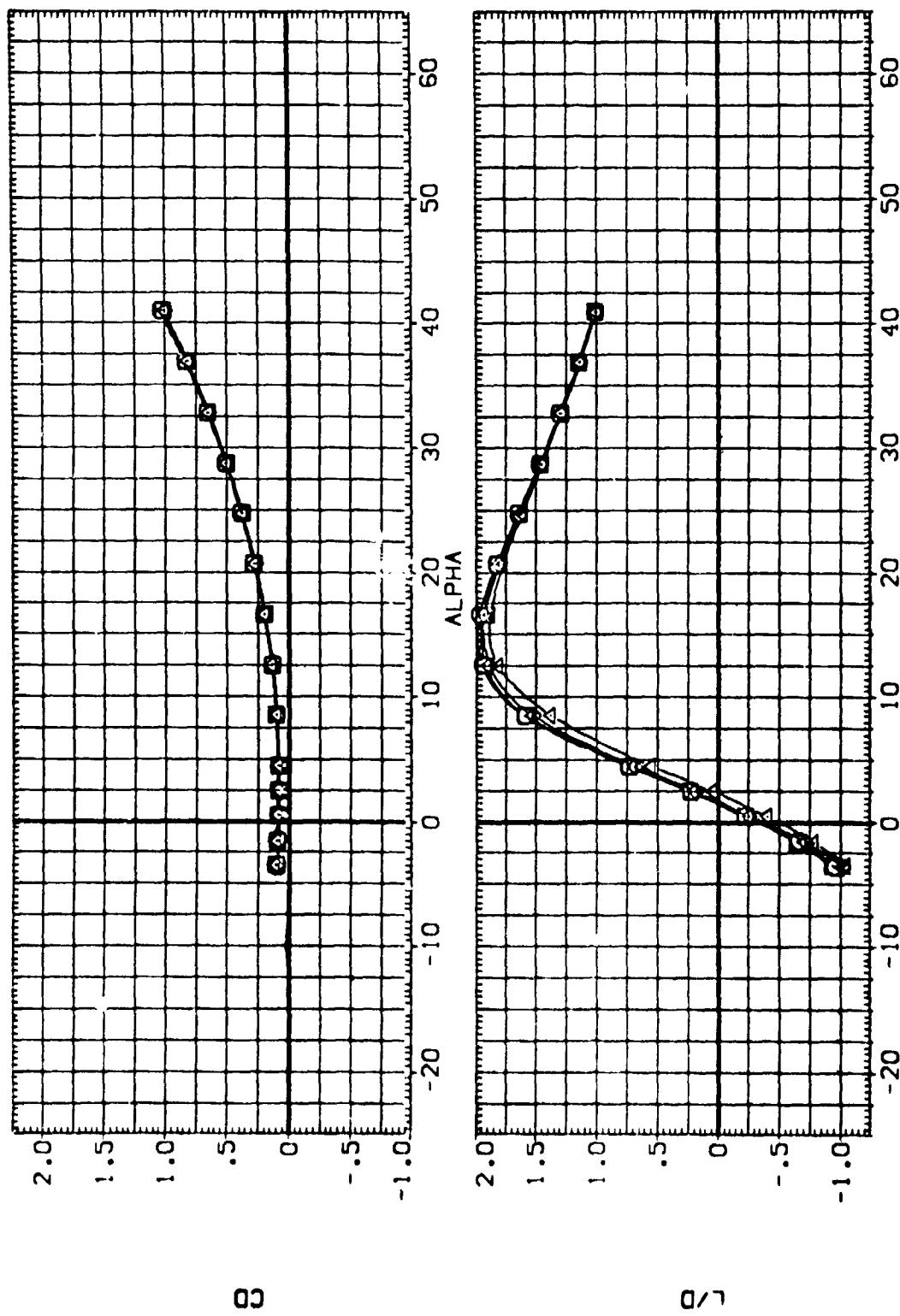
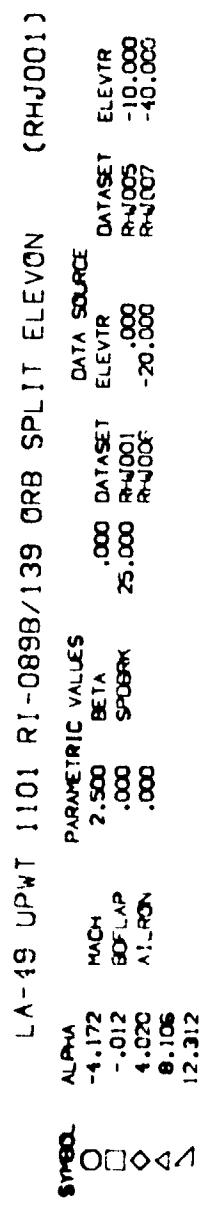


FIGURE 6. OUTBOARD ELEVON PITCH CHARACTERISTICS
($\text{MACH} = 4.60$)



PITCHING MOMENT COEFFICIENT, CLM

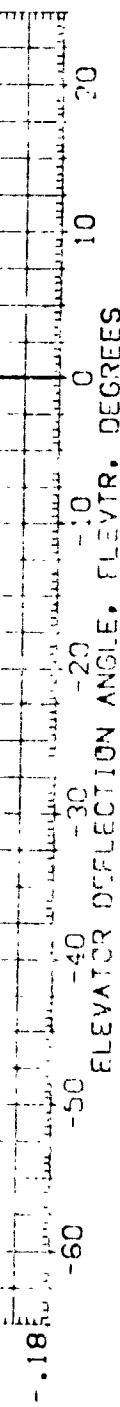


FIGURE 7. FULL SPAN ELEVON PITCH CONTROL EFFECTIVENESS

A-49 PWT 1101 RI-089B/139 ORB SPLIT ELEVON (RHJ001)

Symbol	Alpha	Mach	Parametric Values	Beta	Dataset	Elevtr	Dataset	Elevtr
O	16.292	.000	.000	.000	RHJ001	.000	RHJ005	-10.000
U	20.438	.000	.000	.000	RHJ006	-20.000	RHJ007	-40.000
A	24.504	.000	.000	.000				
R	28.669	.000	.000	.000				
V	32.717	.000	.000	.000				

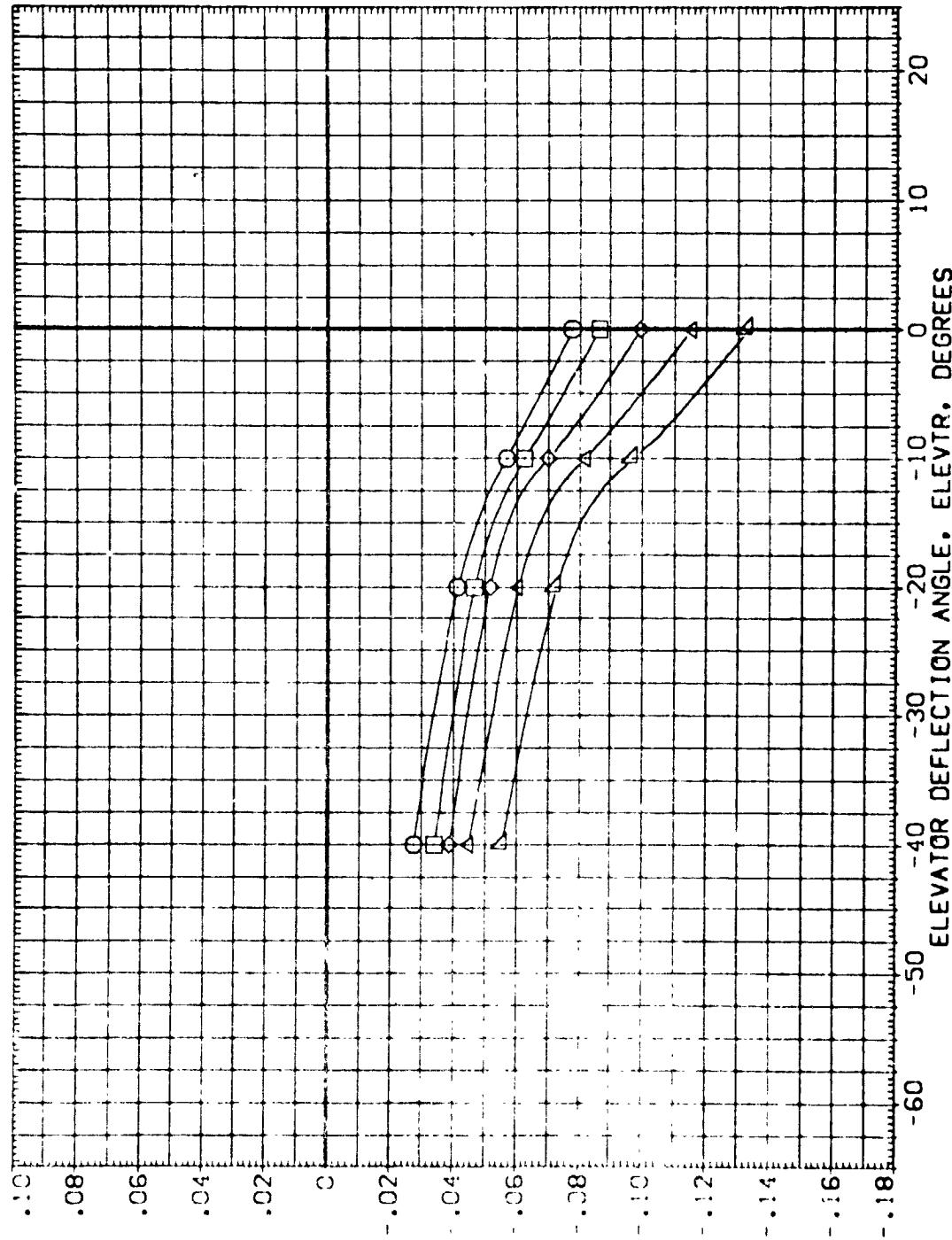
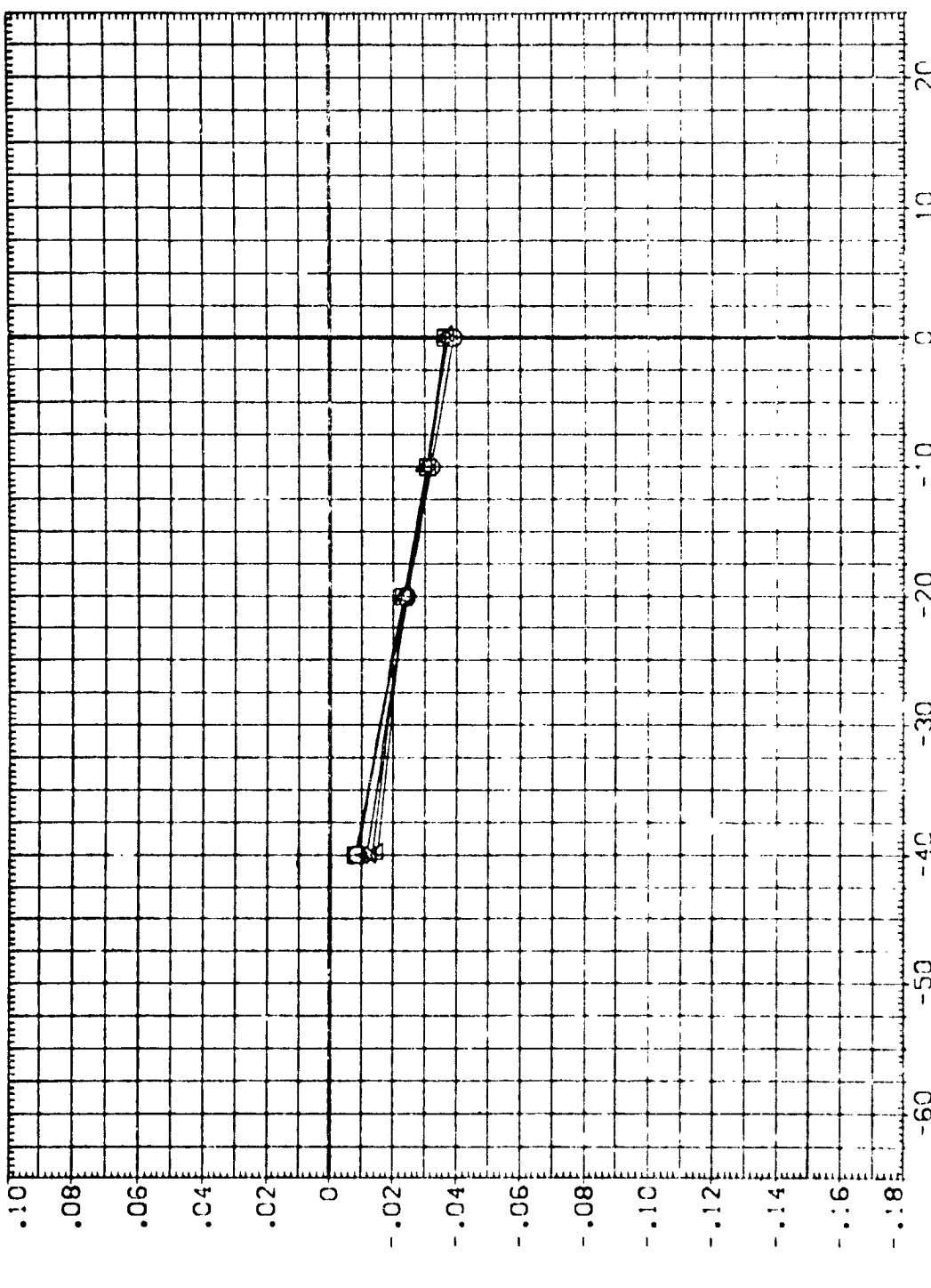


FIGURE 7. FULL SPAN ELEVON PITCH CONTROL EFFECTIVENESS

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L A-49 UPWT 1101 RI-089B/139 ORB SPLIT ELEVON (RHJ001)

SYMBOL	ALPHA	MACH	4.600	BETA	.000	DATASET	ELEVTR	DATASET	ELEVTR
○	-3.548	BOFLAP	.000	SP000K	25.000	RHJ001	RHJ005	-10.000	
□	-1.553	AIRRON	.000			RHJ006	-20.000		-40.000
△	.433					RHJ007			
▽	2.462								
△	4.466								



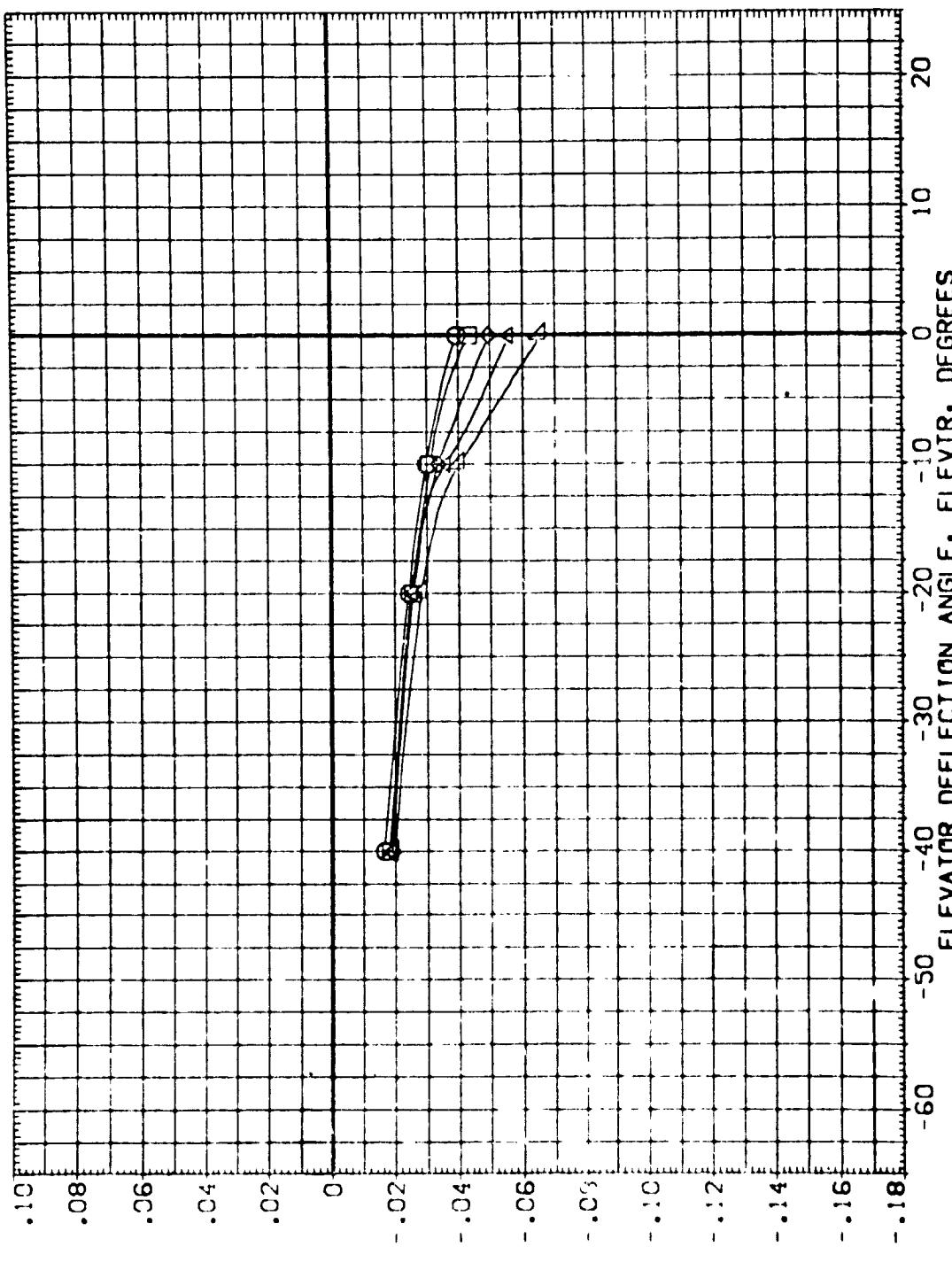
PITCHING MOMENT COEFFICIENT, CLM

FIGURE 7. FULL SPAN ELEVON PITCH CONTROL EFFECTIVENESS

LA-49 UPWT 1101 RI-089B/139 ORB SPLIT ELEVON [RHJ001]

	PARAMETRIC VALUES					
ALPHA	9.540	MACH	4.600	BETA	.000	DATA SET
12.575	BOFLAP	.000	SPOKE	25.000	RHJ001	ELEVTR
16.602	AIRRON	.000		RHJ006	-20.000	DATA SET
20.646				RHJ005	-10.000	ELEVTR
24.715				RHJ007	-40.000	

CLM



PITCHING MOMENT COEFFICIENT, CLM

FIGURE 7. FULL SPAN ELEVON PITCH CONTROL EFFECTIVENESS

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LA-49 UPWT 1101 RI-089B/139 ORB SPLIT ELEVON (RHJ001)
 ALPHAS MACH 4.600 BETA .000 DATASET ELEVTR
 28.780 0.000 SPAN 75.000 RHJ001 .000
 32.804 BOFLAP .000 SPANR 75.000 RHJ005 -20.000
 36.891 AIRRON .000 SPANL 75.000 RHJ007 -40.000
 40.981

SOURCE: O C D □ ◇ ▲

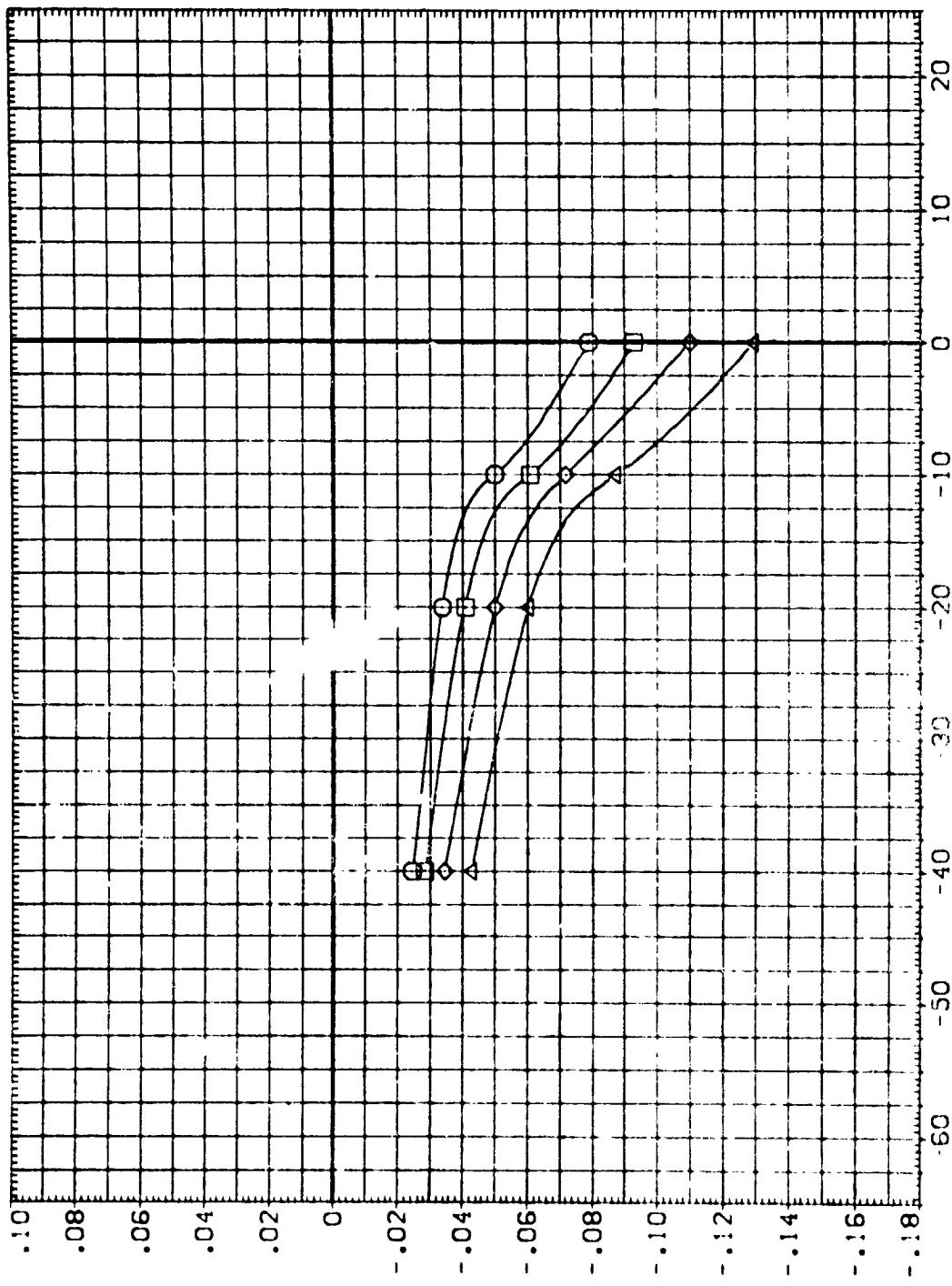


FIGURE 7. FULL SPAN ELEVON PITCH CONTROL EFFECTIVENESS

A-49 UPWT 1101 RI-089B/139 ORB SPLIT ELEVON (BHJF01)

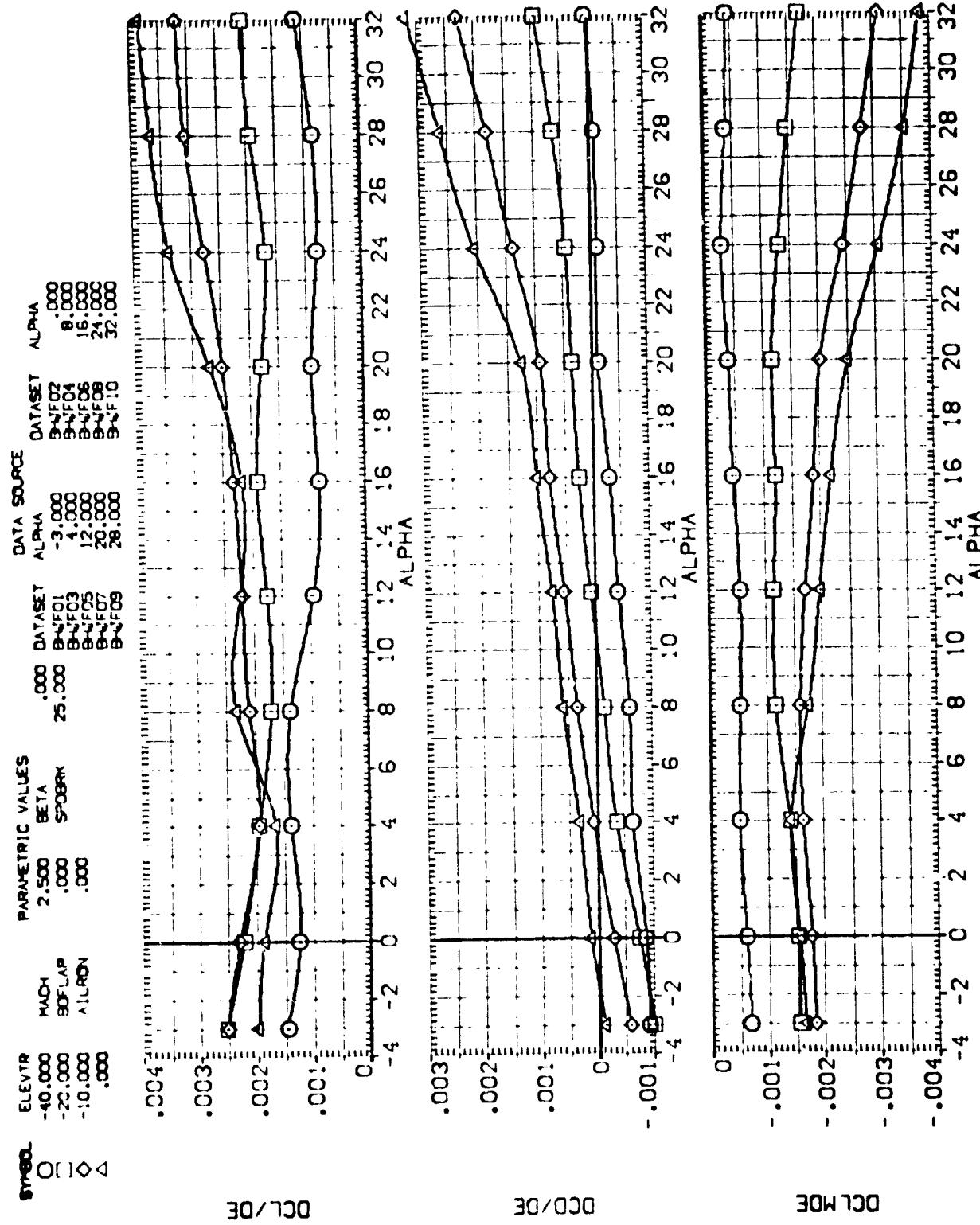


FIGURE 7. FULL SPAN ELEVON PITCH CONTROL EFFECTIVENESS

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LA-49 UPWT 1101 RI-089B/139 ORB SPLIT ELEVON (BHJF01)
 ELEVATOR
 MACH 4.630 BETA .000 DATASET ALPHA .000
 BDFAP .000 SPBRK 25.000 BDF01 -3.000
 AIRDN .000 BDF02 4.000 BDF04 8.000
 .000 BDF03 12.000 BDF06 16.000
 BDF05 20.000 BDF08 24.000
 BDF07 28.000 BDF10 32.000
 BDF09

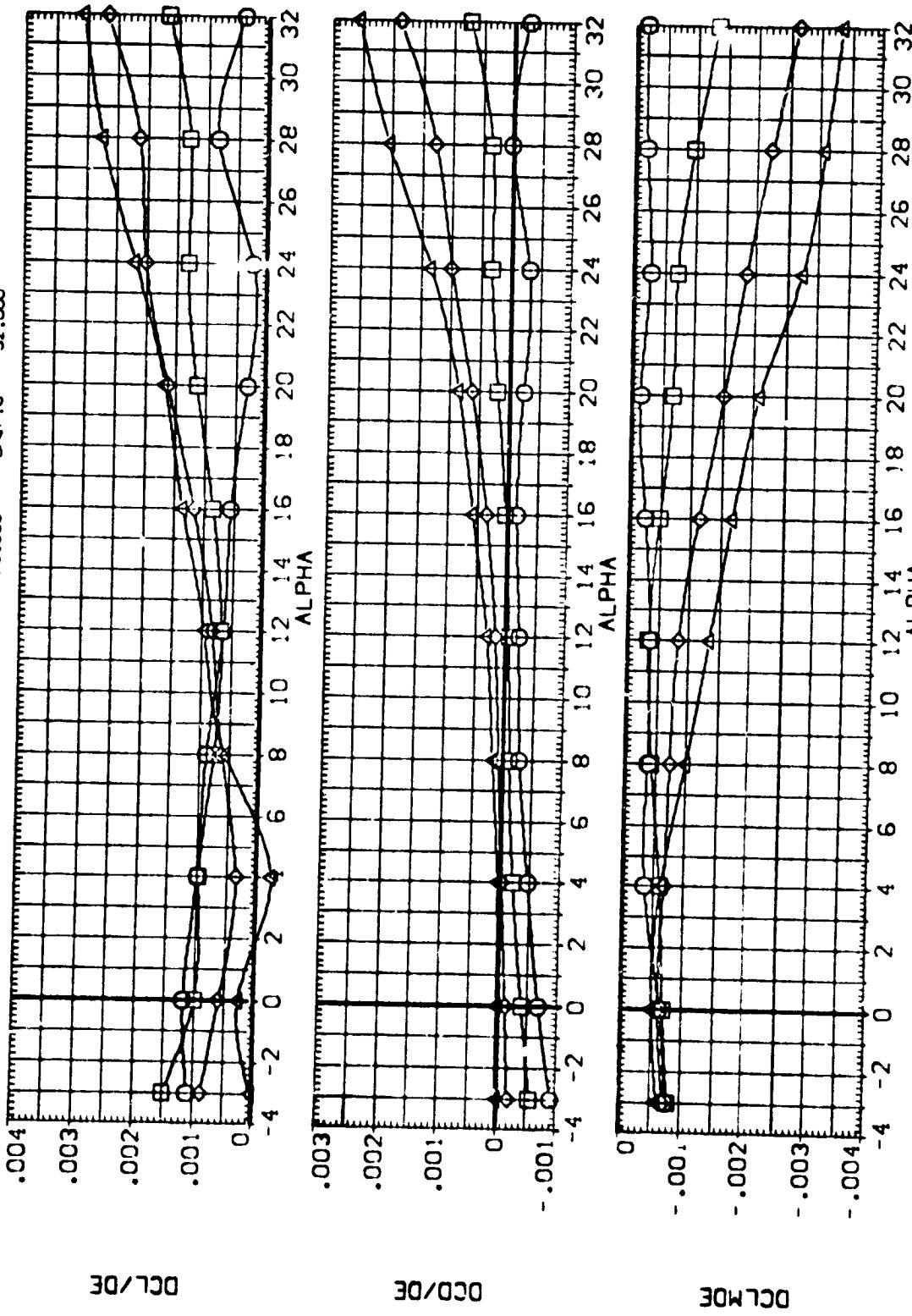


FIGURE 7. FULL SPAN ELEVON PITCH CONTROL EFFECTIVENESS

LA-49 UPWT 1101 RI-089B/139 ORB SPLIT ELEVON (RHJ004)
 ALPH_W MACH_W BETA_W DATASET_W ELV-L1
 -.072 .000 .000 RI-J004 -40.000 DATASET_L
 -.004 ELV-L0 .000 RI-J004 -40.000 RI-J003 -20.000
 .002 BOFLAP .000 RI-J002 -10.000 RI-J001 .000
 .130 AIRDN .000
 12.224

PARAMETRIC VALUES
 ALPH_W MACH_W BETA_W DATASET_W ELV-L1
 -.072 .000 .000 RI-J004 -40.000 DATASET_L
 -.004 ELV-L0 .000 RI-J004 -40.000 RI-J003 -20.000
 .002 BOFLAP .000 RI-J002 -10.000 RI-J001 .000
 .130 AIRDN .000
 12.224

PITCHING MOMENT COEFFICIENT. CLM

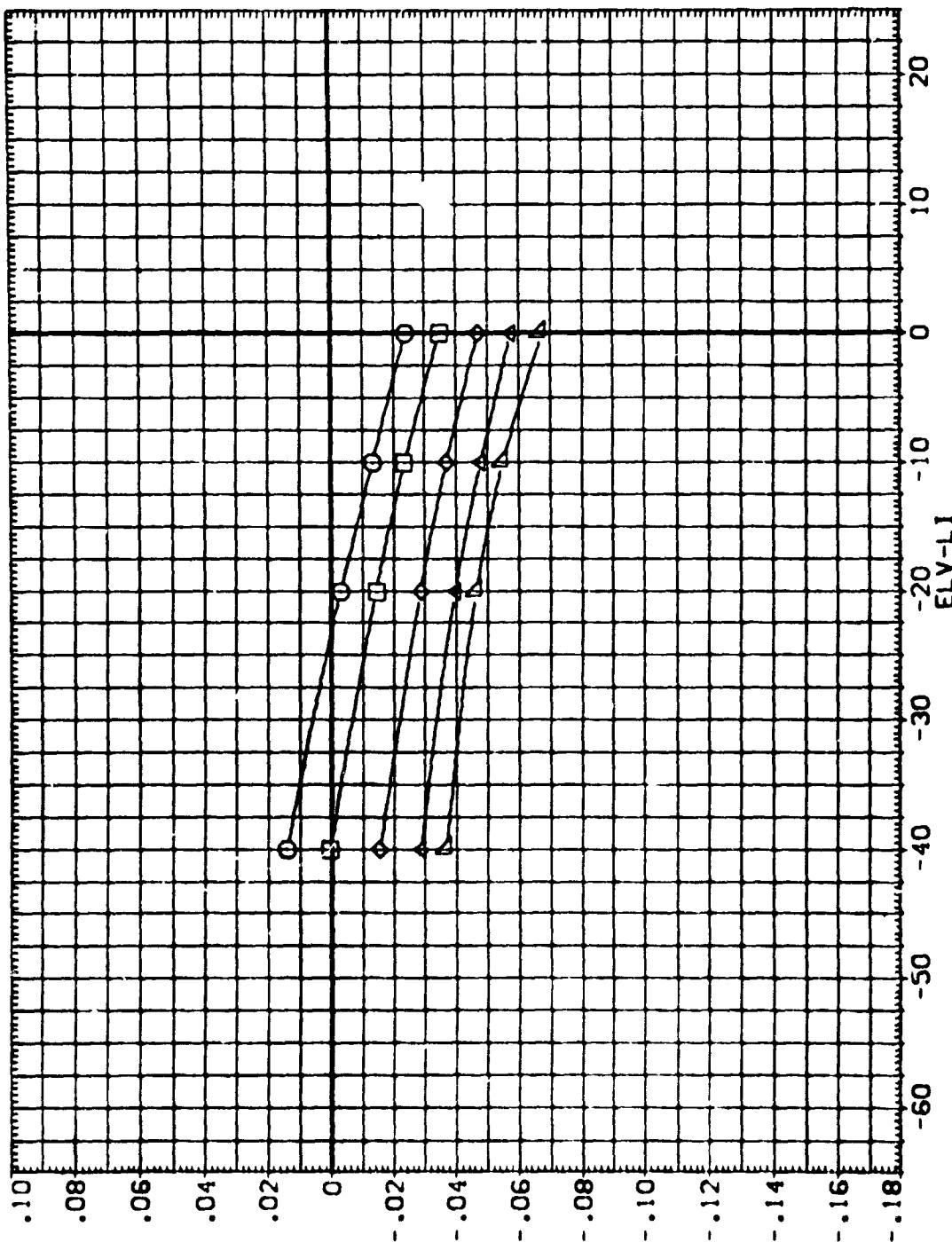


FIGURE 8. INBOARD ELEVON PITCH CONTROL EFFECTIVENESS

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-A-49 UP#T 1101 RI-089B/139 ORB SPLIT ELEVON (RHJ004)

SIMUL	ALPHA	MACH	2.500	BETA	.000	DATA SET	ELV-LI	DATA SOURCE
O	16.315	E., .0	.000	ELV-R0	.000	RJ003	-40.000	ELV-LI
O	20.436	E., .0	.000	ELV-R0	.000	RJ003	-10.000	RJ001
O	24.518	SOFLAP	.000	SPOISK	25.000	RJ002	-20.000	RJ001
O	28.652	AIRDN	.000				.000	
	32.770							

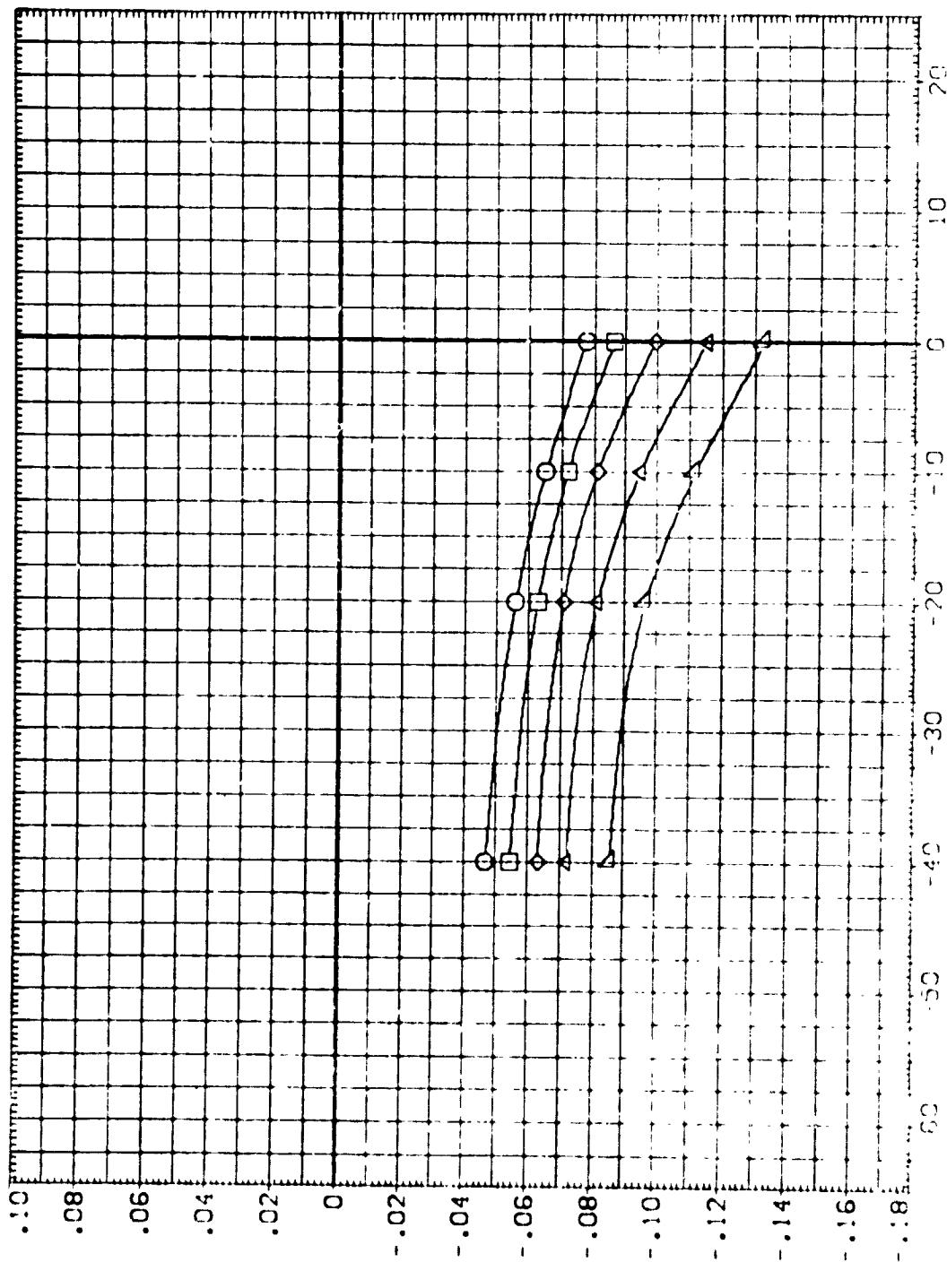


FIGURE 8. EFFECTIVE LIFTED AREA PREDICTION FOR A-49

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LA-49 UPWT 1101 RI-089B/139 ORB SPLIT ELEVON (RHJ004)

PARAMETRIC VALUES		DATA SOURCE	
		ELV-LI	ELV-LI
ALPHA	-3.524	MACH	.600
SYMBOL	○ □ ◇ ▲ △	ELV-L0	.000
	-1.576	ELV-R0	.000
	.453	SP08R	25.000
	2.165	BOFLAP	.000
	1.194	AILRON	.000

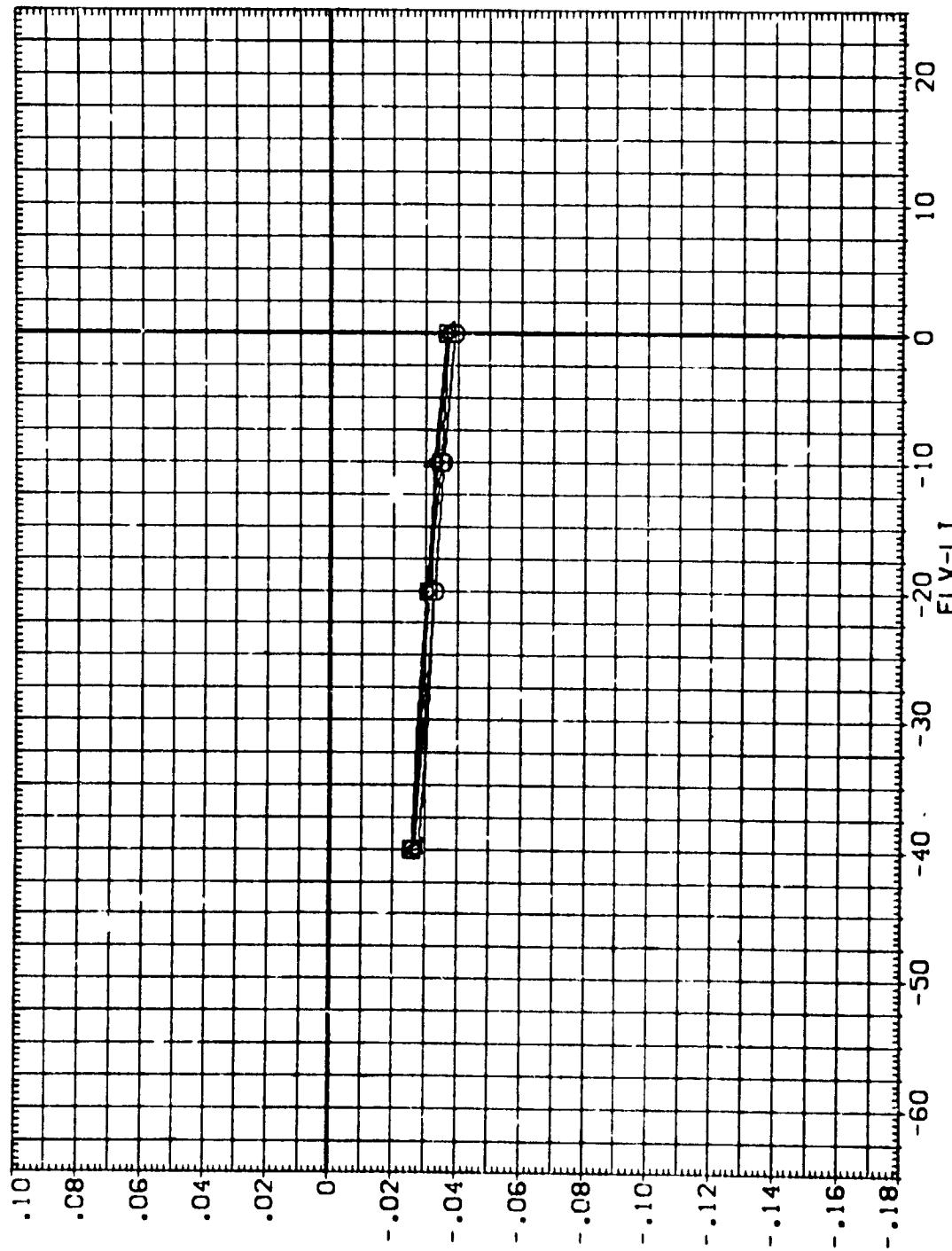


FIGURE 8. INBOARD ELEVON PITCH CONTROL EFFECTIVENESS

LA-49 UPWT 1101 RI-089B/139 ORB SPLIT ELEVON (RHJ004)
 DATA SOURCE
 ALPHA MACH .4.600 BETA .000 DATASET ELV-L1
 12.570 ELV-L0 .000 ELV-R0 .000 RHJ004 -10.000 RHJ003 -20.000
 16.591 BDFLAP .000 SPDBRK 25.000 RHJ002 -10.000 RHJ001 .000
 20.636 ALTRON .000
 24.738

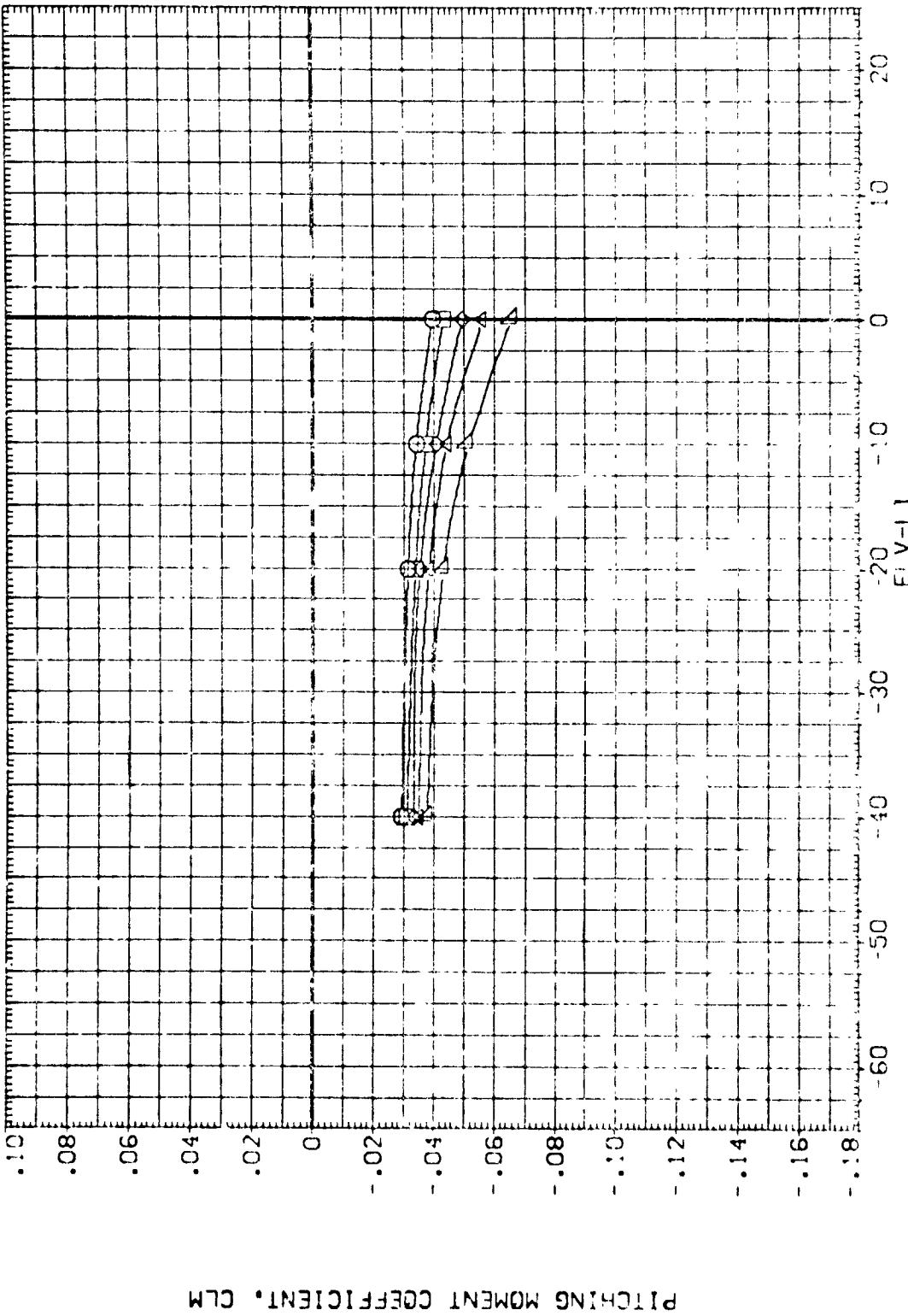
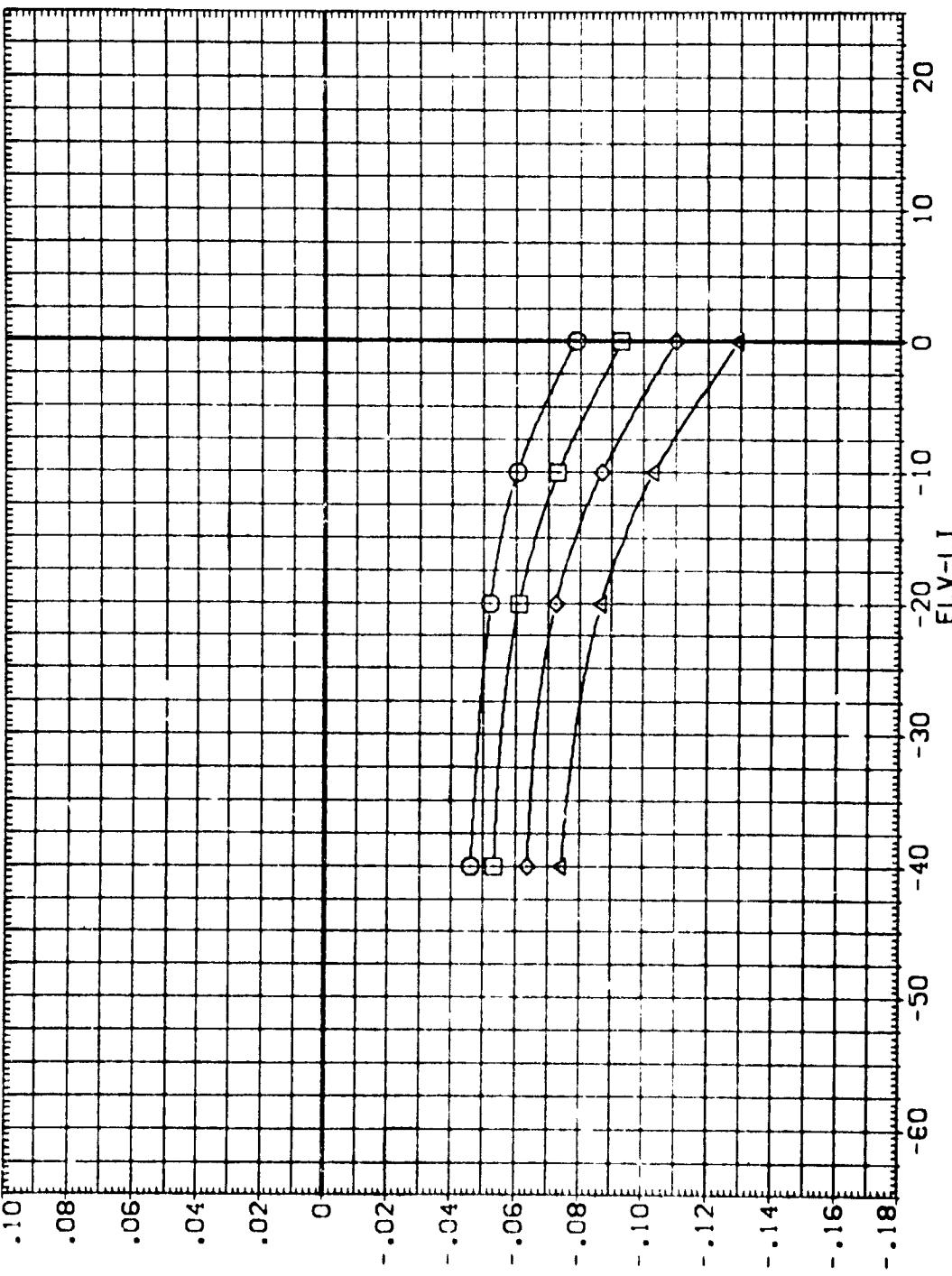


FIGURE 8, INBOARD FLOWN PITCH CONTROL EFFECTIVENESS

LA-49 UPWT 1101 RI-089B/139 ORB SPLIT ELEVON (RHJ004)
 ALPHA MACH .600 PARAMETRIC VALUES
 28.760 ELV-LI .000 DATA SOURCE
 32.826 ELV-RD .000 DATASET
 36.901 BOFLAP 25.000 ELV-LI
 40.956 AILRON .000 RI-004
 RI-002 RI-003 RI-001 RI-000
 RI-0001 RI-0002 RI-0003 RI-0001
 RI-0000 RI-0001 RI-0002 RI-0003
 RI-0000

SYMBOLS



PITCHING MOMENT COEFFICIENT. CLM

FIGURE 8. INBOARD ELEVON PITCH CONTROL EFFECTIVENESS

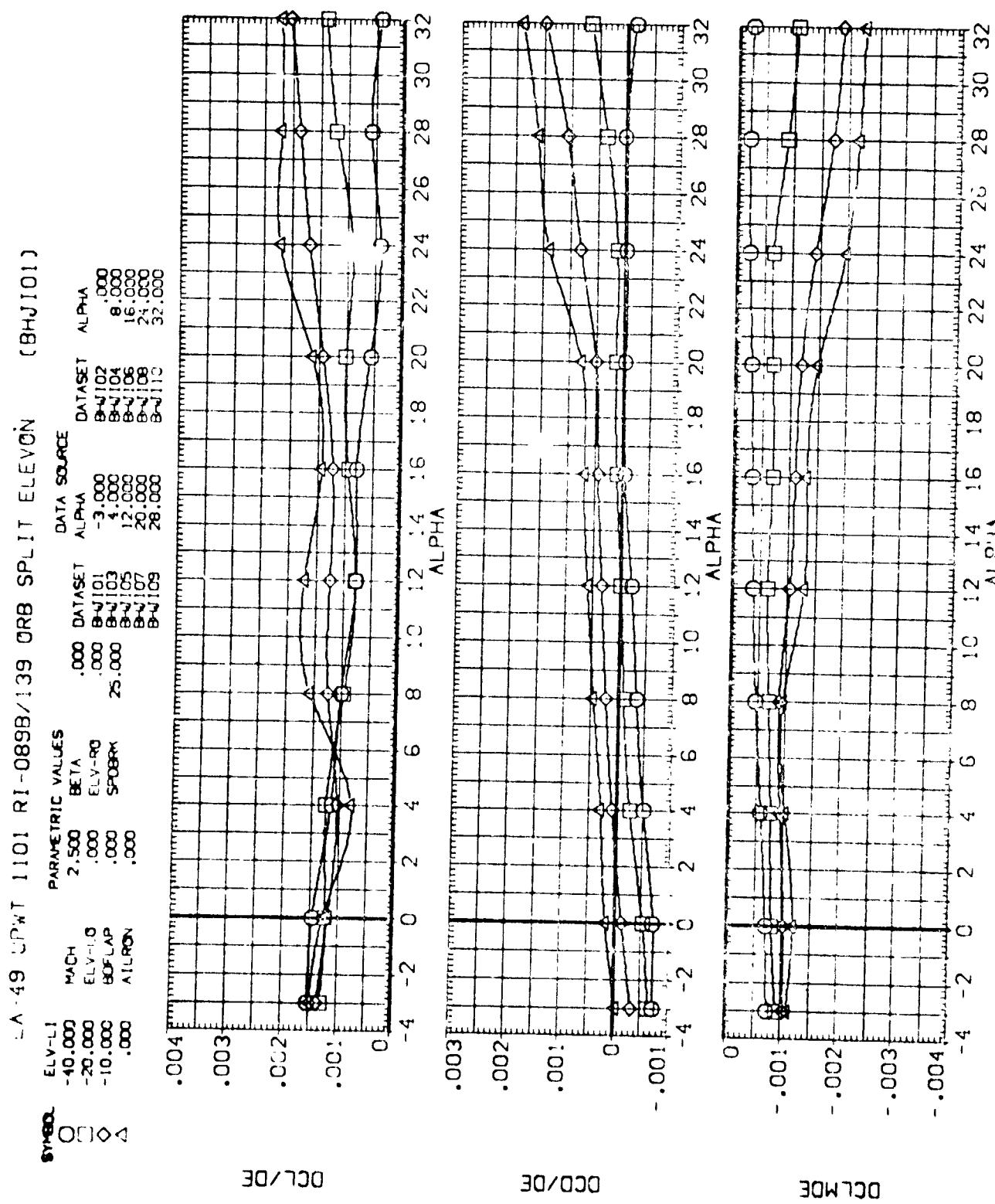


FIGURE 8. INBOARD ELEVON PITCH CONTROL EFFECTIVENESS

LA-49 UPWT 1101 RI-089B/139 ORB SPLIT ELEVON (BHJ101)

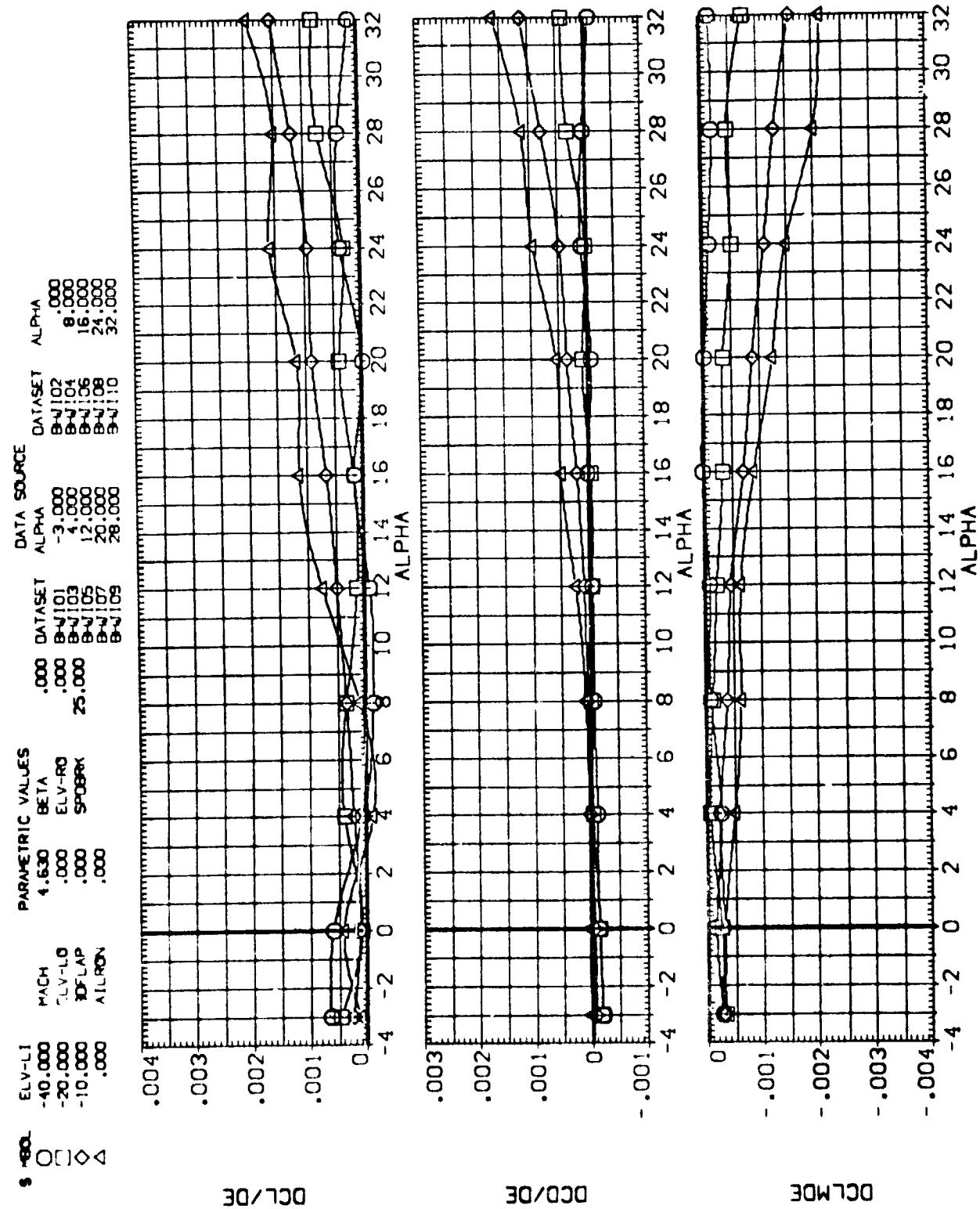
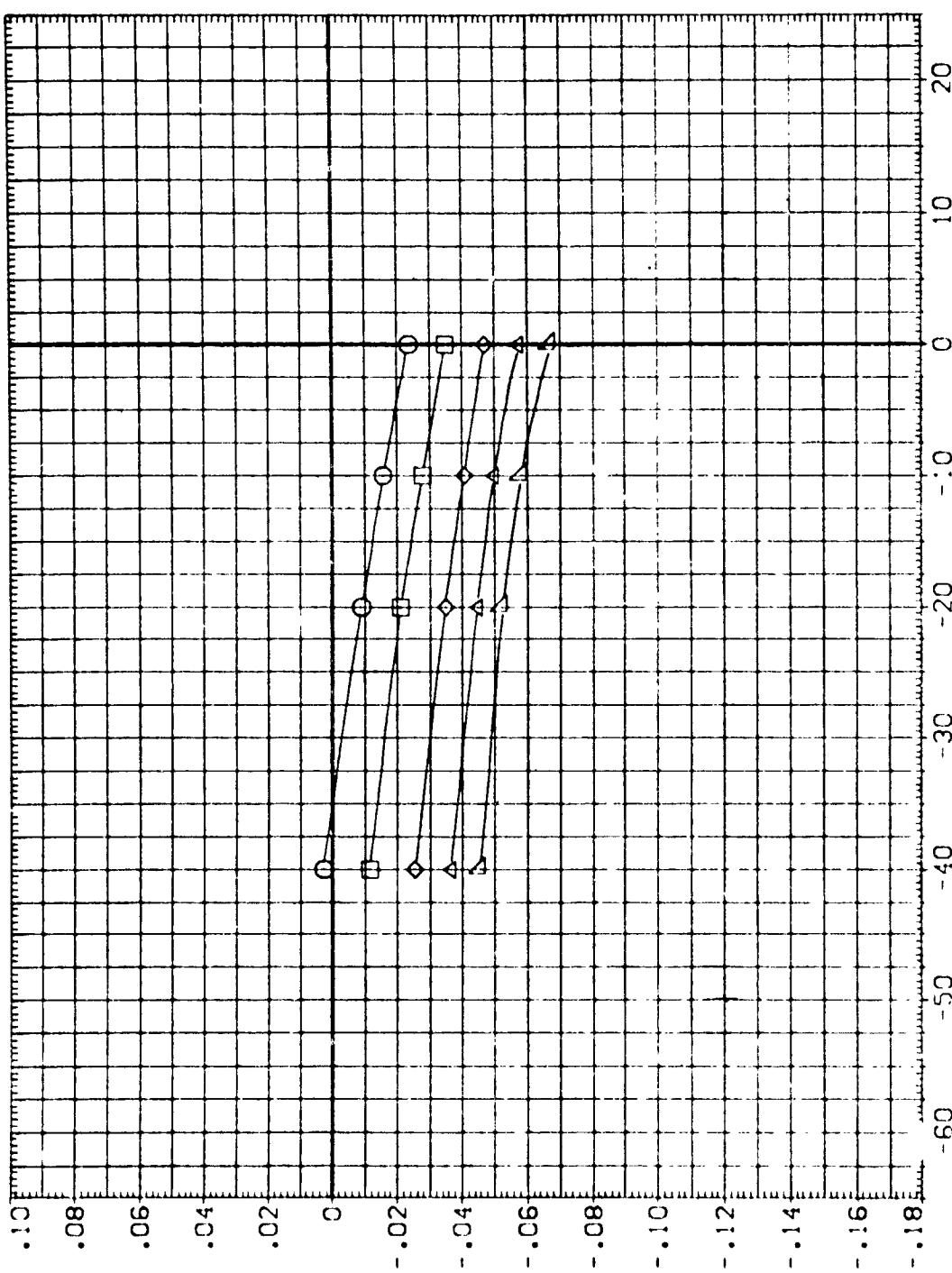


FIGURE 8. INBOARD ELEVON PITCH CONTROL EFFECTIVENESS

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LA-49 UP, r 1101 RI-089B/139 G/R SPLIT E-EVN
 (RHJ001)

	ALPHA	MACH	BETA	.000	DATASET	ELV-L0	DATA SOURCE
SYMBOL	-4.172	.000	ELV-R1	.000	RJ001	0.000	RJ008
O	-0.012	.000	SP00X	.000	RJ009	-20.000	RJ010
+	4.020	.000				-40.000	
X	8.106	.000					
V	12.312	.000					



PITCHING MOMENT COEFFICIENT, CLM

FIGURE 9. AUTOMATED EVEN PITCH CONTROL EFFECTIVENESS

LA-49 UPWT 1101 RI-089B/139 ORB SPLIT ELEVON (RHJ001)
 PARAMETRIC VALUES
 ALPHA 16.292 MACH 2.500 BETA .000 DATASET RI-001 ELV-L0
 20.438 E-V-L1 .000 ELV-RI .000 DATASET RI-008 E-V-L0
 24.504 BDF-LAP .000 SPD-BRK 25.000 DATASET RI-009 RI-010 -10.000
 28.669 AIL-RDN .000 -20.000 -40.000
 32.717

SYMBOLS

O \square \diamond \triangle
 0 0 0 0

PITCHING MOMENT COEFFICIENT, CLM

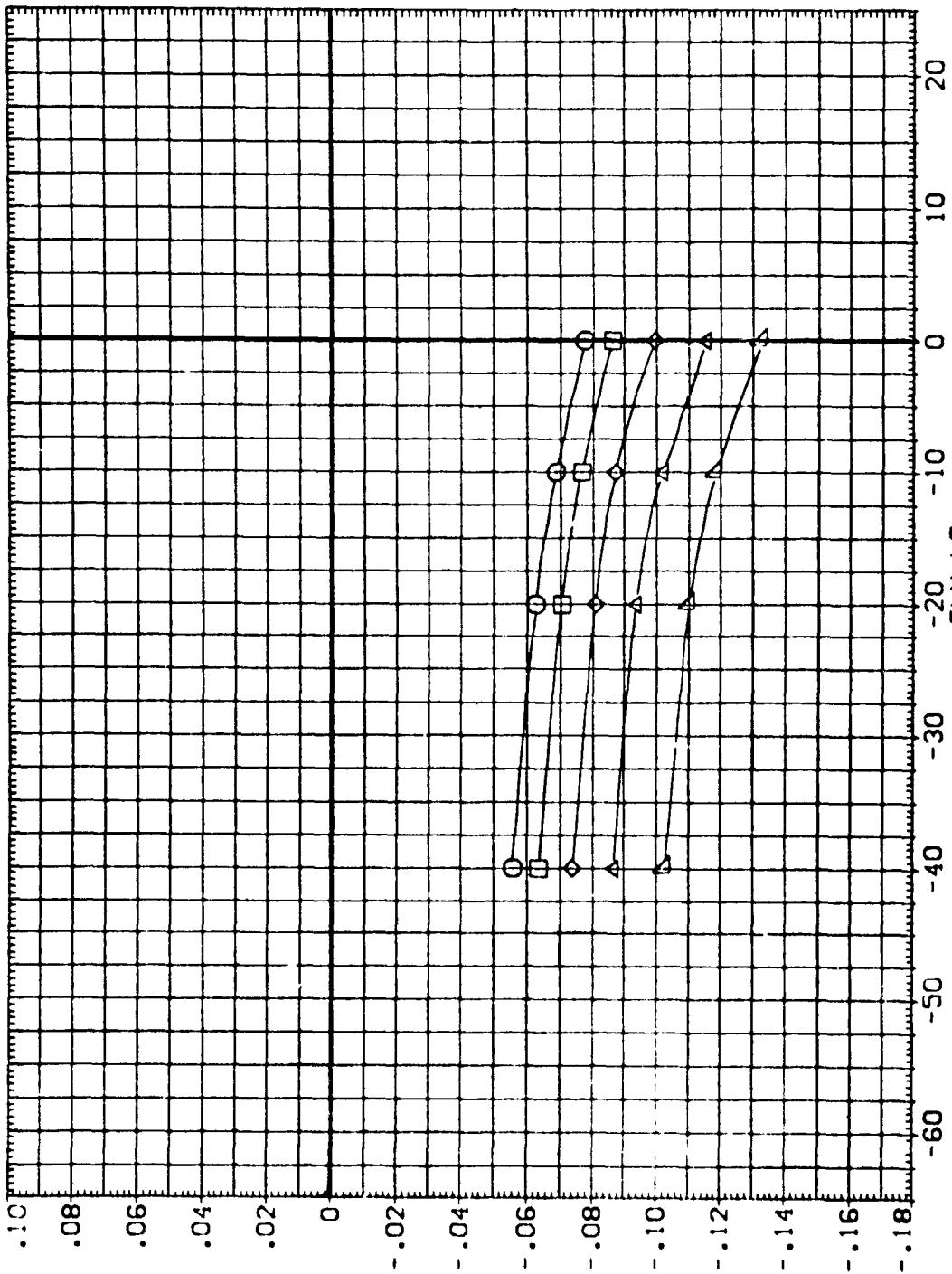
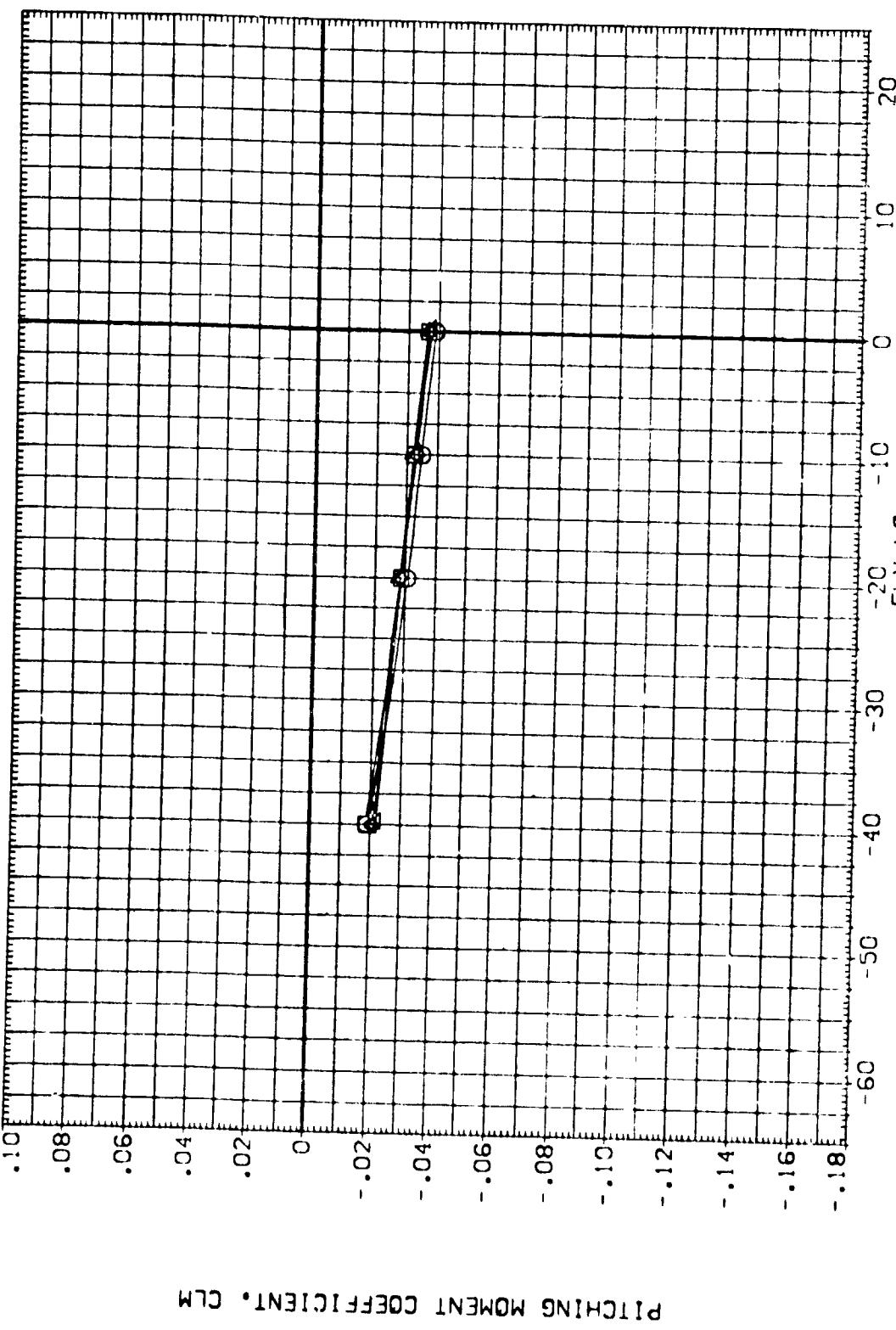


FIGURE 9. OUTBOARD ELEVON PITCH CONTROL EFFECTIVENESS

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LA-49 L₁, T 1101 RI-089B/139 ORB SPLIT ELEVON (RHJ001)

	PARAMETRIC VALUES		DATA SOURCE
α_{PMA}	MACH 4.600	$\beta_{\text{T}} .000$	DATASET ELV-L0
-3.58	E_V-L1 .000	ELV-R1 .000	ELV-L0
-1.553	BDFLAP .000	RHJ001 RHJ009	RHJ009
.433	SPDBRK .000	RHJ009 -20.000	RHJ010
2.462	ALRDN .000		-40.000
4.466			



PITCHING MOMENT COEFFICIENT, CLM

FIGURE 9. OUTBOARD ELEVON PITCH CONTROL EFFECTIVENESS

LA-49 UPWT 1101 RI-089B/139 ORB SPLIT ELEVON (RHJ001)

PARAMETRIC VALUES	DATA SOURCE
ALPHA .540	MACH .500
ELV-LI 12.575	BETA .000
BOFLAP 16.602	ELV-RI .000
AIRRON 20.646	SPBPK 25.000
24.715	RHJ001 RHJ003 RHJ010

SYMBOLS

- O MACH
- ELV-LI
- △ BOFLAP
- ▽ AIRRON

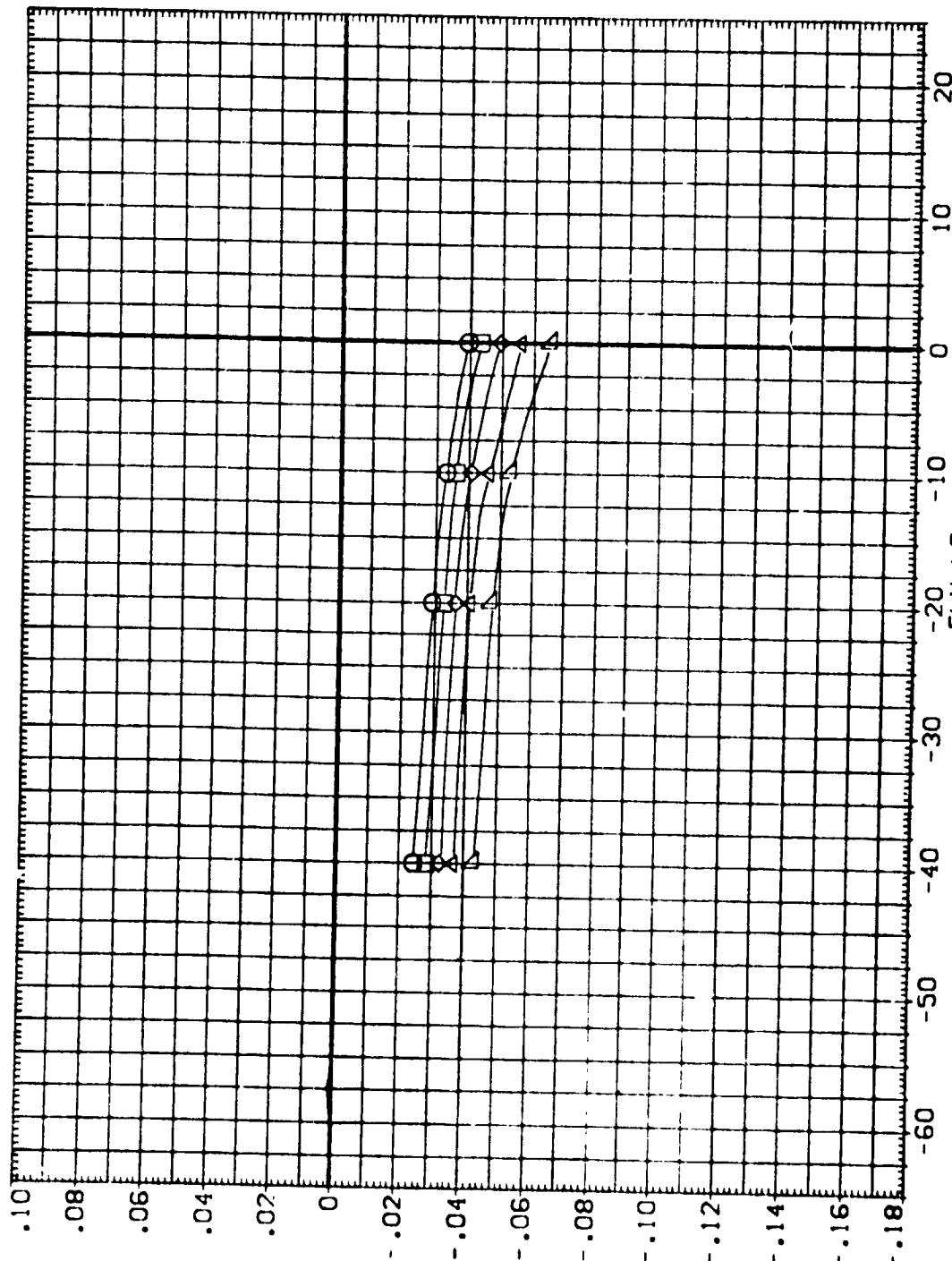


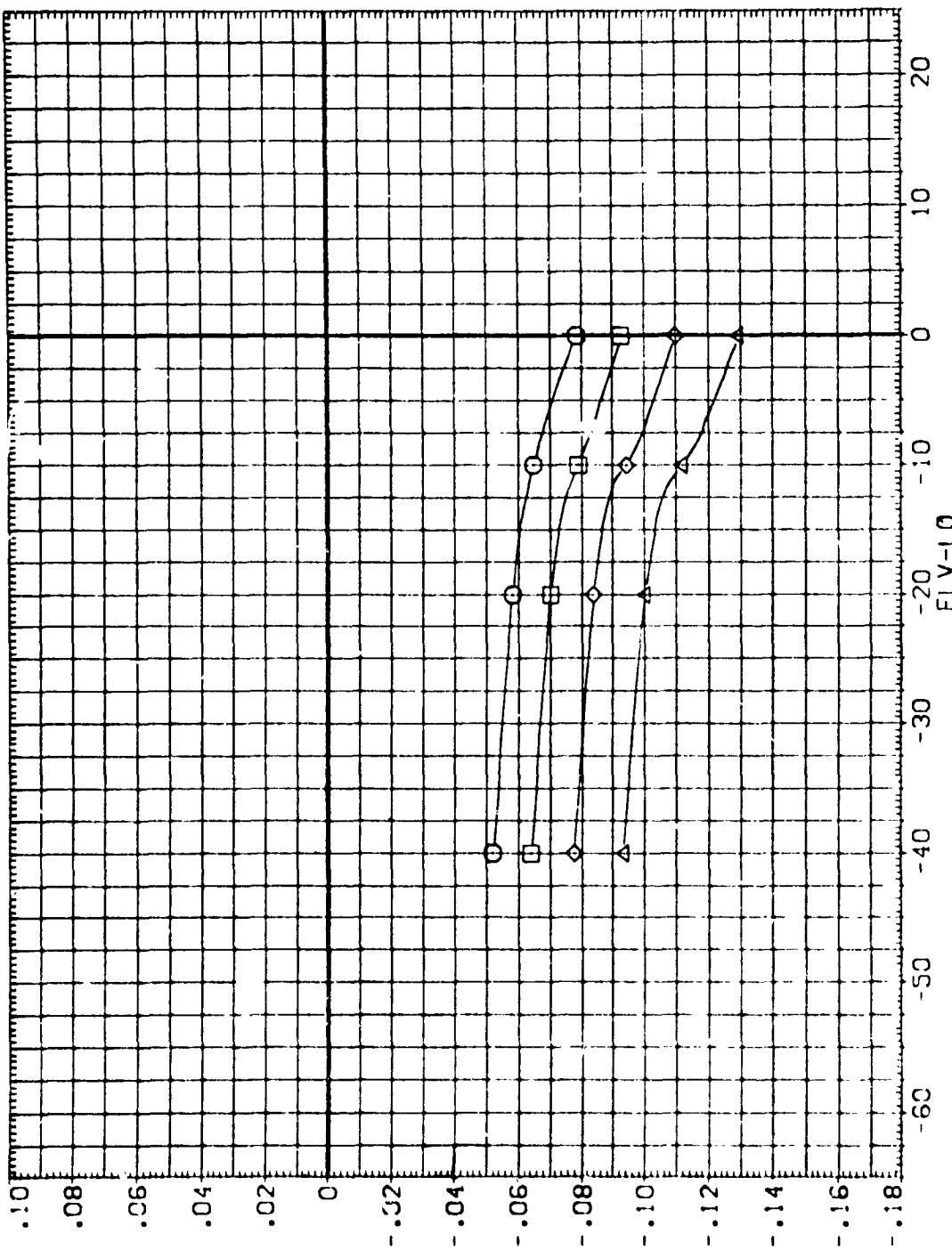
FIGURE 9. OUTBOARD ELEVON PITCH CONTROL EFFECTIVENESS

LA-49 UPWT 1101 RI-089B/139 ORB SPLIT ELEVON (RHJ001)

	PARAMETRIC VALUES			DATA SOURCE	ELV-L0	ELV-R0	DATA SET	ELV-L0	ELV-R0
ALPHA	4.500	BETA	.000	DATA SET	.000	RJ001	RJ002	-10.000	-40.000
MACH	.000	ELV-L1	.000	DATA SET	.000	RJ001	RJ002	-20.000	RJ010
FLV-L1	.000	ELV-R1	.000	DATA SET	.000	RJ001	RJ002	-20.000	RJ010
SP09X	.000	SP09Y	25.000	DATA SET	RJ009	RJ009	RJ009	RJ009	RJ009
BDFLAP	.000	AIRDN	.000						

SYMOL

O □ △ ▲



PITCHING MOMENT COEFFICIENT, CLM

FIGURE 9. FORWARD ELEVON PITCH CONTROL EFFECTIVENESS

LA-49 UPWT 1101 R!-089B/139 ORB SPLIT ELEVON (BHJ001)

	PARAMETRIC VALUES							
ELV-LG	MACH	2.500	BETA	.000	DATA SET	ALPHA	DATA SOURCE	ALPHA
O	ELV-LI	.000	ELV-RI	.000	B-V001	-3.000	B-V002	.000
□	BDF LAP	.300	SPLIT	.25000	B-V003	4.000	B-V004	8.000
△	AILRON	.000			B-V005	12.000	B-V006	16.000
▽					B-V007	20.000	B-V008	24.000
					B-V009	28.000	B-V010	32.000

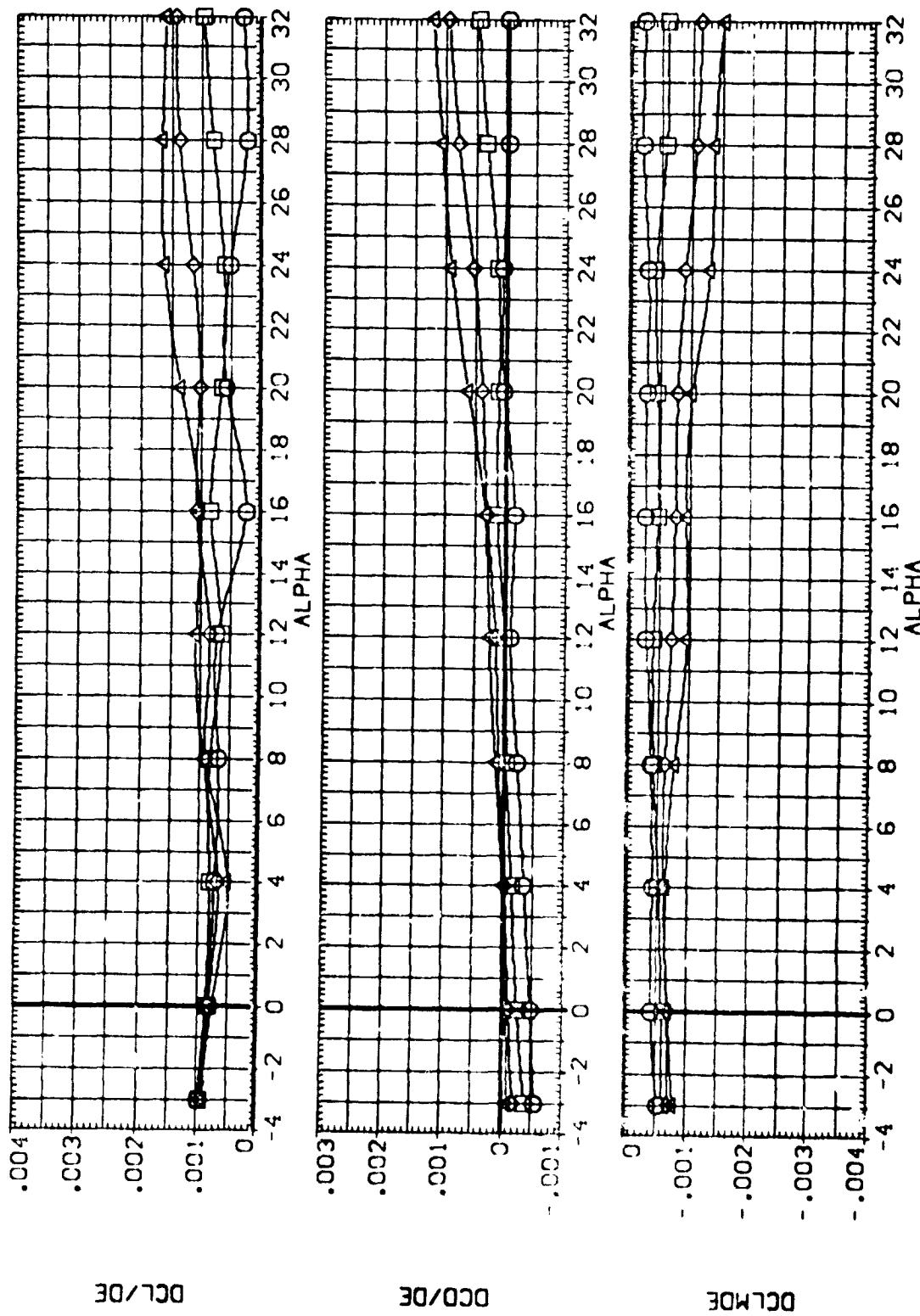


FIGURE 9. OUTBOARD ELEVON PITCH CONTROL EFFECTIVENESS

A-49 UPWT 1101 PI-0898/139 ORB SPLIT ELEVON (BHJ001)

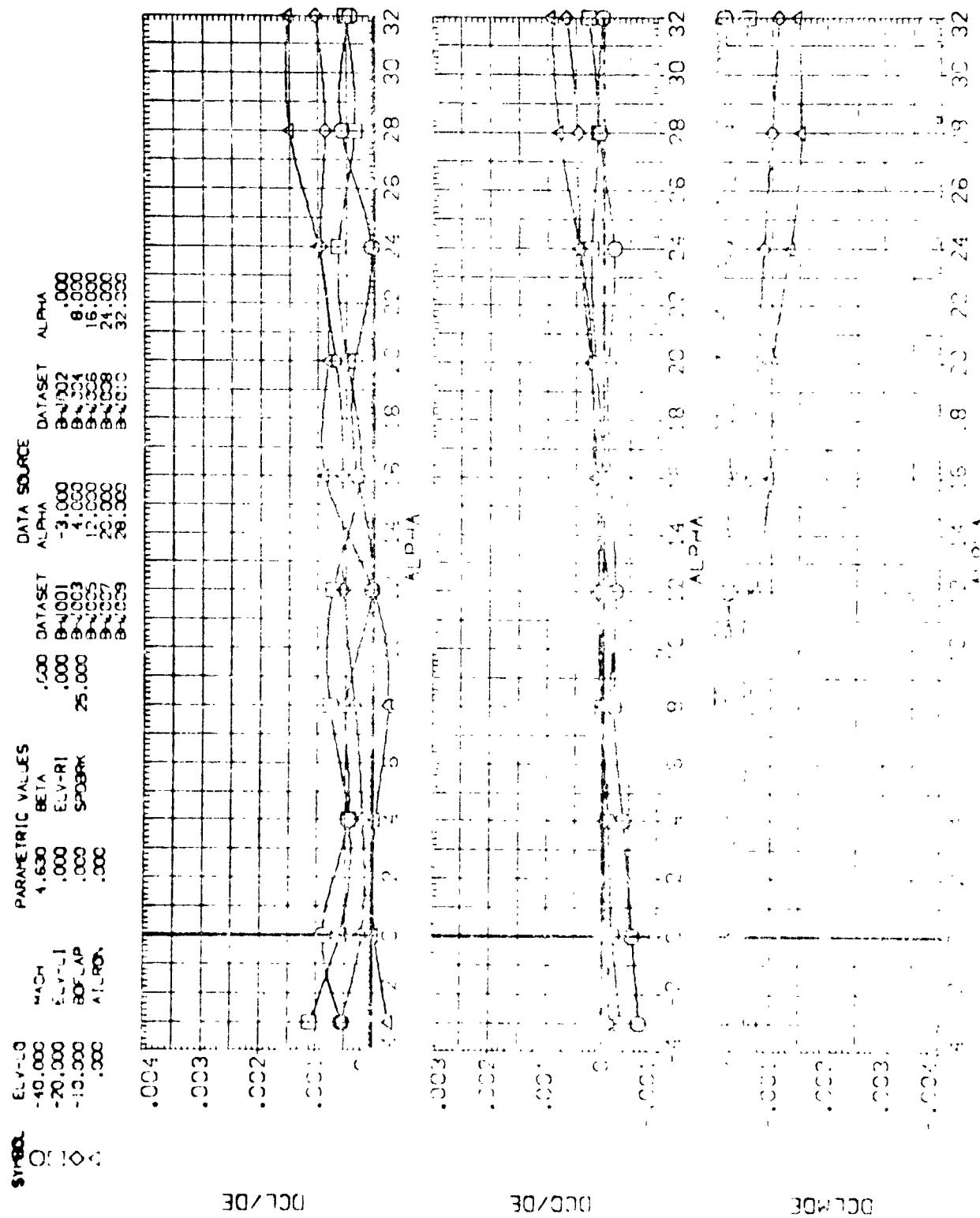
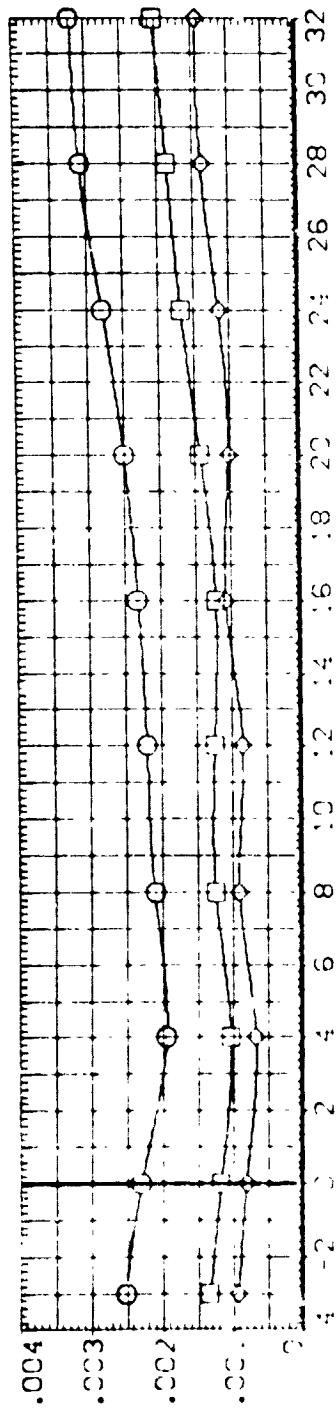


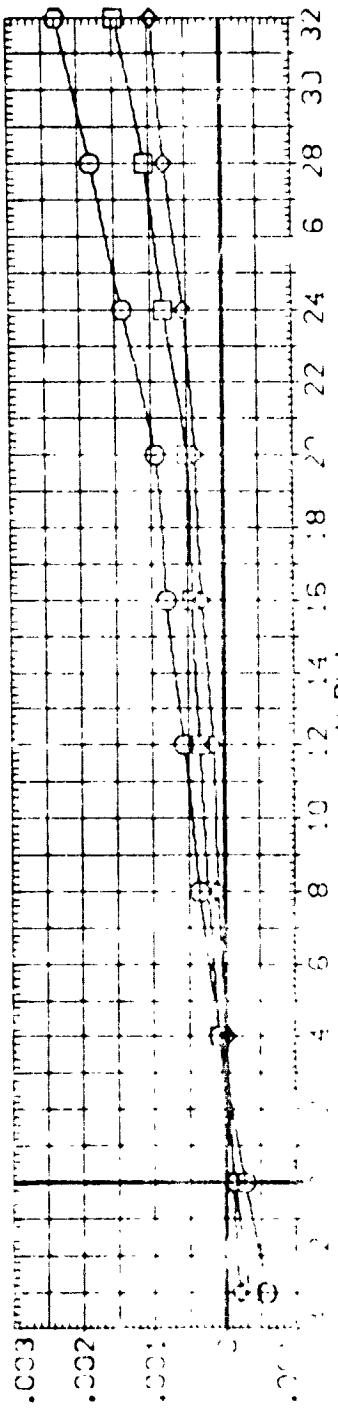
FIGURE A-49 UPWT 1101 PI-0898/139 ORB SPLIT ELEVON (BHJ001)

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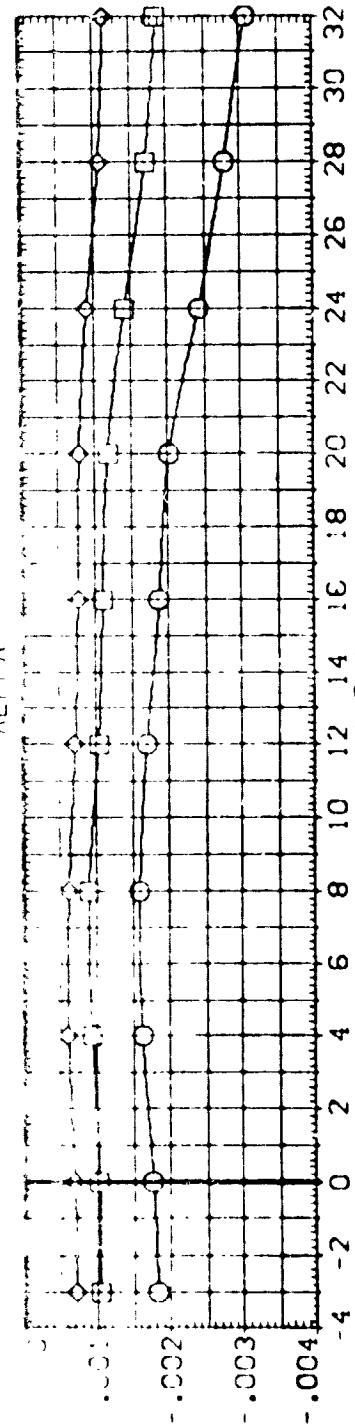
DATA SET	SYM	CONFIGURATION	DESCRIPTION
(EN005)	C	-A-19 SPV	R1-0889A; 39 088 SP-L1 ELEVON
(EN005)	O	-A-19 SPV	R1-0889A; 39 088 SP-R1 ELEVON
(EN005)	S	-A-19 SPV	R1-0889A; 39 088 SP-L2 ELEVON
(EN005)	D	-A-19 SPV	R1-0889A; 39 088 SP-R2 ELEVON



001/000



000/000



000/000

**FIGURE 10. COMPARISON OF FULL SPAN, INBOARD AND OUTBOARD PITCH CONTROL ($\alpha_E = -10^\circ$)
 $(\text{MACH} = 2.50)$**



DATA SET	STRUCTURE	DESCRIPTION	ELV-L0	ELV-R1	ELV-R0
ICN 005	DA-49 SPAN 1	1101 R-0888/ 39 098 SPAN ELEV 6	-10.000	-10.000	-10.000
ICN 006	DA-49 SPAN 1	1101 R-0888/ 39 098 SPAN ELEV 6	-10.000	-10.000	-10.000
ICN 007	DA-49 SPAN 1	1101 R-0888/ 39 098 SPAN ELEV 6	-10.000	-10.000	-10.000
ICN 008	DA-49 SPAN 1	1101 R-0888/ 39 098 SPAN ELEV 6	-10.000	-10.000	-10.000

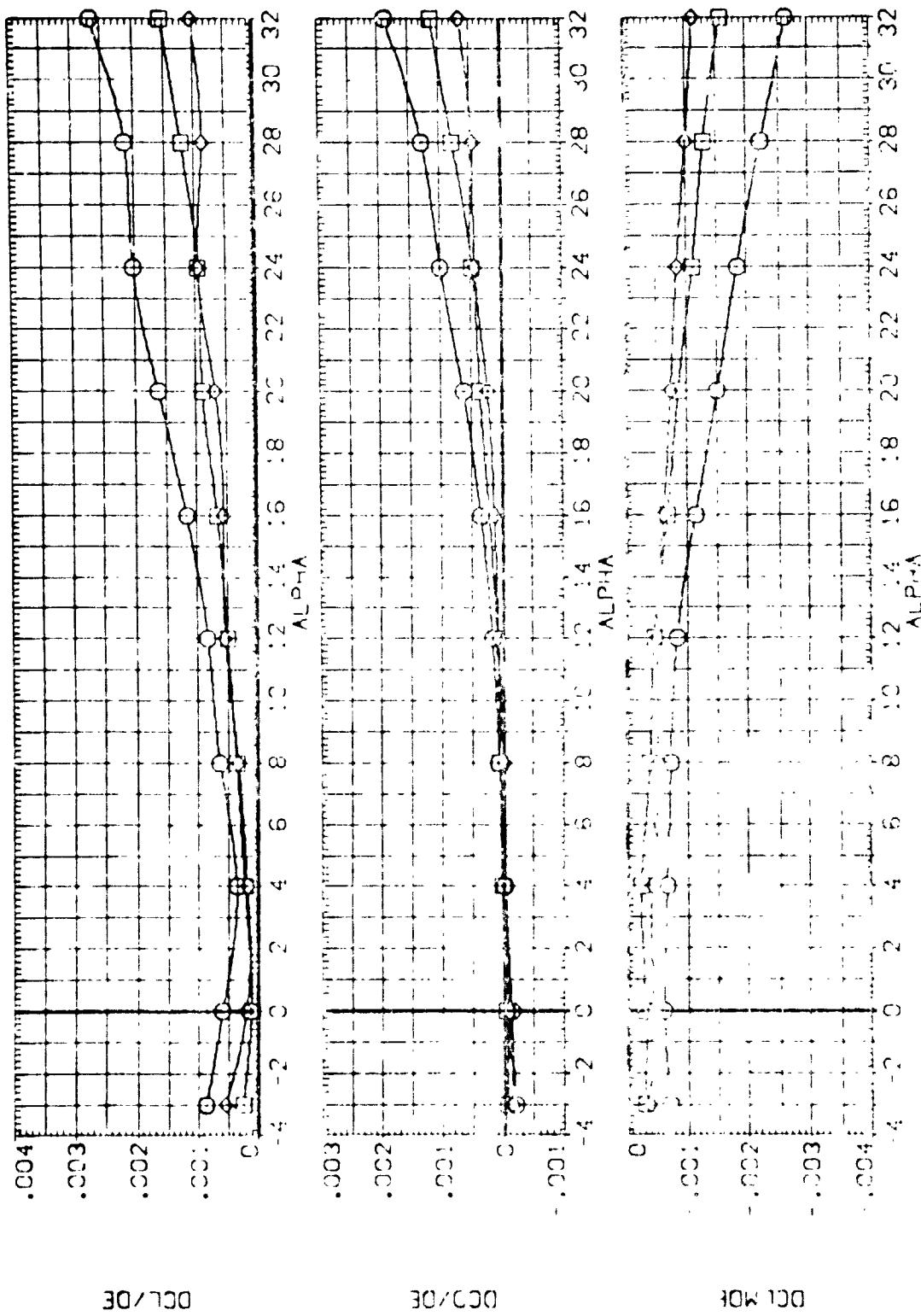


FIGURE 10. COMPARISON OF FLAT SPAN, INCARC AND GUTBOARD PITCH CON RATE (DE = -10)
 DCL = 0.001
 ALPHA = 0.000

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 [DN006] □ U-19 SPVT RI-0688/139 068 SP/11 ELEVON
 [DN006] □ U-19 SPVT RI-0688/139 068 SP/11 ELEVON
 [DN006] □ U-19 SPVT RI-0688/139 068 SP/11 ELEVON

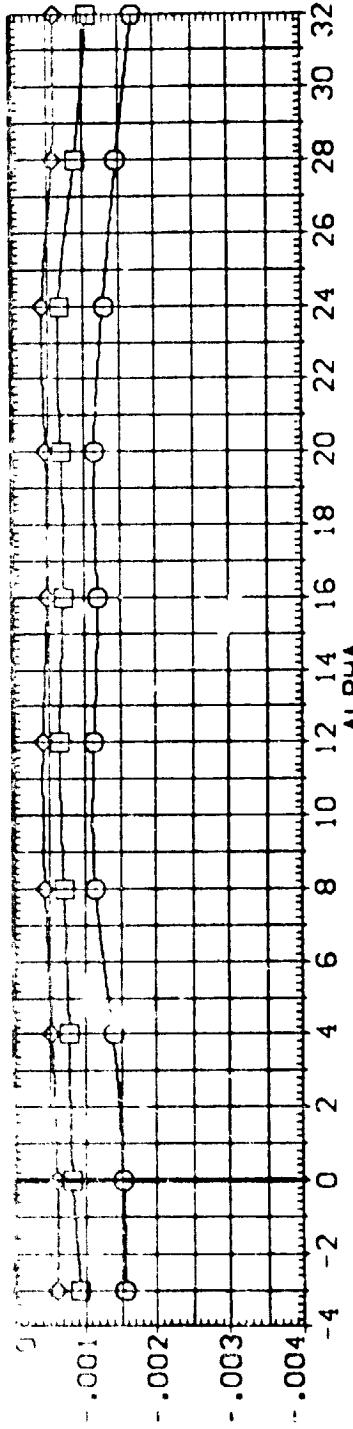
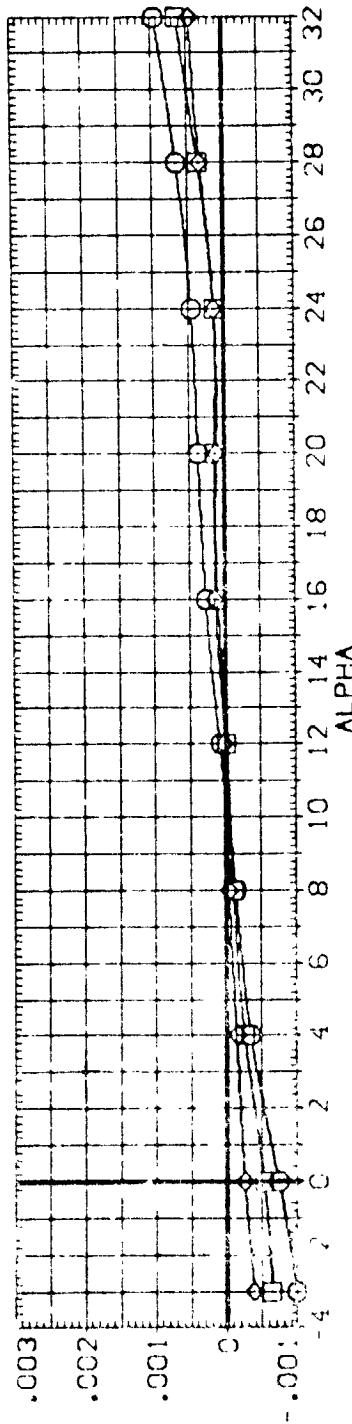
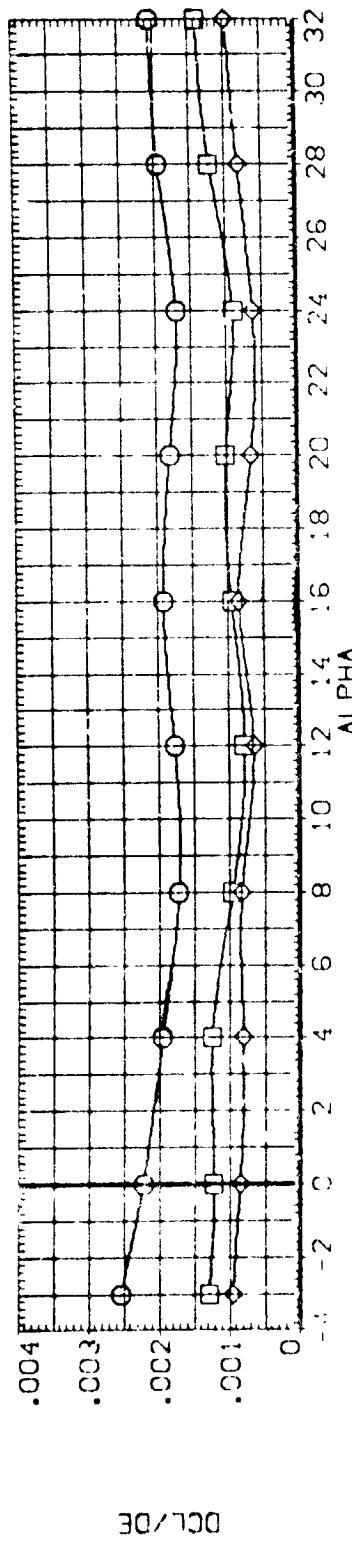


FIGURE 11. COMPARISON OF FULL SPAN, INBOARD AND OUTBOARD PITCH CONTROL (DE=-20)
 MACH = 2.50

REPRODUCED
ORIGINALLY

11/11/81

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DATA SET SYMBOL	CONFIGURATION	DESCRIPTION	ELV-L0	ELV-L1	ELV-RI	ELV-R0
(C) 006	LA-49 SPLIT	101 RI-389B/139	.008 SPLIT ELEVON	-20,000	-20,000	-20,000
(C) 003	LA-49 SPLIT	101 RI-089A/139	.008 SPLIT ELEVON	-20,000	-20,000	-20,000
(C) 009	LA-49 SPLIT	101 RI-089A/139	.008 SPLIT ELEVON	-20,000	-20,000	-20,000

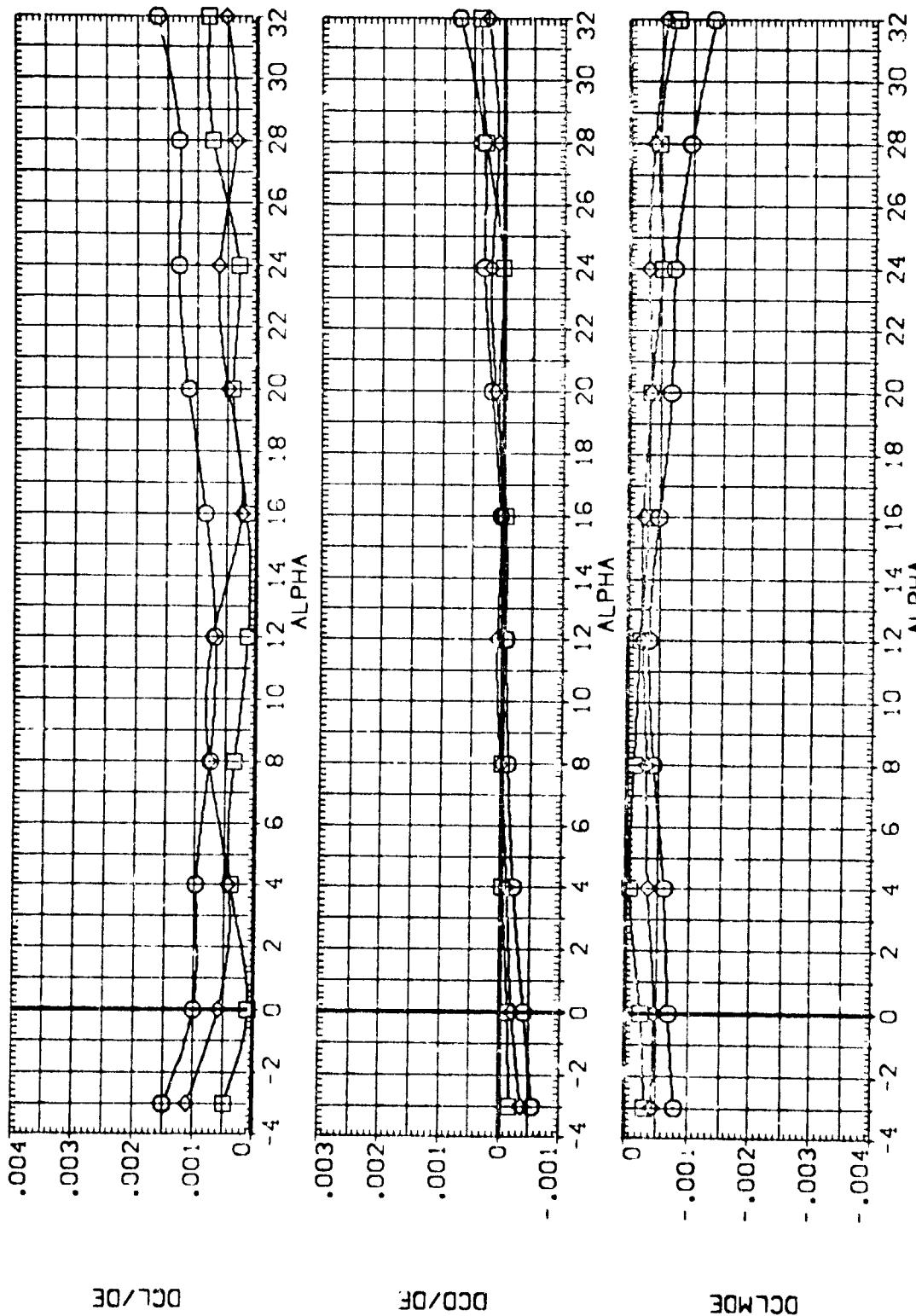


FIGURE 11. COMPARISON OF FULL SPAN, INBOARD AND OUTBOARD PITCH CONTROL (DE=-20)
 $(3)_{MACH} = 4.63$

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 [C] 0071 LA-19 SPOT 1101 RT-0898/139 DCL/DE ELEVON
 [C] 0081 LA-19 SPOT 1101 R-0898/139 DCL/DE ELEVON
 [C] 0091 LA-19 SPOT 1101 R-0898/139 DCL/DE ELEVON
 [C] 0101 LA-19 SPOT 1101 R-0898/139 DCL/DE ELEVON

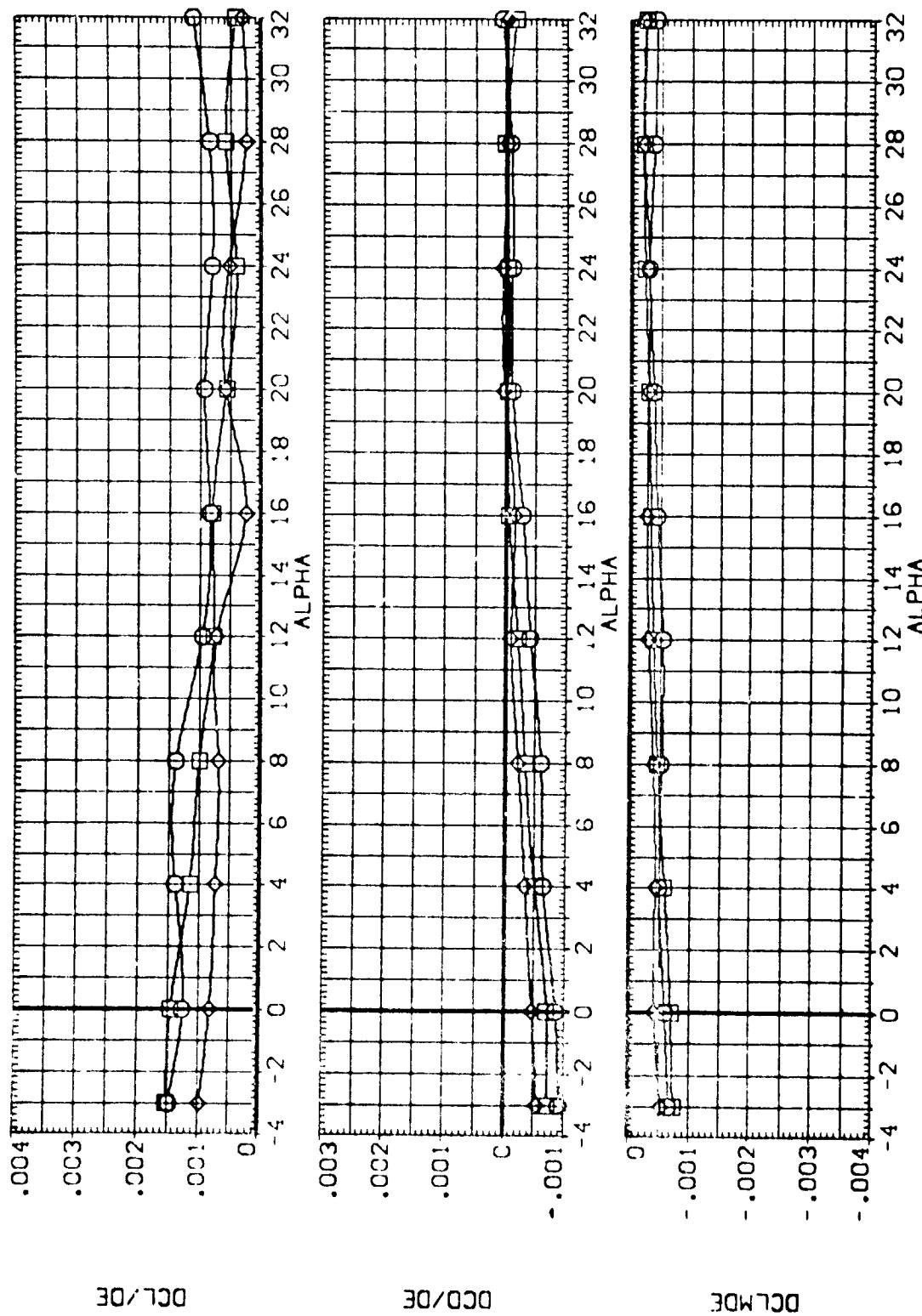


FIGURE 12. COMPARISON OF FULL SPAN, INBOARD AND OUTBOARD PITCH CONTROL (DE = -40)
 $(\lambda)_MACH = 2.50$

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 {CN-000} LA-19 UPV 1101 R1-0898/139 DCL/DE ELEVON -40.000 -40.000 -40.000
 {CN-004} LA-19 UPV 1101 R1-0898/139 DCL/DE ELEVON -40.000 -40.000 -40.000
 {CN-005} LA-19 UPV 1101 R1-0898/139 DCL/DE ELEVON -40.000 .000 .000

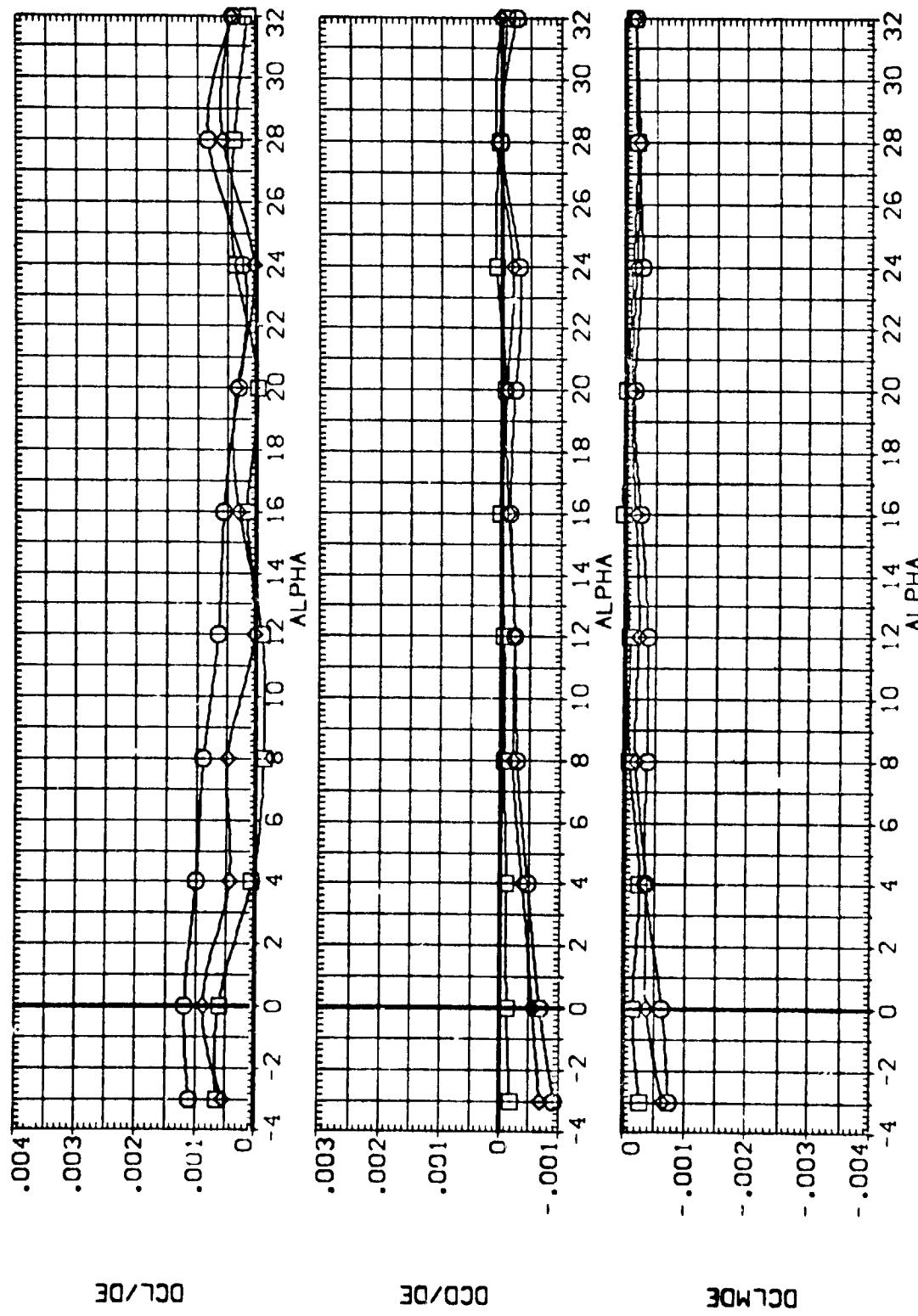


FIGURE 12. COMPARISON OF FULL SPAN. INBOARD AND OUTBOARD PITCH CONTROL ($C_E = -40$)
 $(B)_WACH = 4.63$

DATA SET SW#0016
 CONFIGURATION DESCRIPTION: RI-0893/139 ORB SP-LIT ELEVON
 DATA SET SW#0017
 CONFIGURATION DESCRIPTION: RI-0893/139 ORB SP-LIT ELEVON

	ELV-LI	ELV-RI	ELV-RO
10.000	-20.000	-20.000	-10.000
10.000	-40.000	-40.000	-10.000

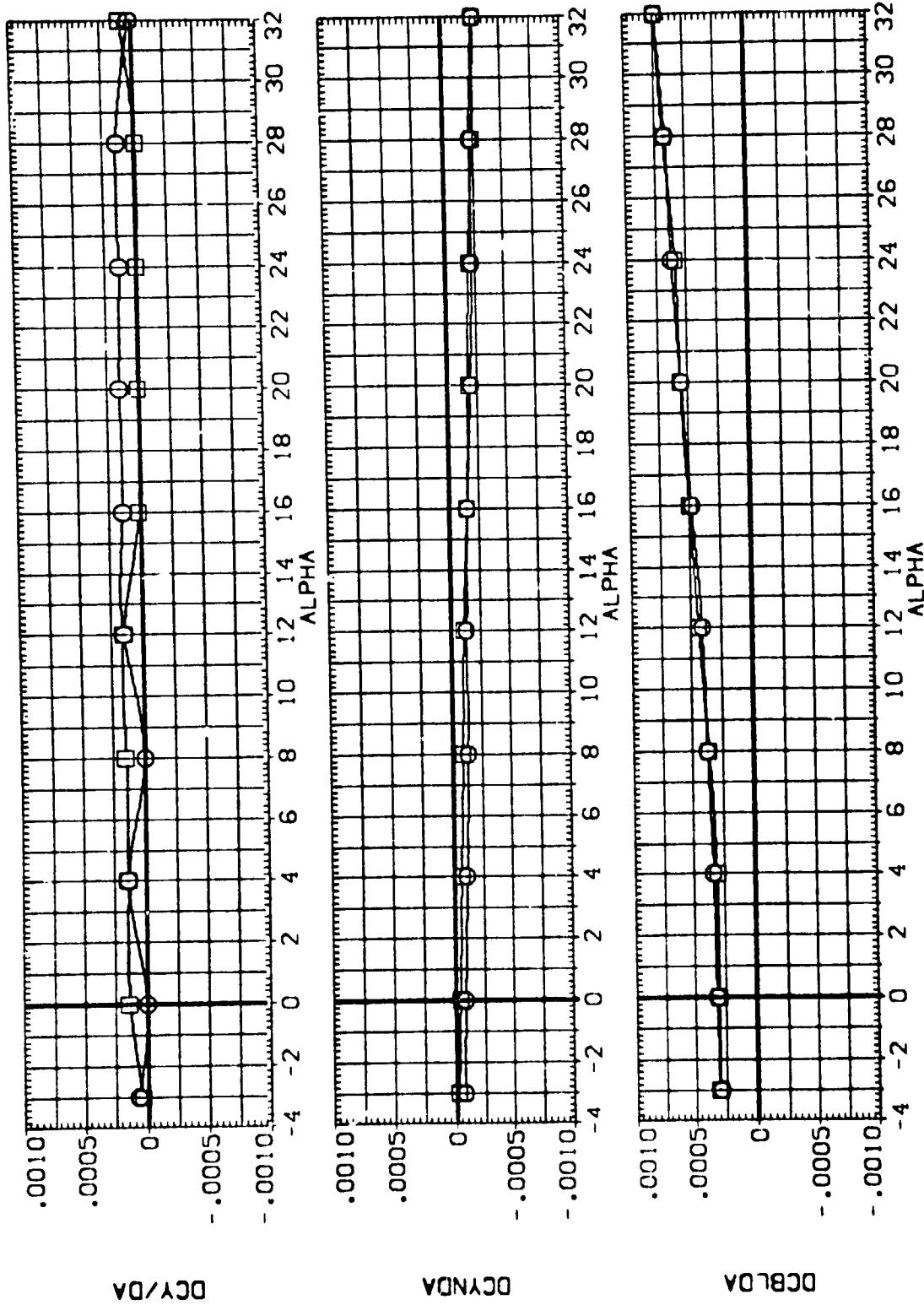


FIGURE 13. OUTBOARD AILERON EFFECTIVENESS WITH INBOARD ELEVONS DEFLECTED
 $(MACH = 2.50)$

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 [3-016] Q LA-19 SPN 1101 RI-0898/139 GRS SPLIT ELEVON
 [3-017] L LA-19 PNT 1101 RI-0898/139 GRS SPLIT ELEVON

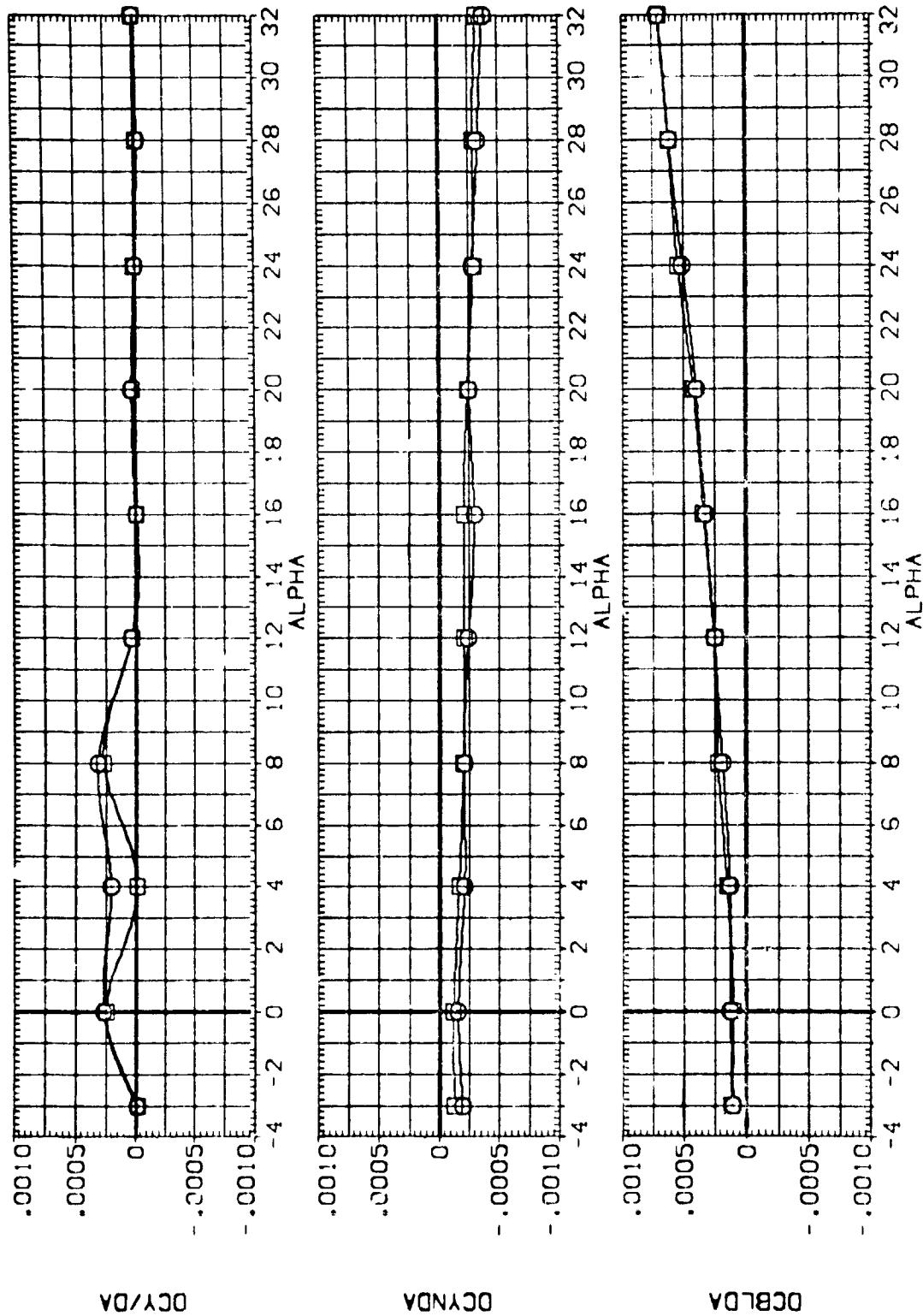


FIGURE 13. OUTBOARD ALILERON EFFECTIVENESS WITH INBOARD ELEVONS DEFLECTED
 $(\beta)_{ACM} = 4.63$

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 {A}011 LA-49 LSP1 101 RI-0898/138 DBB SPLIT ELEVON
 {B}013 LA-49 LSP1 101 RI-0898/138 DBB SPLIT ELEVON

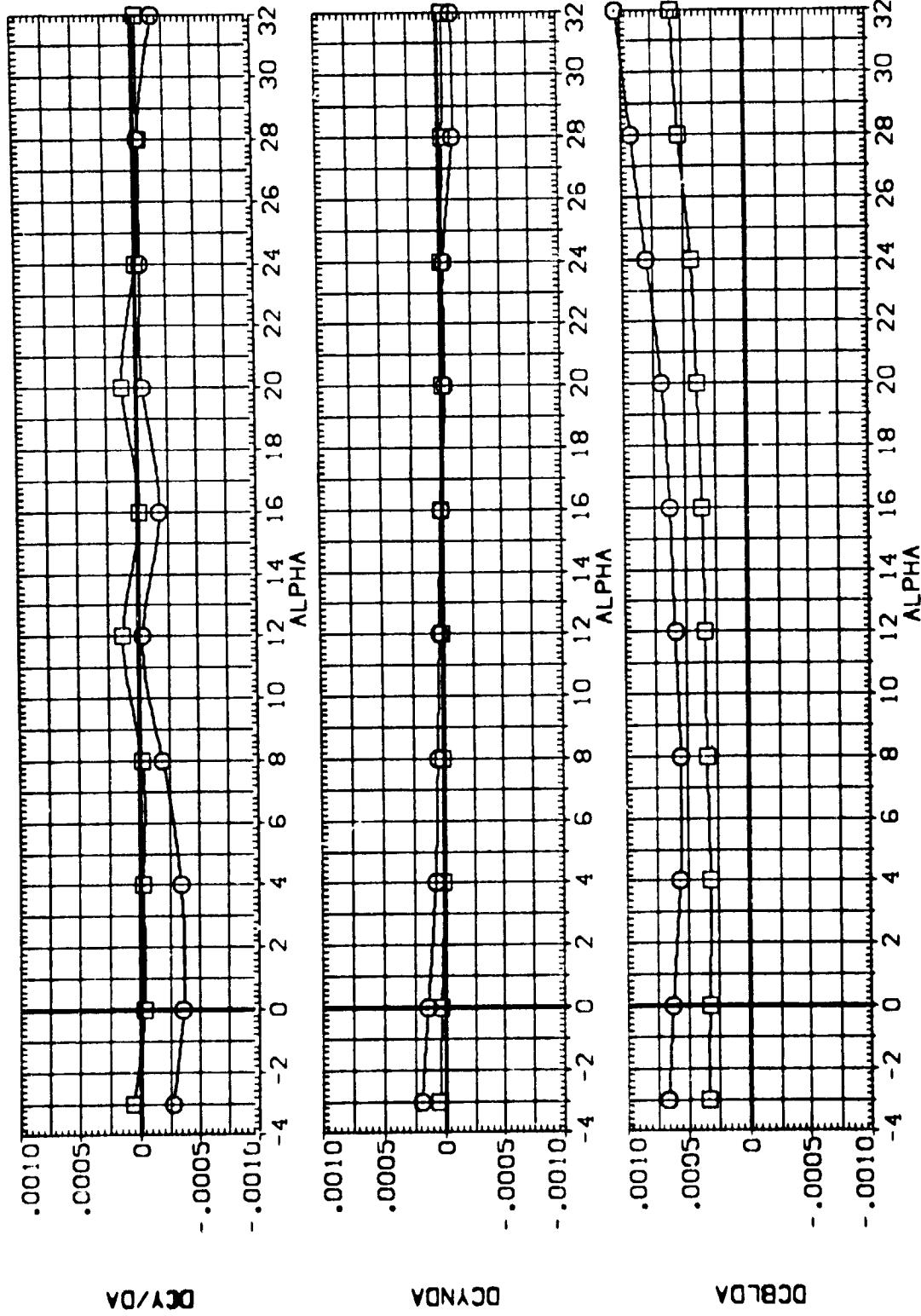


FIGURE 14. COMPARISON OF FULL SPAN AND OUTBOARD ELEVON DEFLECT. FOR ROLL (DE=-10)
 $(\Delta)MACH = 7.50$

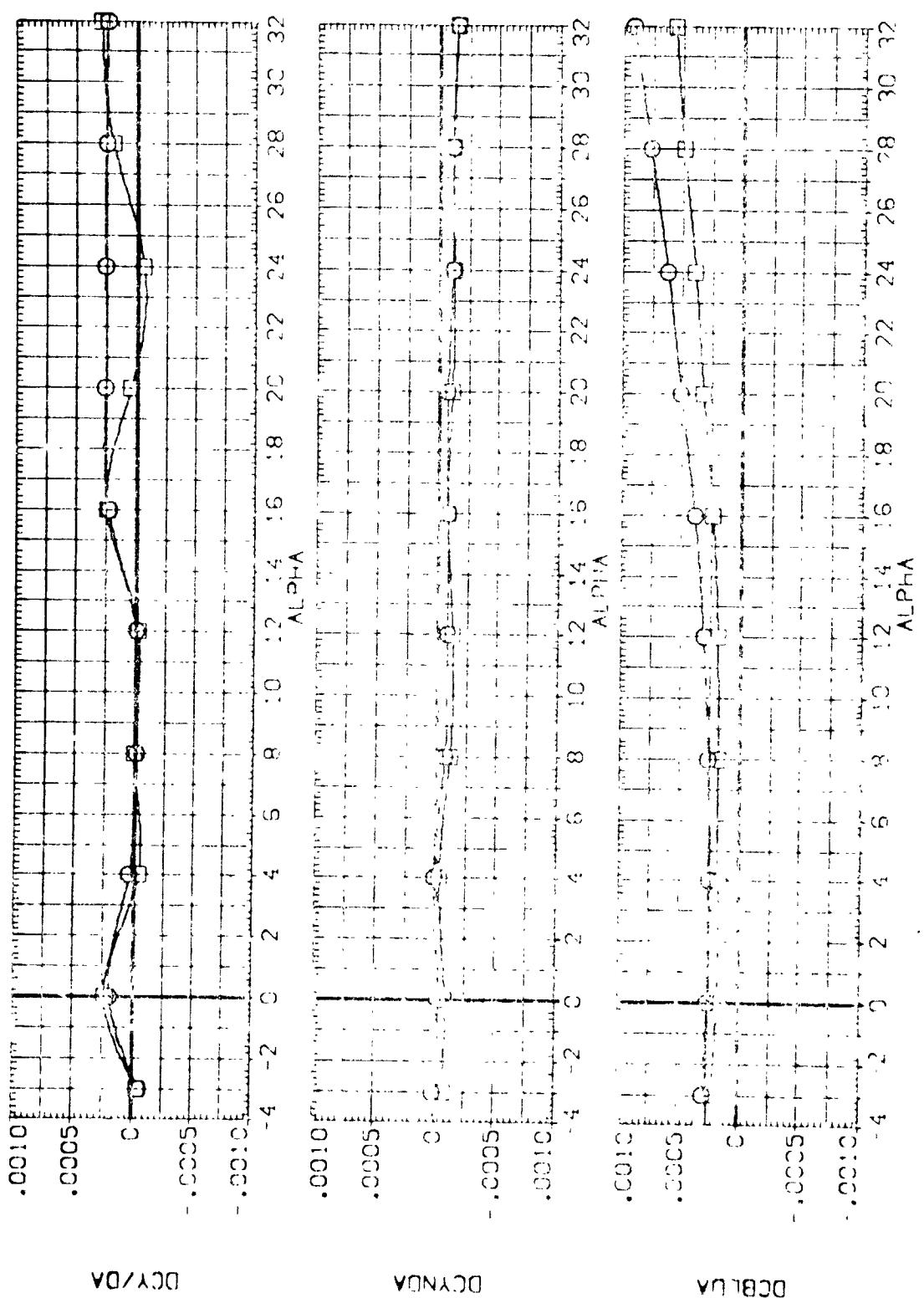


FIGURE 16. SENSITIVITY OF ELEVATION EFFECTS TO VARIOUS ELEVATION CHANGES
 (ELEV = 100)

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CMC = 4.63

DATA SET INDEX	CONFIGURATION DESCRIPTION	ELEV-L0	ELEV-L1	ELEV-R1	ELEV-R0
1	RI 388/15 RI 388/15 RI 388/15 RI 388/15	.000	-10,000	-10,000	-20,000
2	RI 388/15 RI 388/15 RI 388/15 RI 388/15	.000	-10,000	-10,000	-20,000



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (B-4012) C 1-13 2-14 3-15 4-16 5-17 6-18 7-19 8-20 9-21 10-22 11-23 12-24 13-25 14-26 15-27 16-28 17-29 18-30 19-31 20-32 21-33 22-34 23-35 24-36 25-37 26-38 27-39 28-40 29-41 30-42 31-43 32-44 33-45 34-46 35-47 36-48 37-49 38-50 39-51 40-52 41-53 42-54 43-55 44-56 45-57 46-58 47-59 48-60 49-61 50-62 51-63 52-64 53-65 54-66 55-67 56-68 57-69 58-70 59-71 60-72 61-73 62-74 63-75 64-76 65-77 66-78 67-79 68-80 69-81 70-82 71-83 72-84 73-85 74-86 75-87 76-88 77-89 78-90 79-91 80-92 81-93 82-94 83-95 84-96 85-97 86-98 87-99 88-100

ELV-L:	-10.000
ELV-M:	-20.000
ELV-R:	30.000
ELV-U:	-10.000
ELV-V:	-20.000
ELV-W:	30.000
ELV-X:	-10.000
ELV-Y:	-20.000
ELV-Z:	30.000

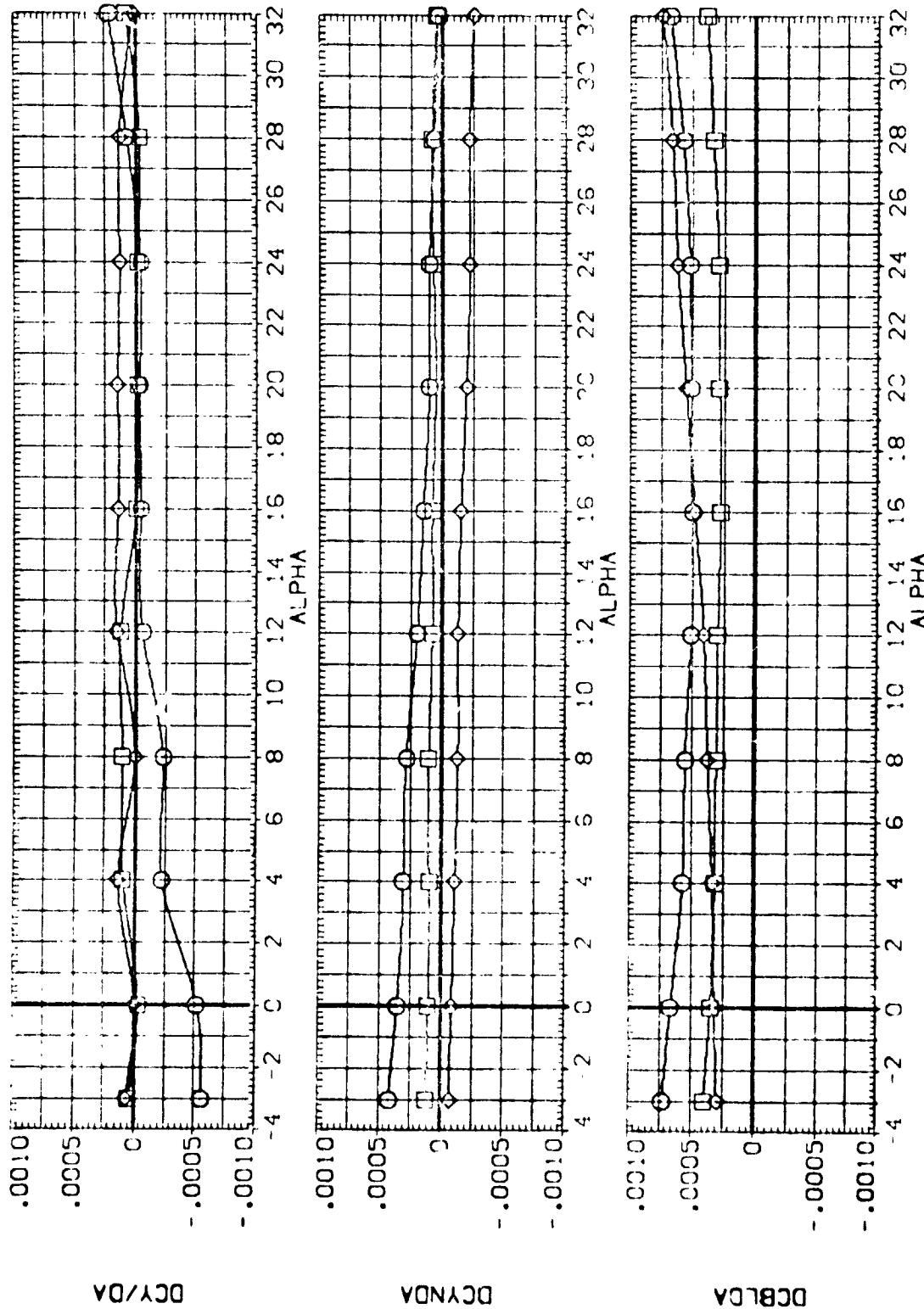


FIGURE 15. COMPARISON OF FULL SPAN AND CUTCARD ELEVON DEFLECT. FOR ROLL (DE=-20)
 $(\text{A})_{\text{MACH}} = 2.50$

DATA SET SPAN: CONFIGURATION DESCRIPTION: ELEV-L0 ELEV-L1 ELEV-R1 ELEV-R0
 1.19 SPAN 1101 R1-0889/138 088 STAT IT ELEVON -10,000 -10,000 -30,000 -30,000
 1.19 SPAN 1102 R1-0889/139 088 STAT IT ELEVON -10,000 -20,000 -20,000 -20,000
 1.19 SPAN 1103 R1-0889/139 088 STAT IT ELEVON 10,000 -20,000 -20,000 -10,000

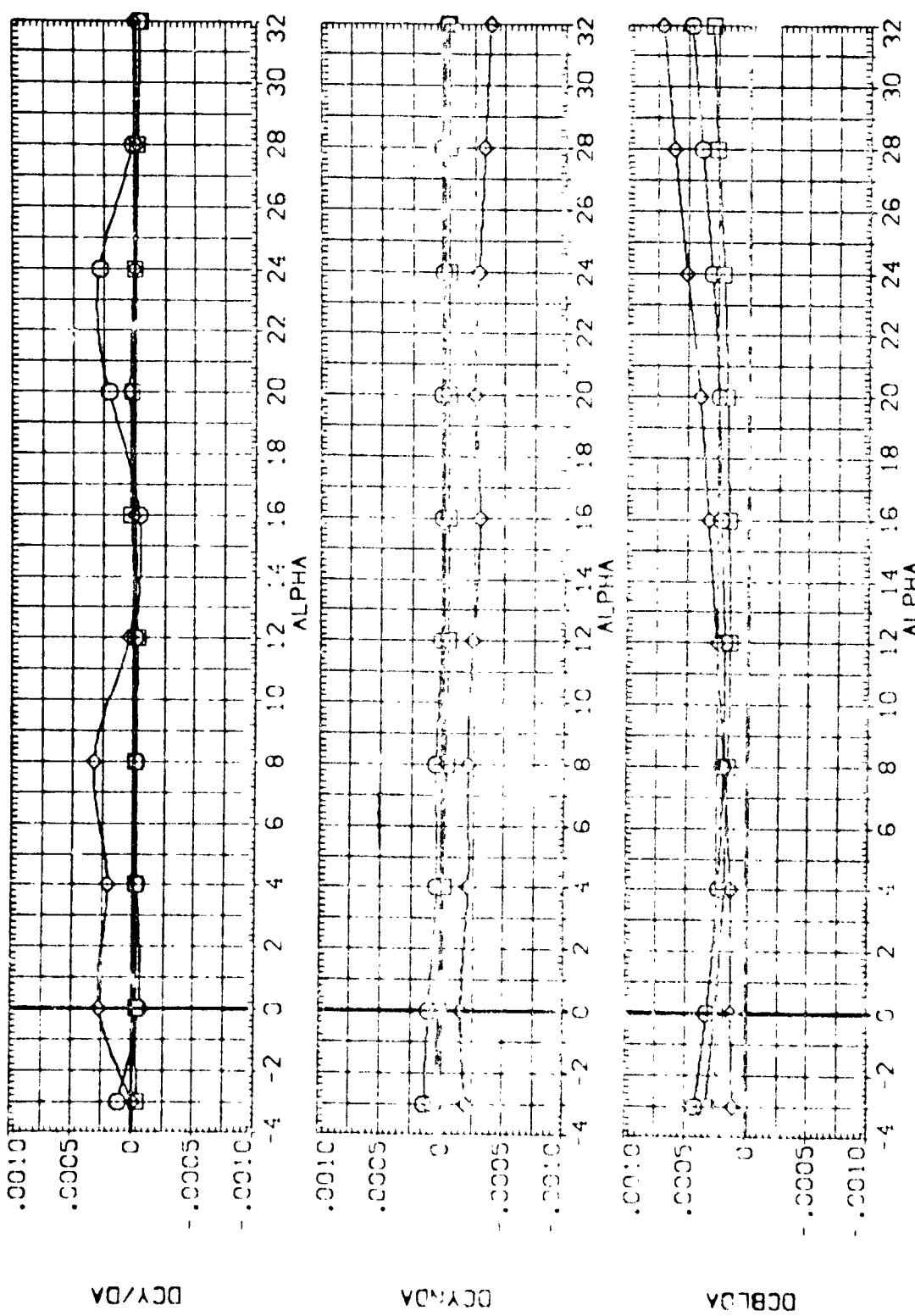


FIGURE 15. COMPARISON OF FULL SPAN AND OUTBOARD ELEVON EFFECT. FOR ROLL ($\Delta E = -20^\circ$)
 (3) w/cn = 4.63

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 [LA-19 U] LA-19 U 1101 RI-0898/139 088 SP/LT ELEVON
 [LA-19 U] LA-19 U 1101 RI-0898/139 088 SP/LT ELEVON
 [LA-19 U] LA-19 U 1101 RI-0898/139 088 SP/LT ELEVON

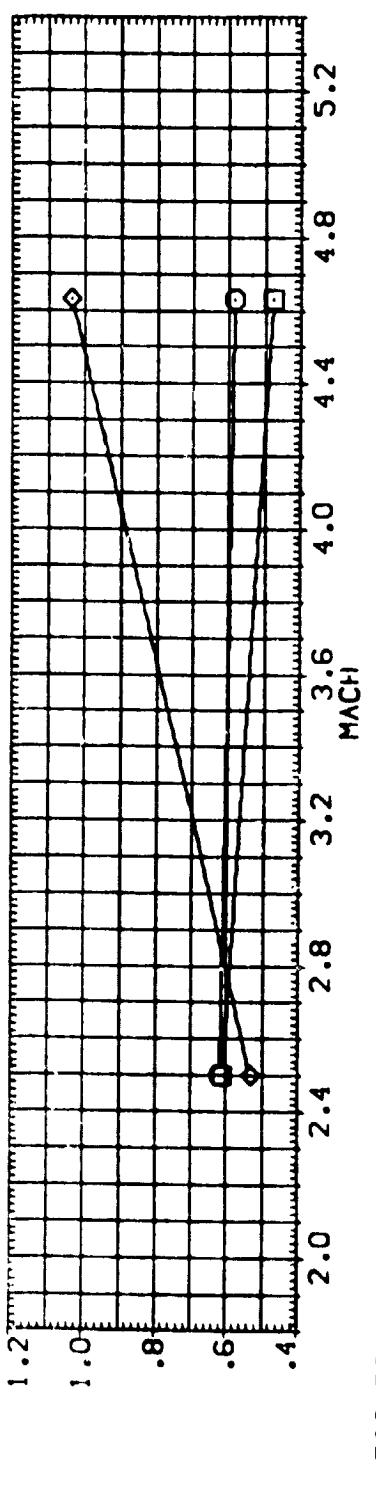
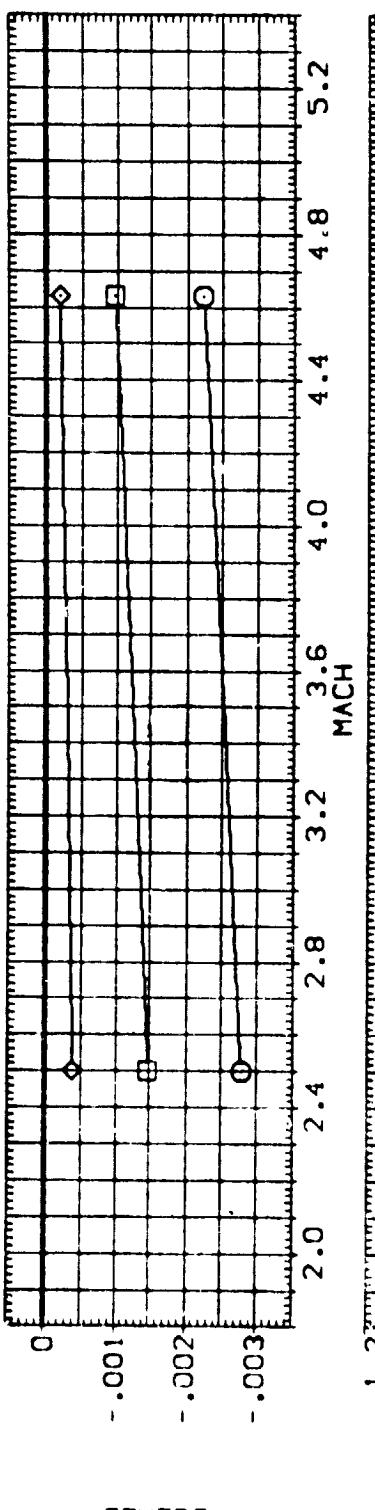
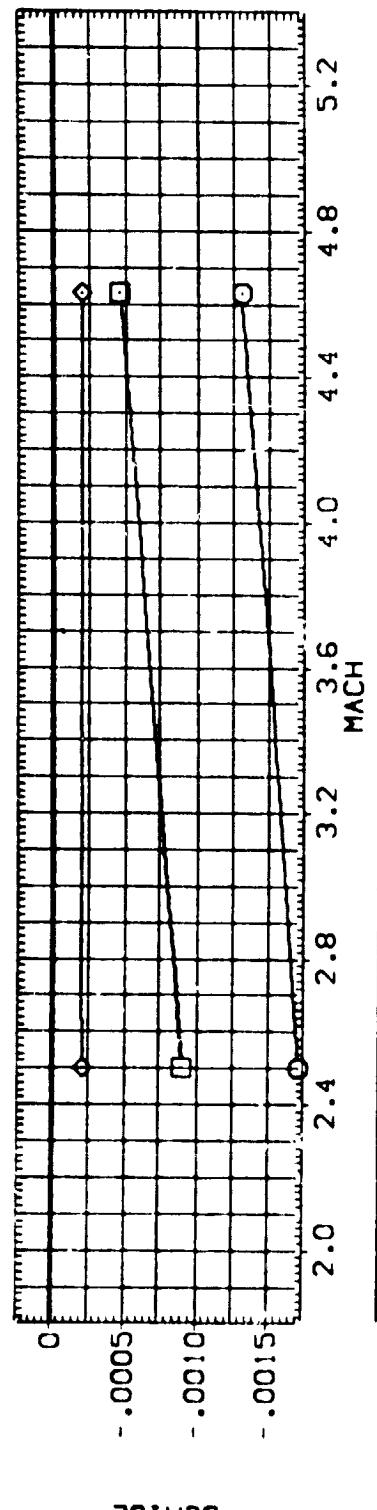
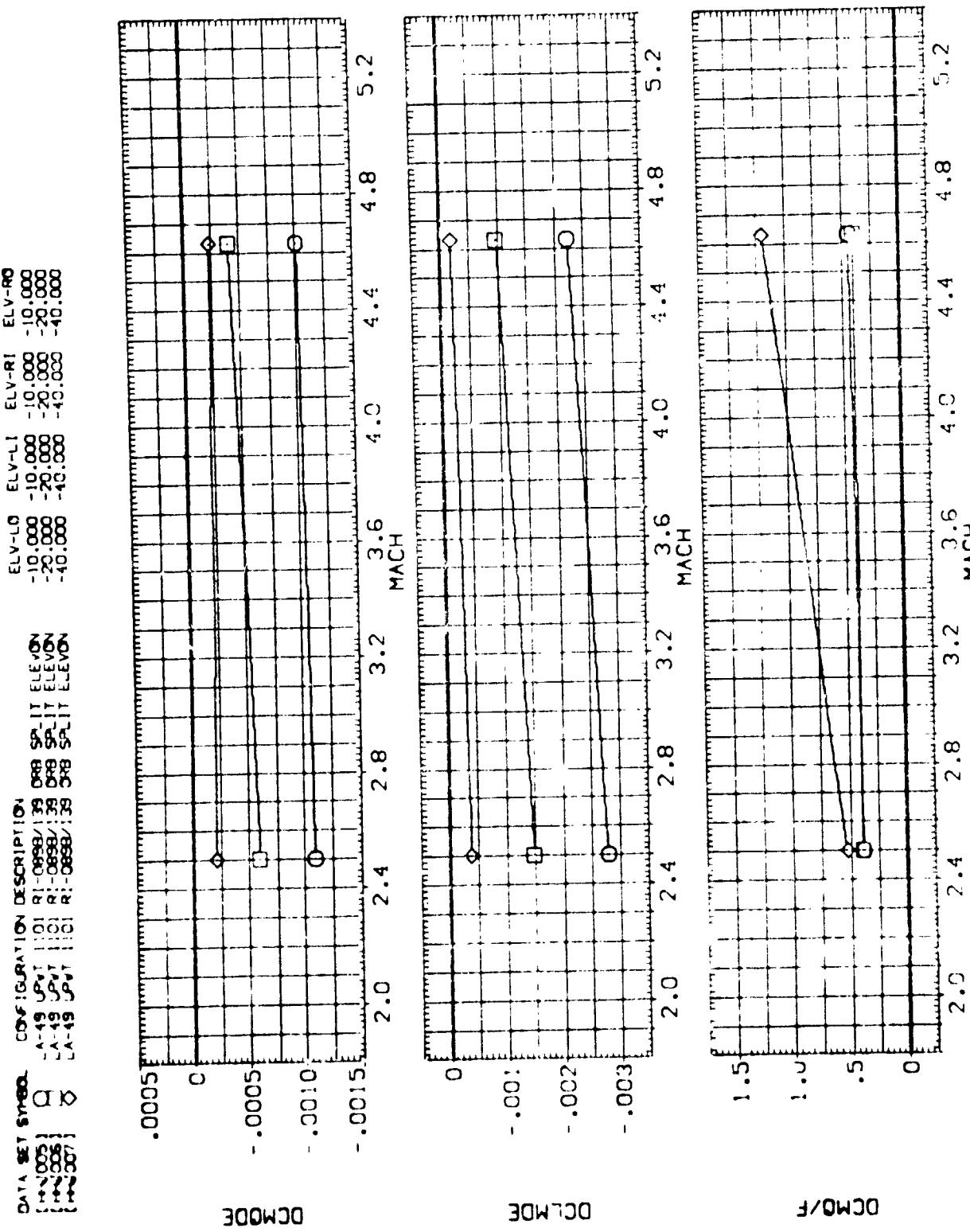


FIGURE 16. COMPARISON OF CONTROL EFFECTIVENESS FOR FULL SPAN AND INBD. ELEVON
 (A)ALPHA = 28.00

FIGURE 17. COMPARISON OF CONTROL EFFECTIVENESS FOR ELEVATOR SPAN AND CANTER, CIRCULAR
 $(\Delta)\alpha_{\text{PA}} = 28.00$

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APPENDIX
TABULATED SOURCE DATA

Plotted data tabulation are available from the Chrysler DATAMAN system
on request.

LA49 TABULATED SOURCE DATA

LA-49 UPNT 1101 R1-0698/139 CFB SPLIT ELEVON

PAGE 1

(RHJ001)

PARAMETRIC DATA

BETA =	.000	ELV-LO =	.000
EV-L1 =	.000	EV-HI =	.000
EV-FWD =	.000	EDFLAP =	.000
SPSFK =	25.000	AIRCON =	.000
EVTR =	.000		

RHO = 26/0

BETA	CN	CA	CD ₀	CD ₁
0.000	1.0142	-0.017	-0.2352	-0.0031
0.125	1.0142	-0.017	-0.2352	-0.0031
0.250	1.0142	-0.017	-0.2352	-0.0031
0.375	1.0142	-0.017	-0.2352	-0.0031
0.500	1.0142	-0.017	-0.2352	-0.0031
0.625	1.0142	-0.017	-0.2352	-0.0031
0.750	1.0142	-0.017	-0.2352	-0.0031
0.875	1.0142	-0.017	-0.2352	-0.0031
1.000	1.0142	-0.017	-0.2352	-0.0031

CD ₂	CD ₃	CD ₄	CD ₅	CD ₆
-0.0031	-0.0031	-0.0031	-0.0031	-0.0031
-0.0031	-0.0031	-0.0031	-0.0031	-0.0031
-0.0031	-0.0031	-0.0031	-0.0031	-0.0031
-0.0031	-0.0031	-0.0031	-0.0031	-0.0031
-0.0031	-0.0031	-0.0031	-0.0031	-0.0031
-0.0031	-0.0031	-0.0031	-0.0031	-0.0031
-0.0031	-0.0031	-0.0031	-0.0031	-0.0031
-0.0031	-0.0031	-0.0031	-0.0031	-0.0031
-0.0031	-0.0031	-0.0031	-0.0031	-0.0031
-0.0031	-0.0031	-0.0031	-0.0031	-0.0031

RHO = 26/0

א-49 מ-101 מ-101 מ-101 מ-101 מ-101 מ-101 מ-101 מ-101

(RH J 002)

EPISTEMIC DATA

LA-49 TABULATED SOURCE DATA

LA-49 UPAT 1101 R1-0898/139 CRB SPLIT ELEMN

PAGE 3

(RHJ003)

PARAMETRIC DATA

BETA =	.000	ELV-LO =	.000
ELV-L1 =	-20.000	ELV-R1 =	-20.000
ELV-R0 =	.000	BDFAP =	.000
SPOERK =	25.000	ELVTR =	-20.000
ATLTON =	.000		

RUN NO. 22/ 0

MACH	ALPHA	BETA	CN	CA	CLM	CDL	CN	CA	CLM	CDL	CN	CA	CLM	CDL	L/D
2.500	-4.043	.00054	-.15578	.15285	-.00292	-.00142	.00023	-.00235	-.14667	.13433	-.09191				
2.500	-.0021	.02026	-.02150	.12053	-.01427	-.00144	.00001	-.00139	-.02146	.12006	-.17874				
2.500	4.089	.00082	.11671	.11624	-.02057	-.00126	.00020	-.00156	.10412	.12426	.07013				
2.500	1.138	.02153	.24933	.11377	-.05005	-.00114	.00021	-.00217	.23072	.4792	.15039				
2.500	12.231	.00088	.15288	.11092	-.04554	-.00115	.00043	-.00142	.35673	.19057	.16723				
2.500	16.276	.02001	.15375	.13056	-.05590	-.00123	.00042	-.00142	.40372	.25453	.190072				
2.500	20.413	.00019	.69296	.10568	-.05205	-.00142	.00020	-.00140	.61249	.34070	.179774				
2.500	24.551	.02145	.05974	.10580	-.07101	-.00125	.00021	-.00124	.73870	.45545	.159874				
2.500	28.686	.02216	.103240	.094075	-.08152	-.00144	-.000370	-.00159	.85082	.50159	.147634				
2.500	32.740	.02146	.121225	.03344	-.07663	-.00141	-.000290	-.00147	.96734	.73344	.131471				

RUN NO. 4/ 0

MACH	ALPHA	BETA	CN	CA	CLM	CDL	CN	CA	CLM	CDL	CN	CA	CLM	CDL	L/D
4.600	-3.537	.10540	-.10289	.00005	.00049	-.00421	.00032	-.00032	.00032	.00032	.00032				
4.600	4.442	.03977	-.06350	.00055	.00005	-.00017	.00145	-.00145	.00145	.00145	.00145				
4.600	2.457	.00035	-.00250	.00457	-.00005	.00005	-.000474	-.000474	.000474	.000474	.000474				
4.600	4.550	.03995	.31181	.00088	-.00009	.00009	-.00035	-.00035	.00035	.00035	.00035				
4.600	8.320	.00017	.00062	.00011	-.00011	-.00011	-.00034	-.00034	.00034	.00034	.00034				
4.600	12.511	.04748	.00033	.00029	-.00029	-.00029	-.00033	-.00033	.00033	.00033	.00033				
4.600	16.597	.00017	.00070	.00019	-.00019	-.00019	-.00036	-.00036	.00036	.00036	.00036				
4.600	20.664	.04642	.00042	.00033	-.00033	-.00033	-.00042	-.00042	.00042	.00042	.00042				
4.600	24.722	.00042	.00022	.00026	-.00026	-.00026	-.00045	-.00045	.00045	.00045	.00045				
4.600	28.797	.04572	.00038	.00022	-.00022	-.00022	-.00057	-.00057	.00057	.00057	.00057				
4.600	32.837	.00051	.000450	.00012	-.00099	-.00099	-.00059	-.00059	.00059	.00059	.00059				
4.600	36.890	.04094	.000203	.00019	-.00233	-.00233	-.00050	-.00050	.00050	.00050	.00050				
4.600	40.955	.00019	.000115	.00002	-.00000	-.00000	-.00085	-.00085	.00085	.00085	.00085				

PARAMETRIC DATA

BETA	= .000	ELV-LO = .000
ELV-L1	= -40.000	ELV-RI = -40.000
ELV-RO	= .000	BDFLAP = .000
SPDRK	= 25.000	ELEVTR = -40.000
ATLRCN	= .000	

RUN NO. 23/ 0

MACH	ALPHA	BETA	CN	CA	CLH	CBL	CYN	CT	CL	CD	L/D
2.500	-4.072	.02054	-1.06637	.13543	.01396	-.00141	.00021	-.00228	-.17658	.14835	-1.19033
2.500	-4.004	.02113	-.04824	.13293	.00066	-.00114	-.00001	-.00259	-.04623	.13293	-.36281
2.500	4.692	.02157	.09402	.12613	-.01544	-.00094	-.00022	-.00269	.08478	.13232	.63973
2.500	6.130	.02217	.23538	.12167	-.02887	-.00091	-.00044	-.00302	.21106	.15306	1.37897
2.500	12.224	.02240	.37379	.11588	-.03695	-.00109	-.00044	-.00338	.34056	.19338	1.76111
2.500	16.315	.02252	.52105	.11333	-.04887	-.00127	-.00044	-.00216	.48224	.23514	1.83522
2.500	20.405	.02245	.68082	.11512	-.05454	-.00137	-.00044	-.00234	.59953	.34090	1.75863
2.500	24.518	.02228	.84530	.10710	-.06329	-.00126	-.00045	-.00234	.72555	.44664	1.61723
2.500	28.632	.02161	1.01544	.10370	-.07231	-.00157	-.00034	-.00202	.84138	.57789	1.45597
2.500	32.770	.02090	1.19972	.09940	-.08383	-.00168	-.00043	-.00266	.95498	.73295	1.30292

FLR. NO. 5/ 0

MACH	ALPHA	BETA	CN	CA	CLH	CBL	CYN	CT	CL	CD	L/D
4.600	-3.524	.04072	-1.11664	.09479	-.02612	-.00015	-.00037	-.00455	-.11059	.10178	-1.08661
4.600	-1.576	.04033	-.07530	.09072	-.02997	-.00002	-.00037	-.00353	-.07277	.09276	-.78457
4.600	4.453	.04087	-.03007	.08733	-.02643	-.00007	-.00038	-.00498	-.03076	.08709	-.35322
4.600	2.465	.04043	.01531	.08398	-.02688	-.00029	-.00039	-.00382	.01168	.08456	.13819
4.600	4.494	.04097	-.06060	-.02801	-.00050	-.00039	-.00039	-.00263	.05416	.08311	.63638
4.600	8.534	.04050	.16715	.07899	-.02920	-.00031	-.00002	-.00021	.15388	.10094	1.32439
4.600	12.570	.03980	.23102	.07750	-.03100	-.00018	-.00041	-.00228	.25746	.13660	1.84470
4.600	16.591	.04052	.40962	.07977	-.03329	-.00054	-.00042	-.00420	.36748	.19168	1.91716
4.600	20.636	.04011	.45112	.08187	-.03492	-.00040	-.00043	-.00316	.48691	.27085	1.79767
4.600	24.778	.04058	.70306	.08467	-.03783	-.00060	-.00046	-.00444	.60311	.37111	1.82517
4.600	28.790	.04131	.96030	.08638	-.04616	-.00080	-.00109	-.00531	.71279	.48975	1.45542
4.600	32.826	.04148	1.03721	.08697	-.05351	-.00061	-.00130	-.00581	.82335	.63703	1.29249
4.600	36.901	.04156	1.21673	.09158	-.06370	-.00071	-.00129	-.00617	.91603	.80381	1.14206
4.600	40.956	.04166	1.39242	.09257	-.07453	-.00082	-.00129	-.00660	.99090	.98200	1.00445



LA49 TABULATED SOURCE DATA

LA-49 UPNT 1101 RI-0898/139 TRS SPLIT ELEVON

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(RH1005)

PARAMETRIC DATA

BETA =	.000	ELV-LO =	-10.000
ELV-L1 =	-10.000	ELV-R1 =	-10.000
ELV-R0 =	-10.000	EFLAP =	.000
SPDRK =	25.000	AIRCON =	.000
ELEVTR =	-10.000		

RUN NO. 24/0

MACH	ALPHA	CN	CM	CL	CD	L/D
2.500	-4.000	.02174	.14981	.00140	-.14095	.13016
2.500	-4.012	.02147	.01750	-.00151	-.00109	-.00287
2.500	4.055	.02158	.11874	.00124	-.00043	-.14929
2.500	6.133	.02190	.25140	.00114	-.00144	.11707
2.500	12.213	.02219	.39014	-.00113	-.00205	.12285
2.500	15.301	.02236	.53800	-.00113	-.00055	.14687
2.500	20.419	.02202	.68355	-.00123	-.00065	.15820
2.500	24.530	.02230	.85396	-.00124	-.00140	.1.0617
2.500	28.648	.02210	.99409	-.00153	-.00044	.1.91598
2.500	32.753	.02144	1.21391	-.00151	-.00281	.25781
					-.00065	.49630
					-.00123	.61369
					-.00040	.33883
					-.00123	.81182
					-.00040	.1.45227
					-.00124	.73811
					-.00044	.44673
					-.00153	.57742
					-.00032	.1.48773
					-.00151	.57147
					-.00208	.1.37427
					-.00151	.75359

RUN NO. 6/0

MACH	ALPHA	CN	CM	CL	CD	L/D
4.500	-3.540	.00065	-.00112	-.00036	-.00397	.09674
4.500	-3.540	.00065	-.00112	-.00037	-.00398	-.59447
4.500	-1.569	.00065	-.00112	-.00037	-.00398	-.56144
4.500	-1.569	.00065	-.00112	-.00037	-.00398	-.58654
4.500	-1.569	.00065	-.00112	-.00037	-.00398	-.58356
4.500	-1.569	.00065	-.00112	-.00037	-.00398	-.27515
4.500	-1.569	.00065	-.00112	-.00037	-.00398	-.04210
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.23574
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.74199
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.08349
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.03932
4.500	-1.569	.00065	-.00112	-.00037	-.00398	1.5
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.23
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.13468
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.91553
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.1.04612
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.1.01010
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.21122
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.49131
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.35131
4.500	-1.569	.00065	-.00112	-.00037	-.00398	1.0
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.3136
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.48997
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.1.47283
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.63768
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.1.30481
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.80381
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.1.15557
4.500	-1.569	.00065	-.00112	-.00037	-.00398	.1.02076

PARAMETRIC DATA

BETA =	.000	ELV-LO =	-20.000
ELV-L1 =	-20.000	ELV-R1 =	-20.000
ELV-HO =	-20.000	EDFLAP =	.000
SPDRK =	.25.000	AIRRCN =	.000
ELVTR =	-20.000		

RUN NO. 25/0

MACH	BETA	CN	CA	CLH	CL	CLN	CT	CD	L/D
2.500	.02011	-1.17009	.12766	.01138	-.00167	-.000002	-.00192	-.15993	-1.20506
2.500	-.018	.02038	.04184	.12272	-.00108	-.00150	-.00024	-.04180	.12273
2.500	4.000	.02225	.09942	.11766	-.01611	-.00131	-.00056	.08984	.12434
2.500	8.134	.02297	.23314	.11406	-.02658	-.00129	-.00091	.00292	.14597
2.500	12.235	.02293	.37010	.11071	-.03404	-.00128	-.00068	.00352	.18683
2.500	16.317	.02216	.51568	.10816	-.04140	-.00138	-.00068	.00245	.18676
2.500	20.401	.02283	.47100	.10487	-.04686	-.00139	-.00045	.00245	.33240
2.500	24.537	.02283	.43314	.10111	-.05199	-.00141	-.00047	.00240	.17817
2.500	28.652	.02256	.17272	.09535	-.06066	-.00161	-.00059	.00373	.163460
2.500	32.750	.02255	.110320	.09158	-.07277	-.00198	-.00045	.00374	.147475

RUN NO. 7/0

MACH	BETA	CN	CA	CLH	CL	CLN	CT	CD	L/D
3.504	.01114	.11164	.03173	-.02402	-.00030	-.00060	-.00450	-.11033	.10075
4.000	-1.576	.04076	.07510	.00991	-.02319	-.00021	-.00040	-.07280	.09194
4.000	4.000	.04031	.02977	.08822	-.02432	-.00042	-.00061	-.00234	.08597
4.000	2.475	.04085	.01590	.08288	-.02410	-.00047	-.00081	.01191	.08347
4.000	4.501	.04043	.01979	.06076	-.02387	-.00052	-.00062	-.00269	.14266
4.000	8.516	.04045	.15971	.07650	-.02445	-.00070	-.00083	-.00278	.04458
4.000	12.555	.02143	.27358	.07553	-.02557	-.00037	-.00126	-.00245	.64223
4.000	16.595	.02120	.39059	.07757	-.02579	-.00057	-.00128	-.00186	.09931
4.000	20.632	.02120	.53999	.08042	-.02540	-.00061	-.00129	-.00158	.1.47635
4.000	24.723	.02234	.60111	.08223	-.02837	-.00080	-.00132	-.00196	.1.86635
4.000	28.769	.02168	.84972	.08354	-.03368	-.00099	-.00154	-.00326	.1.3417
4.000	32.857	.02194	1.01870	.08533	-.04063	-.00100	-.00174	-.00399	.25040
4.000	36.890	.02207	1.19842	.08560	-.05014	-.00128	-.00174	-.00694	.1.95977
4.000	40.999	.02307	1.37796	.08643	-.06034	-.00138	-.00174	-.00729	.2.96112

PARAMETRIC DATA									
RUN NO.	26, 0	BETA	CN	CA	CLM	CYN	CR	CD	L/D
MACH									
1.2, .500	-4.073	.02058	-.21508	.14402	.00137	-.00158	.00018	.15922	-1.30023
1.2, .500	.002	.02160	.07267	.13894	.01713	-.00102	.00019	.13698	-.92291
1.2, .500	4.034	.02145	.06766	.13057	-.00009	.00110	-.00025	.13503	.43143
1.2, .500	6.138	.02268	.20452	.12695	.01166	-.00108	-.00024	.15482	1.19318
1.2, .500	12.234	.02196	.34746	.12072	-.01903	-.00997	-.00024	.16387	1.63871
1.2, .500	16.908	.02214	.49290	.11684	-.02750	-.00103	-.00024	.25036	1.75878
1.2, .500	20.430	.02220	.64928	.11215	-.03409	-.00107	-.00024	.33140	1.71501
1.2, .500	24.534	.02235	.81233	.10871	-.03972	-.00147	-.00049	.43621	1.59063
1.2, .500	28.358	.02239	.97573	.10503	-.04519	-.00149	-.00051	.56203	1.45957
1.2, .500	32.755	.02160	1.15439	.10109	-.05559	-.00149	-.00294	.91613	.29105
RUN NO.									
MACH									
1.1, .600	-3.541	.02568	-.14329	.10876	-.00850	-.00040	-.00548	.11740	-1.16997
1.1, .600	-1.583	.02131	-.09765	.10326	-.00636	-.00011	-.00125	.13630	.89473
1.1, .600	.487	.02160	-.03245	.09761	-.01155	-.00031	-.00125	.09477	.10592
1.1, .600	2.653	.02134	-.00712	.09319	-.01337	-.00032	-.00125	.05324	.54784
1.1, .600	4.468	.02145	-.01178	.09374	-.01510	-.00033	-.00126	.08279	.11582
1.1, .600	8.338	.02181	.14450	.09343	-.01640	-.00057	-.00127	.11372	.37078
1.1, .600	12.565	.02025	.28225	.00284	-.01816	-.00024	-.00127	.103951	1.25546
1.1, .600	16.519	.02141	.38482	.04312	-.01906	-.00061	-.00129	.13771	1.72813
1.1, .600	20.870	.02175	.52034	.02979	-.01955	-.00064	-.00130	.19106	1.82210
1.1, .600	24.724	.02160	.65005	.02055	-.01957	-.00067	-.00133	.26744	1.77398
1.1, .600	28.771	.02201	.83077	.03692	-.02449	-.00085	-.00155	.36390	1.58723
1.1, .600	32.060	.02246	1.00375	.06254	-.02844	-.00088	-.00176	.47956	1.42277
1.1, .600	36.925	.02246	1.17575	.09529	-.03460	-.00100	-.00176	.78255	1.32994
1.1, .600	40.568	.02246	1.35153	.07150	-.03719	-.00114	-.00176	.82269	.05053

LA-49 TABULATED SOURCE DATA

LA-49 UPUT 1101 RL-0898/139 CRB SPLIT ELEMN

(RHJ008)

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PARAMETRIC DATA

BETA =	.000	EL-V-LO =	-10,000
EL-V-L1 =	.000	EL-V-R1 =	.000
EL-V-RO =	-10,000	20FLAP =	.000
SPDTR =	25,000	ELEVTR =	-10,000
ATLRDN =	.000		

RUN NO. 27/0

MACH	ALPHA	BETA	CN	CA	CLW	CBL	CYN	CY	CL	CD	L/D
2.500	-4.089	.01970	-.13744	.11899	-.01373	-.00155	-.00020	.00004	-.12870	.12784	-1.00666
2.500	-.023	.02139	-.00542	.11789	-.02705	-.00148	-.00043	-.00185	-.00537	.11789	-.04599
2.500	4.082	.02242	.013073	.11657	-.04070	-.00150	-.00087	-.00216	.12216	.12234	.97466
2.500	6.133	.02277	.26532	.11442	-.05046	-.00110	-.00087	-.00214	.24646	.15081	1.53432
2.500	12.251	.23638	.40438	.11191	-.05826	-.00100	-.00087	-.00324	.37143	.19317	1.90312
2.500	16.314	.02316	.55172	.10971	-.06891	-.00127	-.00086	-.00340	.49869	.26028	1.91598
2.500	20.328	.02276	.70958	.10437	-.07864	-.00128	-.00083	-.00347	.62852	.34347	1.81933
2.500	24.322	.02349	.87738	.10287	-.08727	-.00109	-.00064	-.00459	.75553	.45774	1.65056
2.500	28.633	.02253	1.03405	.09831	-.10211	-.00136	-.00052	-.00387	.87904	.59139	1.48472
2.500	32.749	.02274	1.24166	.09543	-.11874	-.00134	-.00060	-.00425	.99266	.75196	1.32011

RUN NO. 9/0

MACH	ALPHA	BETA	CN	CA	CLW	CBL	CYN	CY	CL	CD	L/D
4.000	-3.824	.02152	-.09785	.09144	-.05342	-.00011	-.00078	-.00371	-.09204	.09728	-1.94616
4.000	-1.561	.02119	-.06015	.08763	-.03263	-.00017	-.00079	-.00284	-.05774	.08924	-.64704
4.000	-.454	.02118	-.01640	.08481	-.03134	-.00021	-.00121	-.00171	-.01949	.08465	-.23025
4.000	2.473	.02177	.02281	.08217	-.03232	-.00044	-.00123	-.00328	.08924	.08308	-.23162
4.000	4.491	.02131	.06015	.07963	-.03278	-.00048	-.00123	-.00213	.06171	.08472	-.72834
4.000	8.324	.02129	.17452	.07673	-.03322	-.00050	-.00124	-.00205	.16122	.10175	1.58444
4.000	12.547	.02154	.28831	.07705	-.03640	-.00016	-.00167	-.00166	.26468	.13784	1.92019
4.000	16.650	.02231	.411076	.07887	-.04076	-.00034	-.00549	-.00277	.37801	.19321	1.93639
4.000	20.667	.02170	.56769	.08151	-.04643	-.00018	-.00168	-.00219	.50052	.27591	1.81404
4.000	24.729	.02208	.72118	.08379	-.05345	-.00019	-.00170	-.00327	.62000	.37780	1.64106
4.000	28.786	.02230	.88520	.08628	-.06312	-.00018	-.00191	-.00391	.73515	.50235	1.46341
4.000	32.831	.02233	1.06634	.08817	-.07932	-.00021	-.00210	-.00412	.84784	.65280	1.29976
4.000	36.906	.02229	1.25336	.09119	-.09423	-.00008	-.00209	-.00413	.94745	.82557	1.14764
4.000	40.956	.02317	1.44028	.09244	-.11190	-.00015	-.00208	-.00425	1.02709	1.01391	1.01300

LA-49 URGENT 1101 R1-0098/139 CRR SPLIT ELEVON

(RHS JOC001)

RUN NO.		28/ 0				10/ 0				10/ 0			
	BETA	CN	CA	CLW	CDL	CN	CA	CLW	CDL	CN	CA	CLW	CDL
ALPHA	.52004	-.14747	.12122	-.09913	-.00168	-.00042	-.13649	-.13136	L/D	.000	ELV-LO	-.20.000	
	-.0173	-.01348	-.11961	-.02126	-.00151	-.00044	-.01346	-.11961		.000	ELV-R1	-.000	
MACH	2.500	4.052	4.052	1.12276	-.03483	-.00142	-.00098	-.00115	-1.12249	.000	BDFLAP	-.20.000	
	2.500	6.143	6.143	1.25703	-.04491	-.00112	-.00028	-.00171	.90578	.000	ELVTR	-.20.000	
	12.222	.02235	.39534	1.11269	-.03274	-.00121	-.00098	-.00360	1.56051	.000	AIRCON	25.000	
	15.314	.02253	.54124	1.11028	-.02677	-.00120	-.00087	-.00246	1.47531	.000			
	20.406	.02368	.70082	1.0505	-.07053	-.00149	-.00064	-.00392	1.89786	.000			
	24.533	.02241	.68694	1.0337	-.08395	-.00130	-.00042	-.00360	1.80915	.000			
	28.631	.02296	1.0134	.09874	-.09363	-.00139	-.00053	-.00445	1.64272	.000			
	32.754	.02320	1.22173	.179808	-.11033	-.00147	-.00062	-.00487	1.47953	.000			
									1.31603	.000			
									.74454	.000			
									.90737	.000			
										.000			
ALPHA	-.02207	-.0921	.09361	-.03086	-.00048	-.00121	-.00402	-.10321	L/D	.000	ELV-LO	-.20.000	
	-.547	-.567	-.657	-.02861	-.00053	-.00122	-.00306	-.06513		.000	ELV-R1	-.000	
MACH	4.000	4.484	4.484	-.52222	-.02839	-.00058	-.00123	-.00193	-1.71773	.000	BDFLAP	-.20.000	
	4.000	2.481	2.481	.01923	-.02885	-.00062	-.00123	-.00193	-2.56762	.000	ELVTR	-.20.000	
	4.000	4.490	4.490	.08461	-.02867	-.00067	-.00124	-.00239	1.87174	.000	AIRCON	25.000	
	5.526	.02241	.15744	.37716	-.02922	-.00089	-.00125	-.00236	1.88555	.000			
	12.548	.02165	.28101	.07754	-.03303	-.00052	-.00167	-.00192	1.92812	.000			
	16.623	.02091	.41449	.07899	-.03660	-.00071	-.00126	-.00113	1.92812	.000			
	20.650	.02113	.52557	.08153	-.04098	-.00072	-.00127	-.00124	1.92812	.000			
	24.700	.02245	.71556	.152930	-.04212	-.00073	-.00171	-.00168	1.92812	.000			
	28.781	.02244	.80592	.08347	-.05837	-.00090	-.00193	-.00245	1.46115	.000			
	32.851	.02254	1.05371	.08876	-.07056	-.00069	-.00212	-.00459	1.29428	.000			
	36.891	.02253	1.23906	.09048	-.08402	-.00098	-.00211	-.00474	1.14733	.000			
	40.972	.02245	1.42602	.10031	-.10031	-.00106	-.00210	-.00470	1.00421	.000			

PARAMETRIC DATA

BETA =	.000	ELV-L0 =	-40.000
ELV-L1 =	.000	ELV-RL =	.000
ELV-RO =	-40.000	EDFLAP =	.120
SPDRK =	25.000	EL.EVTR =	-40.000
ATLFCN =	.000		

RUN NO. 29/ 0

MACH	ALPHA	BETA	CN	CA	CLW	CDL	CL	CD	L/D
2.500	-4.062	.51994	-.16612	.13038	.00287	.00126	-.00146	.14196	-1.11623
2.500	-.013	.02186	-.02992	.12799	-.01141	-.00127	-.00022	.02699	-.23425
2.500	4.050	.02146	.10821	.12425	-.02535	-.00090	-.00044	.00197	.75382
2.500	6.134	.02177	.24285	.12070	-.03660	-.00068	-.00043	.00248	.1.45156
2.500	12.237	.02299	.36398	.11679	-.04587	-.00068	-.00042	.00429	.1.79229
2.500	16.324	.02209	.53346	.11373	-.05770	-.00076	-.00042	.00303	.1.65259
2.500	20.410	.02308	.68963	.10821	-.06347	-.00105	-.00042	.00450	.1.77946
2.500	24.500	.02337	.95432	.10562	-.07400	-.00105	-.00020	.00557	.45339
2.500	28.613	.02196	1.03210	.10131	-.08676	-.00123	-.00031	.00360	.1.62479
2.500	32.732	.02326	1.21547	.09968	-.10271	-.00103	-.00040	.00554	.1.74107
								.00016	1.30697

RUN NO. 11/ 0

MACH	ALPHA	BETA	CN	CA	CLW	CDL	CL	CD	L/D
4.000	-3.526	.00102	-.12401	.10423	-.01664	-.00019	-.00081	.00231	.11176
4.000	-1.550	.02052	-.08221	.09922	-.01830	-.00024	-.00081	.00128	.07996
4.000	-.467	.12. 4	-.03746	.09398	-.02030	-.00027	-.00124	.00260	.03622
4.000	2.479	.02119	.00793	.08961	-.02075	-.00049	-.00124	.00144	.00404
4.000	4.495	.02149	.05799	.08632	-.02169	-.00053	-.00123	.00278	.05014
4.000	6.517	.02158	.10699	.08136	-.02454	-.00071	-.00126	.00280	.1.4231
4.000	12.568	.02176	.27775	.04080	-.02832	-.00037	-.00168	.00219	.25332
4.000	16.589	.02134	.40773	.08199	-.03263	-.00073	-.00127	.00145	.1.86392
4.000	20.627	.02165	.54137	.09446	-.03702	-.00057	-.00170	.00285	.44643
4.000	24.700	.02174	.75751	.09707	-.04334	-.00075	-.00172	.00193	.1.77696
4.000	28.759	.02264	.86932	.08960	-.05232	-.00075	-.00193	.00476	.1.61651
4.000	32.855	.02173	1.05017	.09215	-.06445	-.00074	-.00213	.00240	.64714
4.000	36.893	.02270	1.23373	.09508	-.07807	-.00084	-.00212	.00516	.1.28597
4.000	40.970	.02263	1.41639	.09643	-.06354	-.00075	-.00211	.00516	.81672
								1.00773	1.00492

LA-49 TABULATED SOURCE DATA

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LA-49 UPDAT 11G1 RI-0696/139 CR8 SPLIT ELEVON

(RHJ011)

PARAMETRIC DATA

RUN NO.	SCV 0	PARAMETRIC DATA					
		BETA = .000	EL-V-LO = -.020,000	EL-V-HI = .020,000	BDFLAP = .000	ALFCN = 10,000	L/D = -1.07385
4034	ALPHA -4.500	BETA A .151631	CN -.152255	CLM -.122271	CDL .50211	CL -.142994	L/D -.13317
2	2.500	2.514	2.515	2.5175	2.527	2.528	2.529
2	2.505	4.0237	4.0242	4.0246	4.0249	4.0252	4.0257
2	2.507	4.0239	4.0244	4.0248	4.0251	4.0254	4.0258
2	2.510	4.0241	4.0246	4.0250	4.0253	4.0256	4.0260
2	2.512	4.0243	4.0248	4.0252	4.0255	4.0258	4.0262
2	2.515	4.0245	4.0250	4.0254	4.0257	4.0260	4.0264
2	2.517	4.0247	4.0252	4.0256	4.0259	4.0262	4.0266
2	2.520	4.0249	4.0254	4.0258	4.0261	4.0264	4.0268
2	2.522	4.0251	4.0256	4.0260	4.0263	4.0266	4.0270
2	2.524	4.0253	4.0258	4.0262	4.0265	4.0268	4.0272
2	2.526	4.0255	4.0260	4.0264	4.0267	4.0270	4.0274
2	2.528	4.0257	4.0262	4.0266	4.0269	4.0272	4.0276
2	2.530	4.0259	4.0264	4.0268	4.0271	4.0274	4.0278
2	2.532	4.0261	4.0266	4.0270	4.0273	4.0276	4.0280
2	2.534	4.0263	4.0268	4.0272	4.0275	4.0278	4.0282
2	2.536	4.0265	4.0270	4.0274	4.0277	4.0280	4.0284
2	2.538	4.0267	4.0272	4.0276	4.0279	4.0282	4.0286
2	2.540	4.0269	4.0274	4.0278	4.0281	4.0284	4.0288
2	2.542	4.0271	4.0276	4.0280	4.0283	4.0286	4.0290
2	2.544	4.0273	4.0278	4.0282	4.0285	4.0288	4.0292
2	2.546	4.0275	4.0280	4.0284	4.0287	4.0290	4.0294
2	2.548	4.0277	4.0282	4.0286	4.0289	4.0292	4.0296
2	2.550	4.0279	4.0284	4.0288	4.0291	4.0294	4.0298
2	2.552	4.0281	4.0286	4.0290	4.0293	4.0296	4.0300
2	2.554	4.0283	4.0288	4.0292	4.0295	4.0298	4.0302
2	2.556	4.0285	4.0290	4.0294	4.0297	4.0300	4.0304
2	2.558	4.0287	4.0292	4.0296	4.0299	4.0302	4.0306
2	2.560	4.0289	4.0294	4.0298	4.0301	4.0304	4.0308
2	2.562	4.0291	4.0296	4.0300	4.0303	4.0306	4.0310
2	2.564	4.0293	4.0298	4.0302	4.0305	4.0308	4.0312
2	2.566	4.0295	4.0300	4.0304	4.0307	4.0310	4.0314
2	2.568	4.0297	4.0302	4.0306	4.0309	4.0312	4.0316
2	2.570	4.0299	4.0304	4.0308	4.0311	4.0314	4.0318
2	2.572	4.0301	4.0306	4.0310	4.0313	4.0316	4.0320
2	2.574	4.0303	4.0308	4.0312	4.0315	4.0318	4.0322
2	2.576	4.0305	4.0310	4.0314	4.0317	4.0320	4.0324
2	2.578	4.0307	4.0312	4.0316	4.0319	4.0322	4.0326
2	2.580	4.0309	4.0314	4.0318	4.0321	4.0324	4.0328
2	2.582	4.0311	4.0316	4.0320	4.0323	4.0326	4.0330
2	2.584	4.0313	4.0318	4.0322	4.0325	4.0328	4.0332
2	2.586	4.0315	4.0320	4.0324	4.0327	4.0330	4.0334
2	2.588	4.0317	4.0322	4.0326	4.0329	4.0332	4.0336
2	2.590	4.0319	4.0324	4.0328	4.0331	4.0334	4.0338
2	2.592	4.0321	4.0326	4.0330	4.0333	4.0336	4.0340
2	2.594	4.0323	4.0328	4.0332	4.0335	4.0338	4.0342
2	2.596	4.0325	4.0330	4.0334	4.0337	4.0340	4.0344
2	2.598	4.0327	4.0332	4.0336	4.0339	4.0342	4.0346
2	2.600	4.0329	4.0334	4.0338	4.0341	4.0344	4.0348
2	2.602	4.0331	4.0336	4.0340	4.0343	4.0346	4.0350
2	2.604	4.0333	4.0338	4.0342	4.0345	4.0348	4.0352
2	2.606	4.0335	4.0340	4.0344	4.0347	4.0350	4.0354
2	2.608	4.0337	4.0342	4.0346	4.0349	4.0352	4.0356
2	2.610	4.0339	4.0344	4.0348	4.0351	4.0354	4.0358
2	2.612	4.0341	4.0346	4.0350	4.0353	4.0356	4.0360
2	2.614	4.0343	4.0348	4.0352	4.0355	4.0358	4.0362
2	2.616	4.0345	4.0350	4.0354	4.0357	4.0360	4.0364
2	2.618	4.0347	4.0352	4.0356	4.0359	4.0362	4.0366
2	2.620	4.0349	4.0354	4.0358	4.0361	4.0364	4.0368
2	2.622	4.0351	4.0356	4.0360	4.0363	4.0366	4.0370
2	2.624	4.0353	4.0358	4.0362	4.0365	4.0368	4.0372
2	2.626	4.0355	4.0360	4.0364	4.0367	4.0370	4.0374
2	2.628	4.0357	4.0362	4.0366	4.0369	4.0372	4.0376
2	2.630	4.0359	4.0364	4.0368	4.0371	4.0374	4.0378
2	2.632	4.0361	4.0366	4.0370	4.0373	4.0376	4.0380
2	2.634	4.0363	4.0368	4.0372	4.0375	4.0378	4.0382
2	2.636	4.0365	4.0370	4.0374	4.0377	4.0380	4.0384
2	2.638	4.0367	4.0372	4.0376	4.0379	4.0382	4.0386
2	2.640	4.0369	4.0374	4.0378	4.0381	4.0384	4.0388
2	2.642	4.0371	4.0376	4.0380	4.0383	4.0386	4.0390
2	2.644	4.0373	4.0378	4.0382	4.0385	4.0388	4.0392
2	2.646	4.0375	4.0380	4.0384	4.0387	4.0390	4.0394
2	2.648	4.0377	4.0382	4.0386	4.0389	4.0392	4.0396
2	2.650	4.0379	4.0384	4.0388	4.0391	4.0394	4.0398
2	2.652	4.0381	4.0386	4.0390	4.0393	4.0396	4.0400
2	2.654	4.0383	4.0388	4.0392	4.0395	4.0398	4.0402
2	2.656	4.0385	4.0390	4.0394	4.0397	4.0400	4.0404
2	2.658	4.0387	4.0392	4.0396	4.0399	4.0402	4.0406
2	2.660	4.0389	4.0394	4.0398	4.0401	4.0404	4.0408
2	2.662	4.0391	4.0396	4.0400	4.0403	4.0406	4.0410
2	2.664	4.0393	4.0398	4.0402	4.0405	4.0408	4.0412
2	2.666	4.0395	4.0400	4.0404	4.0407	4.0410	4.0414
2	2.668	4.0397	4.0402	4.0406	4.0409	4.0412	4.0416
2	2.670	4.0399	4.0404	4.0408	4.0411	4.0414	4.0418
2	2.672	4.0401	4.0406	4.0410	4.0413	4.0416	4.0420
2	2.674	4.0403	4.0408	4.0412	4.0415	4.0418	4.0422
2	2.676	4.0405	4.0410	4.0414	4.0417	4.0420	4.0424
2	2.678	4.0407	4.0412	4.0416	4.0419	4.0422	4.0426
2	2.680	4.0409	4.0414	4.0418	4.0421	4.0424	4.0428
2	2.682	4.0411	4.0416	4.0420	4.0423	4.0426	4.0430
2	2.684	4.0413	4.0418	4.0422	4.0425	4.0428	4.0432
2	2.686	4.0415	4.0420	4.0424	4.0427	4.0430	4.0434
2	2.688	4.0417	4.0422	4.0426	4.0429	4.0432	4.0436
2	2.690	4.0419	4.0424	4.0428	4.0431	4.0434	4.0438
2	2.692	4.0421	4.0426	4.0430	4.0433	4.0436	4.0440
2	2.694	4.0423	4.0428	4.0432	4.0435	4.0438	4.0442
2	2.696	4.0425	4.0430	4.0434	4.0437	4.0440	4.0444
2	2.698	4.0427	4.0432	4.0436	4.0439	4.0442	4.0446
2	2.700	4.0429	4.0434	4.0438	4.0441	4.0444	4.0448
2	2.702	4.0431	4.0436	4.0440	4.0443	4.0446	4.0450
2	2.704	4.0433	4.0438	4.0442	4.0445	4.0448	4.0452
2	2.706	4.0435	4.0440	4.0444	4.0447	4.0450	4.0454
2	2.708	4.0437	4.0442	4.0446	4.0449	4.0452	4.0456
2	2.710	4.0439	4.0444	4.0448	4.0451	4.0454	4.0458
2	2.712	4.0441	4.0446	4.0450	4.0453	4.0456	4.0460
2	2.714	4.0443	4.0448	4.0452	4.0455	4.0458	4.0462
2	2.716	4.0445	4.0450	4.0454	4.0457	4.0460	4.0464
2	2.718	4.0447	4.0452	4.0456	4.0459	4.0462	4.0466
2	2.720	4.0449	4.0454	4.0458	4.0461	4.0464	4.0468
2	2.722	4.0451	4.0456	4.0460	4.0463	4.0466	4.0470
2	2.724	4.0453	4.0458	4.0462	4.0465	4.0468	4.0472
2	2.726	4.0455	4.0460	4.0464	4.0467	4.0470	4.0474
2	2.728	4.0457	4.0462	4.0			

PARAMETRIC DATA

BETA =	.000	ELV-LO =	-10,000
ELV-L1 =	-10,000	ELV-R1 =	-30,000
ELV-RO =	-30,000	BDFLAP =	,000
SPARK =	25,000	ALTRCN =	10,000
ELEVTR =	-20,000		

RUN NO. 31/0

MACH	ALPHA	BETA	CN	CA	CLH	CDL	CR	CN	CA	CLH	CDL	CR	CN	CA	CLH	CDL	CR	CN	CA	CLH	CDL	CR	
2.500	-4.504	.01826	-1.17268	.12861	.00015	.00569	.00439	.00172	-.16316	.14047	-1.16171												
2.500	-7.013	.01753	-.03795	.12493	-.00247	.00511	.00324	-.00609	-.03768	.12694	-.30317												
2.500	4.567	.01823	.15363	.12076	-.01882	.00435	.00234	.00470	.09423	.12777	.17751												
2.500	9.135	.01929	.15356	.11645	-.02893	.00418	.00198	.00520	.05198	.21701	.14865	1.45986											
2.500	12.234	.01964	.15726	.11317	-.03740	.00372	.00119	.00407	.06119	.34471	.19055	1.46207											
2.500	16.527	.01964	.15289	.11016	-.04153	.00365	.00073	.00292	.09073	.47056	.25264	1.46382											
2.500	20.422	.01964	.15236	.10827	-.05172	.00371	.00050	.00442	.06051	.35699	.179202												
2.500	24.324	.01964	.15213	.10211	-.05795	.00375	.00046	.00446	.072341	.44316	.163239												
2.500	28.623	.01964	.15205	.10104	-.05917	.00375	.00046	.00446	.072341	.44316	.163239												
2.500	32.465	.01964	.15242	.10474	.09408	-.08032	.00561	-.00019	.00292	.09374	.53574	.72570	1.31423										

MACH	ALPHA	BETA	CN	CA	CLH	CDL	CR	CN	CA	CLH	CDL	CR	CN	CA	CLH	CDL	CR	CN	CA	CLH	CDL	CR
4.610	5.514	.01937	.05609	-.02243	.001543	.00049	-.00262	-.11320	.10328	-.100403												
4.610	-4.567	.01970	.05179	-.02154	.001343	.00048	-.00219	-.07501	.09409	-.170723												
4.610	4.444	.01970	.05179	-.02262	.001269	.00065	-.00247	-.03252	.09753	-.37146												
4.610	2.457	.01970	.05179	-.02298	.001215	.00038	-.00429	.00631	.08478	.07441												
4.610	4.471	.01970	.05179	-.02346	.001174	.00038	-.00392	.05283	.08478	.07441												
4.610	4.471	.01970	.05179	-.02470	.001147	.00038	-.00392	.01391	.14548	.16163												
4.610	12.877	.01970	.05179	-.02543	.001111	.00035	-.00249	.24931	.100112	.145306												
4.610	12.877	.01970	.05179	-.02571	.001053	.00035	-.00249	.13590	.13590	.194568												
4.610	1.754	.01970	.05179	-.02537	.001017	.00035	-.00249	.35973	.170149	.190149												
4.610	1.754	.01970	.05179	-.02502	.001012	.00035	-.00249	.25777	.170149	.190149												
4.610	1.754	.01970	.05179	-.02473	.0010130	.00035	-.00249	.170149	.170149	.190149												
4.610	1.754	.01970	.05179	-.02443	.0010152	.00035	-.00249	.100112	.100112	.145306												
4.610	1.754	.01970	.05179	-.02417	.0010111	.00035	-.00249	.01647	.01647	.01647												
4.610	1.754	.01970	.05179	-.02386	.0010111	.00035	-.00249	.09813	.09813	.09813												
4.610	1.754	.01970	.05179	-.02356	.0010111	.00035	-.00249	.09813	.09813	.09813												
4.610	1.754	.01970	.05179	-.02326	.0010111	.00035	-.00249	.09813	.09813	.09813												
4.610	1.754	.01970	.05179	-.02295	.0010111	.00035	-.00249	.09813	.09813	.09813												
4.610	1.754	.01970	.05179	-.02265	.0010111	.00035	-.00249	.09813	.09813	.09813												
4.610	1.754	.01970	.05179	-.02235	.0010111	.00035	-.00249	.09813	.09813	.09813												
4.610	1.754	.01970	.05179	-.02204	.0010111	.00035	-.00249	.09813	.09813	.09813												
4.610	1.754	.01970	.05179	-.02174	.0010111	.00035	-.00249	.09813	.09813	.09813												
4.610	1.754	.01970	.05179	-.02144	.0010111	.00035	-.00249	.09813	.09813	.09813												
4.610	1.754	.01970	.05179	-.02114	.0010111	.00035	-.00249	.09813	.09813	.09813												
4.610	1.754	.01970	.05179	-.02084	.0010111	.00035	-.00249	.09813	.09813	.09813												
4.610	1.754	.01970	.05179	-.02054	.0010111	.00035	-.00249	.09813	.09813	.09813												
4.610	1.754	.01970	.05179	-.02024	.0010111	.00035	-.00249	.09813	.09813	.09813												
4.610	1.754	.01970	.05179	-.01993	.0010111	.00035	-.00249	.09813	.09813	.09813												
4.610	1.754	.01970	.05179	-.01963	.0010111	.00035	-.00249	.09813	.09813	.09813												

REPRODUCED
BY SPERRY

AAS TABULATED SOURCE DATA

LA-49 UNIT 1101 H-1-0398/139 CFS SPLIT ELEV

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(RHJ/D3)

PARAMETRIC DATA

BETA =	,000	EL-V-LO =	,000
EL-V-L1 =	-10,000	EL-V-R1 =	-10,000
EL-V-R0 =	-20,000	EFLAP =	,000
SPOKE =	25,000	ELEVR =	-10,000
AUTPCN =	10,000		

WATER	32.0	CB.	CN	C	L/D
2.50	1.4	3.06	.1049	.14172	.13196
3.00	1.3	2.96	.1049	.14172	.106487
3.50	1.2	2.86	.1049	.14172	.14459
4.00	1.1	2.76	.1049	.14172	.90544
4.50	1.0	2.66	.1049	.14172	.119213
5.00	.9	2.56	.1049	.14172	.49115
5.50	.8	2.46	.1049	.14172	.10110
6.00	.7	2.36	.1049	.14172	.14672
6.50	.6	2.26	.1049	.14172	.40713
7.00	.5	2.16	.1049	.14172	.10713
7.50	.4	2.06	.1049	.14172	.14713
8.00	.3	1.96	.1049	.14172	.41713
8.50	.2	1.86	.1049	.14172	.10713
9.00	.1	1.76	.1049	.14172	.14713
9.50	0	1.66	.1049	.14172	.41713
10.00		1.56	.1049	.14172	.10713
10.50		1.46	.1049	.14172	.14713
11.00		1.36	.1049	.14172	.41713
11.50		1.26	.1049	.14172	.10713
12.00		1.16	.1049	.14172	.14713
12.50		1.06	.1049	.14172	.41713
13.00		.96	.1049	.14172	.10713
13.50		.86	.1049	.14172	.14713
14.00		.76	.1049	.14172	.41713
14.50		.66	.1049	.14172	.10713
15.00		.56	.1049	.14172	.14713
15.50		.46	.1049	.14172	.41713
16.00		.36	.1049	.14172	.10713
16.50		.26	.1049	.14172	.14713
17.00		.16	.1049	.14172	.41713
17.50		.06	.1049	.14172	.10713
18.00			.1049	.14172	.14713

WATER	32.0	CB.	CN	C	L/D
2.50	1.4	3.06	.1049	.14172	.13196
3.00	1.3	2.96	.1049	.14172	.106487
3.50	1.2	2.86	.1049	.14172	.14459
4.00	1.1	2.76	.1049	.14172	.90544
4.50	1.0	2.66	.1049	.14172	.119213
5.00	.9	2.56	.1049	.14172	.49115
5.50	.8	2.46	.1049	.14172	.10110
6.00	.7	2.36	.1049	.14172	.14672
6.50	.6	2.26	.1049	.14172	.40713
7.00	.5	2.16	.1049	.14172	.10713
7.50	.4	2.06	.1049	.14172	.14713
8.00	.3	1.96	.1049	.14172	.41713
8.50	.2	1.86	.1049	.14172	.10713
9.00	.1	1.76	.1049	.14172	.14713
9.50	0	1.66	.1049	.14172	.41713
10.00		1.56	.1049	.14172	.10713
10.50		1.46	.1049	.14172	.14713
11.00		1.36	.1049	.14172	.41713
11.50		1.26	.1049	.14172	.10713
12.00		1.16	.1049	.14172	.14713
12.50		1.06	.1049	.14172	.41713
13.00		.96	.1049	.14172	.10713
13.50		.86	.1049	.14172	.14713
14.00		.76	.1049	.14172	.41713
14.50		.66	.1049	.14172	.10713
15.00		.56	.1049	.14172	.14713
15.50		.46	.1049	.14172	.41713
16.00		.36	.1049	.14172	.10713
16.50		.26	.1049	.14172	.14713
17.00		.16	.1049	.14172	.41713
17.50		.06	.1049	.14172	.10713
18.00			.1049	.14172	.14713

(RHJ01A)

PARAMETRIC DATA

	BETA =	.000	ELV-L0 =	.000
	ELV-L1 =	-20,000	ELV-R1 =	-20,000
	ELV-R0 =	-40,000	BDFLAP =	.000
	SPARK =	25,000	ELEVTR =	-20,000
	AIRFON =	20,000		
RUN NO. 3350				
WIND	BETA	CN	CM	CD
2.5% 2.5%	.0000 .0000	.117500 .117500	.130468 .010560	.00154 .16540
2.5% 2.5%	.0000 .0000	.12561 .12561	.00239 .00119	.00168 .03287
2.5% 2.5%	.0000 .0000	.12116 .12116	.01776 .01776	.00142 .00142
2.5% 2.5%	.0000 .0000	.11745 .11745	.02857 .01672	.00143 .00143
2.5% 2.5%	.0000 .0000	.11339 .11339	.01377 .01672	.00097 .00264
2.5% 2.5%	.0000 .0000	.11021 .11021	.02435 .01457	.00095 .00264
2.5% 2.5%	.0000 .0000	.10775 .10775	.02099 .01056	.00095 .00264
2.5% 2.5%	.0000 .0000	.10524 .10524	.019901 .019901	.00091 .00274
2.5% 2.5%	.0000 .0000	.10315 .10315	.016970 .016970	.00086 .00280
2.5% 2.5%	.0000 .0000	.10114 .10114	.014740 .014740	.00083 .00283
2.5% 2.5%	.0000 .0000	.09913 .09913	.012640 .012640	.00079 .00287
2.5% 2.5%	.0000 .0000	.09712 .09712	.010540 .010540	.00075 .00291
2.5% 2.5%	.0000 .0000	.09511 .09511	.008440 .008440	.00071 .00295
2.5% 2.5%	.0000 .0000	.09310 .09310	.006340 .006340	.00067 .00299
2.5% 2.5%	.0000 .0000	.09109 .09109	.004240 .004240	.00063 .00303
2.5% 2.5%	.0000 .0000	.08908 .08908	.002147 .002251	.00059 .00313
2.5% 2.5%	.0000 .0000	.08707 .08707	.0012300 .0012349	.00055 .00277
2.5% 2.5%	.0000 .0000	.08506 .08506	.0002277 .0002277	.00051 .00327
2.5% 2.5%	.0000 .0000	.08305 .08305	.0002526 .0002526	.00048 .00293
2.5% 2.5%	.0000 .0000	.08104 .08104	.000275 .000275	.00045 .00291
2.5% 2.5%	.0000 .0000	.07903 .07903	.0003074 .000323	.00042 .00285
2.5% 2.5%	.0000 .0000	.07702 .07702	.0003394 .0003513	.00039 .00283
2.5% 2.5%	.0000 .0000	.07501 .07501	.0003704 .0003817	.00036 .00287
2.5% 2.5%	.0000 .0000	.07300 .07300	.0004090 .0004090	.00033 .00293
2.5% 2.5%	.0000 .0000	.07100 .07100	.0004494 .0004494	.00030 .00295
2.5% 2.5%	.0000 .0000	.06900 .06900	.0004898 .0004898	.00027 .00295
2.5% 2.5%	.0000 .0000	.06700 .06700	.0005292 .0005292	.00024 .00295
2.5% 2.5%	.0000 .0000	.06500 .06500	.0005683 .0005683	.00021 .00295
2.5% 2.5%	.0000 .0000	.06300 .06300	.0006070 .0006070	.00018 .00295
2.5% 2.5%	.0000 .0000	.06100 .06100	.0006469 .0006469	.00015 .00295
2.5% 2.5%	.0000 .0000	.05900 .05900	.0006856 .0006856	.00012 .00295
2.5% 2.5%	.0000 .0000	.05700 .05700	.0007235 .0007235	.00010 .00295
2.5% 2.5%	.0000 .0000	.05500 .05500	.0007610 .0007610	.00008 .00295
2.5% 2.5%	.0000 .0000	.05300 .05300	.0007986 .0007986	.00006 .00295
2.5% 2.5%	.0000 .0000	.05100 .05100	.0008351 .0008351	.00004 .00295
2.5% 2.5%	.0000 .0000	.04900 .04900	.0008717 .0008717	.00002 .00295
2.5% 2.5%	.0000 .0000	.04700 .04700	.0009082 .0009082	.00000 .00295
2.5% 2.5%	.0000 .0000	.04500 .04500	.0009447 .0009447	.00000 .00295
2.5% 2.5%	.0000 .0000	.04300 .04300	.0009812 .0009812	.00000 .00295
2.5% 2.5%	.0000 .0000	.04100 .04100	.0010177 .0010177	.00000 .00295
2.5% 2.5%	.0000 .0000	.03900 .03900	.0010542 .0010542	.00000 .00295
2.5% 2.5%	.0000 .0000	.03700 .03700	.0010907 .0010907	.00000 .00295
2.5% 2.5%	.0000 .0000	.03500 .03500	.0011272 .0011272	.00000 .00295
2.5% 2.5%	.0000 .0000	.03300 .03300	.0011637 .0011637	.00000 .00295
2.5% 2.5%	.0000 .0000	.03100 .03100	.0012002 .0012002	.00000 .00295
2.5% 2.5%	.0000 .0000	.02900 .02900	.0012367 .0012367	.00000 .00295
2.5% 2.5%	.0000 .0000	.02700 .02700	.0012732 .0012732	.00000 .00295
2.5% 2.5%	.0000 .0000	.02500 .02500	.0013097 .0013097	.00000 .00295
2.5% 2.5%	.0000 .0000	.02300 .02300	.0013462 .0013462	.00000 .00295
2.5% 2.5%	.0000 .0000	.02100 .02100	.0013827 .0013827	.00000 .00295
2.5% 2.5%	.0000 .0000	.01900 .01900	.0014192 .0014192	.00000 .00295
2.5% 2.5%	.0000 .0000	.01700 .01700	.0014567 .0014567	.00000 .00295
2.5% 2.5%	.0000 .0000	.01500 .01500	.0014941 .0014941	.00000 .00295
2.5% 2.5%	.0000 .0000	.01300 .01300	.0015315 .0015315	.00000 .00295
2.5% 2.5%	.0000 .0000	.01100 .01100	.0015689 .0015689	.00000 .00295
2.5% 2.5%	.0000 .0000	.00900 .00900	.0016063 .0016063	.00000 .00295
2.5% 2.5%	.0000 .0000	.00700 .00700	.0016437 .0016437	.00000 .00295
2.5% 2.5%	.0000 .0000	.00500 .00500	.0016811 .0016811	.00000 .00295
2.5% 2.5%	.0000 .0000	.00300 .00300	.0017185 .0017185	.00000 .00295
2.5% 2.5%	.0000 .0000	.00100 .00100	.0017559 .0017559	.00000 .00295
2.5% 2.5%	.0000 .0000	.00000 .00000	.0017933 .0017933	.00000 .00295

TABULATED SOURCE DATA

JA-49 SPRT 11:01 7 3898/139 CRS SPLIT D.EVN

(RHJN19)

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PARAMETRIC DATA

BETA =	.500	E.V-L0 =	-10.000
E.V-L1 =	-20.000	E.V-R1 =	-20.000
E.V-R0 =	-50.000	E.DFLD =	.500
E.DFLX =	25.000	E.EVRC =	-20.000
A.FRCN =	10.000		

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LA-49 UPWT 1101 N1-0698/139 CRB SPLIT ELEVON

(RH-JD16)

PARAMETRIC DATA

BETA = .0000 ELV-LO z = 10,000
 ELV-LI = -20,000 ELV-RI = -20,000
 ELV-FO = -10,000 BDFLAP = .0000
 SPDRK = 25,000 ELEVTR = -20,000
 AIRCON = 10,000

RUN NO. 35/ 0

MACH	ALPHA	BETA	CN	CA	CLM	CDL	CYN	CT	CL	CD	L/D
2.500	-4.047	.02087	-.13388	.12321	-.00164	.00140	-.00044	-.00103	-.14447	.13575	-1.06426
2.500	.017	.02198	-.01981	.12165	-.01338	.00166	-.00089	-.00146	.01937	.12165	-1.16087
2.500	.049	.02196	.11837	.11818	-.02844	.00203	-.00133	-.00029	.10973	.12624	.86922
2.500	0.126	.02407	.23261	.11579	-.03971	.00253	-.00176	-.00218	.23370	.15034	1.55445
2.500	12.235	.02429	.39762	.11350	-.04898	.00301	-.00178	-.00254	.36458	.19499	1.86971
2.500	16.312	.02596	.53997	.11119	-.03862	.00358	-.00158	-.00151	.48700	.25838	1.86849
2.500	20.434	.02533	.70124	.10841	-.06631	.00414	-.00223	-.00292	.61927	.34641	1.78764
2.500	24.535	.02572	.85447	.10542	-.07369	.00498	-.00247	-.00292	.74264	.45487	1.63266
2.500	28.639	.02578	1.03679	.10090	-.08567	.00526	-.00281	-.00244	.86158	.58548	1.47157
2.500	32.751	.02680	1.21993	.09667	-.10015	.00574	-.00313	-.00307	.97569	.74128	1.31353

RUN NO. 17/ 0

MACH	ALPHA	BETA	CN	CA	CLM	CDL	CYN	CT	CL	CD	L/D
3.500	-3.808	.02248	-.10656	.09361	-.03053	.00109	-.00163	-.00403	-.10036	.10013	-1.00426
3.500	-1.803	.02211	-.06301	.08971	-.02877	.00104	-.00164	-.00304	.06217	.09149	.68282
3.500	.475	.02170	.02364	.08622	-.02998	.00117	-.00164	-.00198	.02435	.08602	.28309
3.500	2.457	.02264	.02186	.08347	-.02975	.00095	-.00207	-.00336	.01628	.08433	.21679
3.500	4.133	.02224	.06333	.08072	-.03028	.00107	-.00208	-.00232	.05690	.08538	.66644
3.500	6.337	.02223	.16597	.07773	-.03069	.00198	-.00208	-.00240	.15277	.10124	1.56904
3.500	12.504	.02293	.27976	.07925	-.03343	.00244	-.00294	-.00201	.25618	.13696	1.87044
3.500	16.403	.02369	.40554	.07933	-.03584	.00295	-.00294	-.00406	.36547	.19120	1.91655
3.500	20.958	.02311	.56867	.08355	-.03930	.00397	-.00293	-.01260	.50117	.28143	1.77079
3.500	24.681	.02411	.70890	.08587	-.04390	.00483	-.00339	-.00418	.60828	.37404	1.62625
3.500	28.652	.02494	.87018	.08795	-.05298	.00589	-.00402	-.00506	.72145	.49442	1.49519
3.500	32.685	.02537	1.03043	.09099	-.06315	.00657	-.00465	-.00544	.83316	.64612	1.28849
3.500	36.844	.02549	1.23056	.09279	-.07344	.00698	-.00464	-.00549	.92698	.81203	1.14403
3.500	41.541	.02607	1.43615	.09314	-.08682	.00742	-.00548	-.00538	1.01316	.99125	1.02210

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LA49 TABULATED SOURCE DATA

LA-49 UWT 1101 RI-0000/139 CRB SPLIT ELEVON

PAGE 17

(RHJ017)

PARAMETRIC DATA

BETA =	.000	ELY-LO =	10,000
ELY-L1 =	-40,000	ELY-RI =	-40,000
ELY-RO =	-10,000	EOLFLAP =	,000
SPOBRK =	25,000	ELEVTR =	-40,000
ATLRCN =	10,000		

RUN NO. 36/ 0

MACH	ALPHA	BETA	CN	CLM	CBL	CFN	CF	CL	CD	L/D
2.500	-4.081	.02102	-.18425	.13852	.01381	-.00001	-.00240	-.17407	.14926	-1.16605
2.500	-1.012	.02101	.04410	.13429	.00046	.00178	-.00045	-.04407	.13430	-3.2615
2.500	4.070	.02103	.02803	.12351	.01679	.00007	-.00089	-.17128	.09869	.15442
2.500	8.141	.022282	.23813	.12351	.02979	.00267	-.00134	.00155	.21924	.15599
2.500	12.234	.022347	.38675	.11693	.03545	.00325	-.00156	.00193	.34690	.19691
2.500	16.324	.022440	.50995	.11563	.03706	.00373	-.00201	.00211	.47604	.25993
2.500	20.430	.022575	.69091	.11267	.05793	.00420	-.00224	.00347	.60812	.34676
2.500	24.533	.022607	.74456	.10972	.06631	.00466	-.00248	.00338	.73186	.45464
2.500	28.606	.022625	1.02533	.10849	.07816	.00531	-.00283	.00377	.84957	.58329
2.500	32.759	.022619	1.25471	.10247	.092920	.00570	-.00315	.00444	.95766	.73904

RUN NO. 18/ 0

MACH	ALPHA	BETA	CN	CLM	CBL	CFN	CF	CL	CD	L/D
4.600	-3.543	.022287	-.11603	.09556	-.02494	.00106	-.00164	-.11184	.10367	-1.07878
4.600	-1.615	.022229	-.07634	.09253	-.02340	.00105	-.00165	-.07350	.09464	-7.77776
4.600	.420	.02184	.03096	.08864	-.02455	.00097	-.00165	-.02333	.03161	.08661
4.600	2.337	.022260	.01052	.08569	-.02577	.00093	-.00208	-.00378	.00702	.08600
4.600	4.442	.022231	.01959	.08211	-.02757	.00106	-.00208	-.00251	.05303	.08648
4.600	6.556	.022290	.18595	.07867	-.02812	.00174	-.00209	-.00251	.10243	.61343
4.600	12.569	.022253	.27512	.07898	-.02933	.00242	-.00232	-.00232	.25231	.10243
4.600	16.491	.022333	.40570	.09090	-.03508	.00284	-.00253	-.00423	.36603	.13716
4.600	20.728	.022326	.55737	.19432	-.03534	.00395	-.00296	-.00301	.49138	.19274
4.600	24.709	.022422	.70510	.08730	-.03913	.00498	-.00340	-.00446	.60384	.27653
4.600	28.738	.022496	.86282	.09008	-.04755	.00567	-.00403	-.00543	.71306	.37450
4.600	33.023	.022551	1.04569	.09320	-.05626	.00534	-.00486	-.00577	.82681	.61551
4.600	37.030	.022603	1.22659	.09822	-.06516	.00695	-.00508	-.00619	.92127	.12969
4.600	41.135	.022634	1.40611	.09771	-.07614	.00754	-.00590	-.00657	.99480	.99617

(RH-1016)

PARAMETRIC DATA

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BETA = .0000 Q.V-I.O = -20.000
ELV-LI = -30.000 Q.V-P1 = -30.000
ELV-FD = -40.000 EDFLAP = .000
SPDRK = 25.000 EL.EVTR = -30.000
ATLCON = 0.000
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RUN NO. 3710

WACH	ALPHA	BETA	CY	CA	CLW	CBL	CYN	CL	CY	CD	L/D	
2.500	-4.060	.01785	-.20014	.13741	.02461	.00127	.00112	-.00034	.16991	.15123	-1.25373	
2.500	-1.001	.01953	-.05960	.13309	.01004	.00069	.00068	-.00066	.05960	.13309	-1.44794	
2.500	4.053	.01953	.01953	.08232	.12711	-.00398	.00136	.00046	.00065	.07334	.13262	.53295
2.500	6.150	.01957	.01957	.12322	.12273	-.01654	.00139	.00070	-.00274	.19861	.15237	1.35625
2.500	12.229	.02031	.06097	.11658	.02465	.00084	.00047	.00036	.00036	.36009	.16735	1.72324
2.500	16.322	.02105	.06284	.11298	.03381	.00038	.00024	-.00024	.00029	.45416	.25067	1.61151
2.500	20.432	.02171	.06403	.10980	.03091	.00046	.00046	-.00046	.58398	.33432	1.74574	
2.500	24.532	.02234	.06453	.09453	.03451	-.00012	.00044	-.00044	.75714	.43264	1.61214	
2.500	28.633	.02292	.06482	.10149	.03189	.00014	-.00013	-.00013	.82250	.55458	1.45666	
2.500	32.733	.02359	.07318	.05386	-.05370	.00024	.00023	-.00023	.93409	.71839	1.35351	

R.JN. (C.) 1970

W.CD	ALPHA	BETA	CY	CA	CLW	CBL	CYN	CL	CY	CD	L/D
4.000	-3.140	.02470	-.12657	.10137	-.01416	.00369	.00048	-.00033	-.12063	.10456	-1.11114
4.000	-1.501	.02471	-.06463	.09732	-.01406	.00286	.00047	-.00046	-.06605	.09961	-.66388
4.000	.586	.02475	-.04664	.09280	-.01590	.00215	.00004	-.00043	-.04726	.09248	-.51104
4.000	2.592	.02474	.05279	.38925	-.01762	.00159	-.00039	-.00042	-.00120	.05629	-.03361
4.000	4.593	.02471	.01721	.0442	-.01971	.00137	-.00040	-.00034	.04533	.04830	.51339
4.000	6.594	.02474	.02354	.01831	-.01657	.00083	-.00083	-.00083	.13818	.10146	1.36195
4.000	8.595	.02473	.02168	.02024	.00064	-.00127	-.00285	-.00285	.24592	.13722	1.79215
4.000	10.596	.02473	.02132	.02063	.00027	-.00129	-.00326	-.00326	.34315	.18479	1.95097
4.000	12.597	.02473	.02124	.02324	-.01876	.00023	-.00130	-.00130	.47601	.26862	1.76284
4.000	14.598	.02474	.02172	.02752	-.02176	.00021	-.00176	-.00176	.59544	.35413	1.85213
4.000	16.599	.02473	.02142	.02699	-.02672	.00019	-.00156	-.00156	.69719	.48431	1.43957
4.000	18.599	.02473	.02164	.02640	-.02313	.00035	-.00218	-.00218	.79799	.52023	1.28595
4.000	20.599	.02473	.02337	.01857	-.03982	.00041	-.00175	-.00175	.89399	.70555	1.13804
4.000	21.739	.02473	.02316	.03654	-.04935	.00066	-.00176	-.00176	.98008	.98298	

